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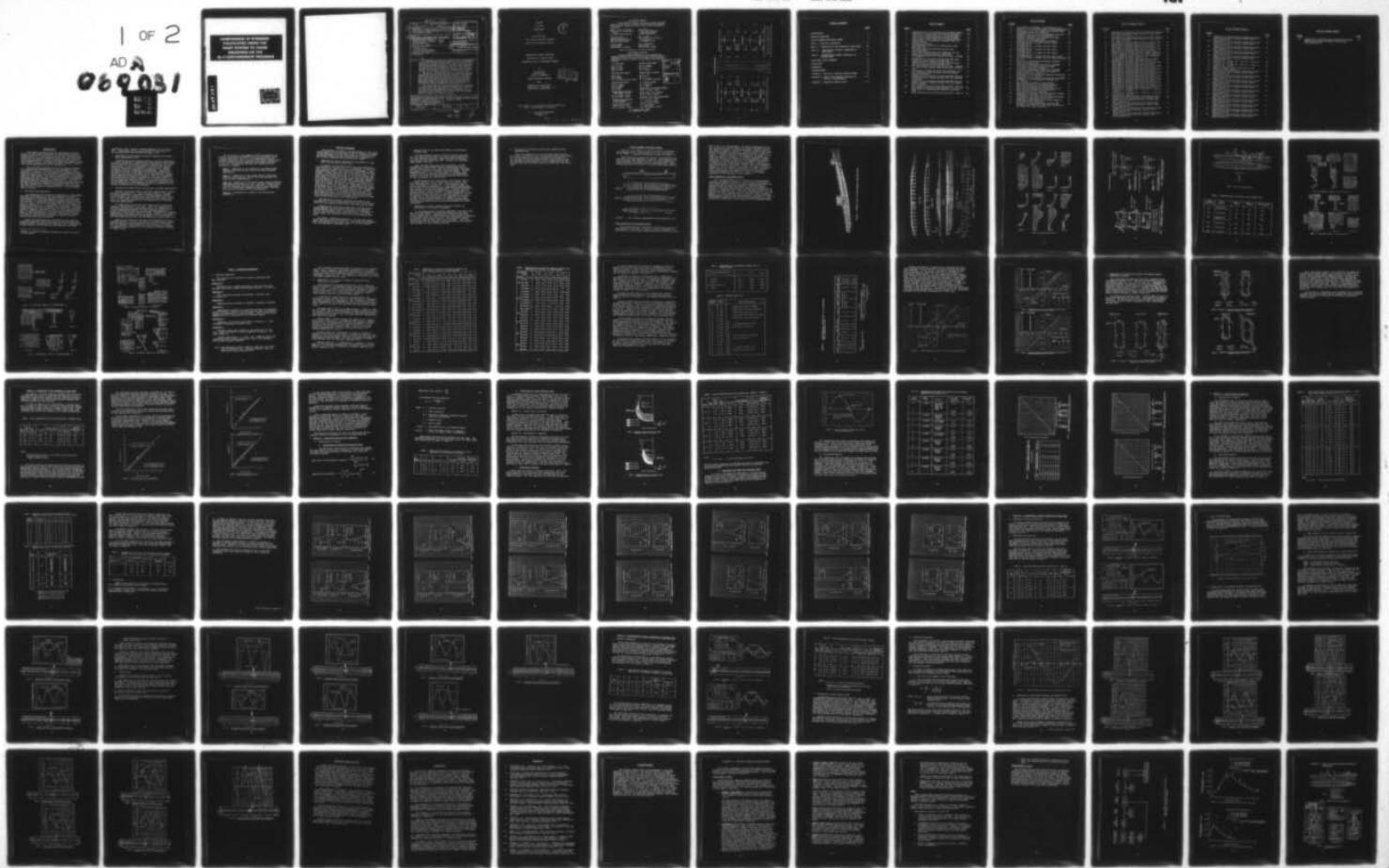
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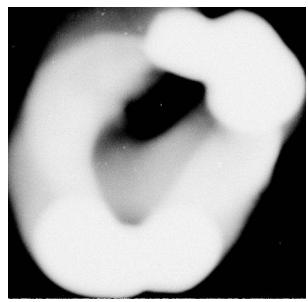
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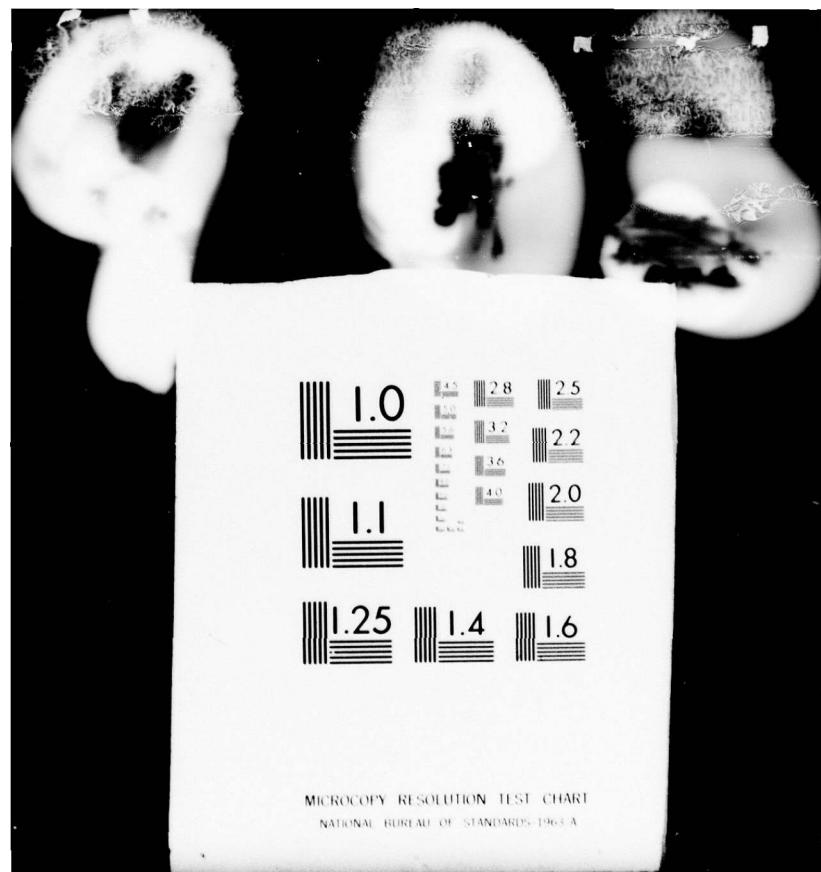
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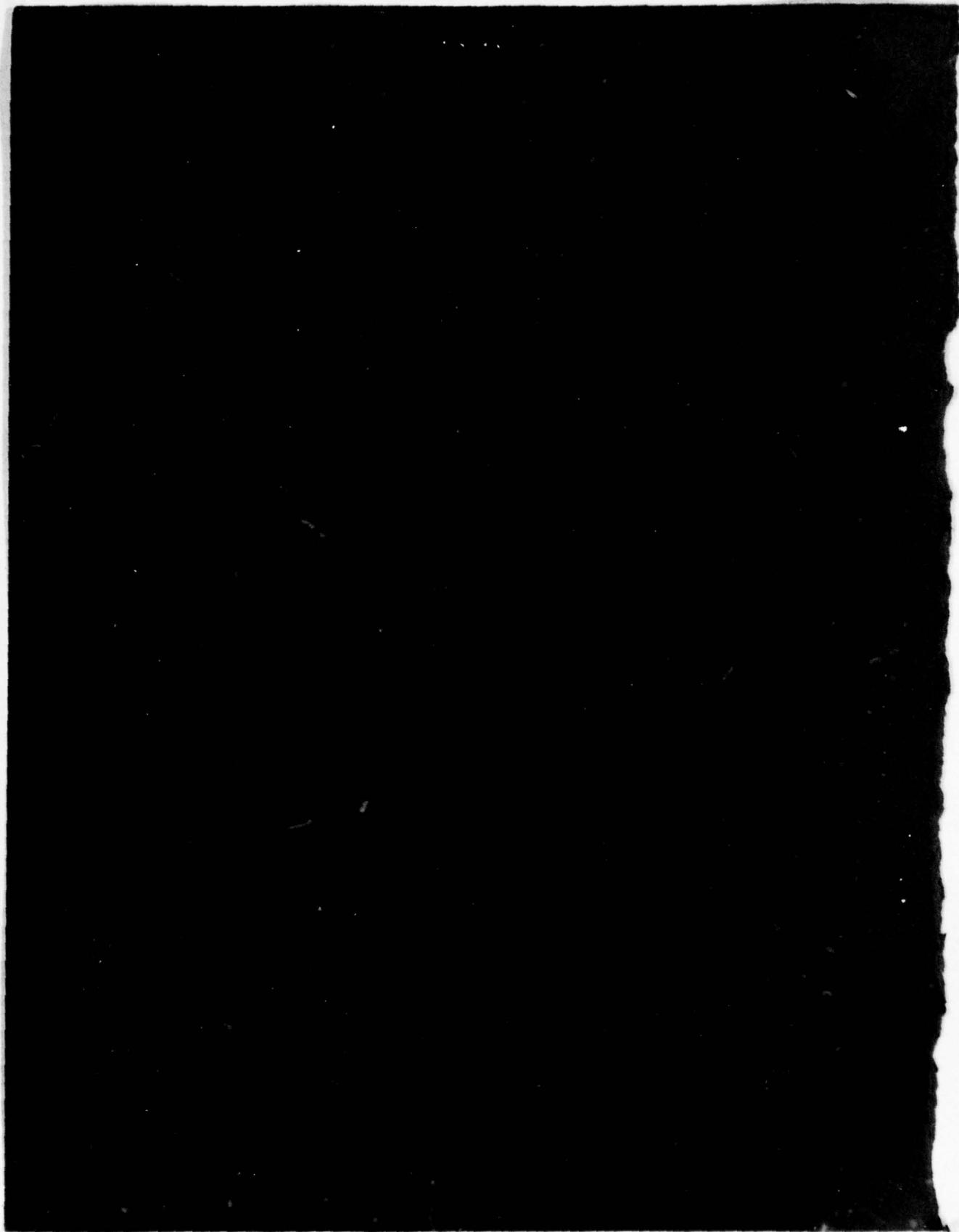




**COMPARISON OF STRESSES  
CALCULATED USING THE  
DAISY SYSTEM TO THOSE  
MEASURED ON THE  
SL-7 CONTAINERSHIP PROGRAM**

*AS 69 031*





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16. Abstract A comparison of stresses calculated using the ABS/DAISY system with those measured on board the SL-7 Containership is undertaken to verify the analytical procedures used in assessing the strength of ships in a seaway. The comparisons and evaluations are performed for four different and progressively more severe technical conditions: dockside calibration, RMS stresses in head seas, instantaneous stresses in head seas and instantaneous stresses in oblique seas.			
The overall comparison between calculated and measured stresses for the dockside calibration is good where thermal effects were small but inconclusive elsewhere. The comparison of RMS stresses in head seas is generally satisfactory, using both the spectrum analysis approach and the equivalent regular wave approach, and the comparison of instantaneous stresses in head seas and in oblique seas is also good for the wave conditions considered.			
The results show that the existing analytical tools for predicting wave loads and structural responses are suitable to assess the overall strength of the hull-girder. All the measured and calculated hull-girder stresses are of low magnitude, and no modifications to the present hull-girder strength standard are deemed necessary.			
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"SL-7 Stress Calculations Compared  
With Full-Scale Measured Values"

COMPARISON OF STRESSES CALCULATED  
USING THE DAISY SYSTEM TO THOSE  
MEASURED ON THE SL-7 CONTAINERSHIP PROGRAM

by  
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under  
Department of Transportation  
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U. S. Coast Guard Headquarters  
Washington, D.C.  
1979

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## INTRODUCTION

This report summarizes the work performed by the Research and Development Department of the American Bureau of Shipping under U.S. Coast Guard Contract No. DOT-CG-63176-A. This project is jointly sponsored by the Ship Structure Committee and the American Bureau of Shipping, and is part of the comprehensive SL-7 Containership Research Program of ship loads, model testing, structural analysis, response analysis, full-scale measurements and data correlation.

The long-range objective of the SL-7 Research Program is to advance the understanding of the responses of hull structures at sea and to verify the current design criteria. Emphasis is focused on the correlation of experimental (full-scale and model) data and theoretical predictions at different phases of the program. The overall research plan, historical background and the interface of various phases of the program were presented and discussed in Reference [1]\*. The SL-7 Research Program was sponsored by the Ship Structure Committee, the American Bureau of Shipping and Sea-Land Services, Inc. The major phases which have been completed, either directly under the SL-7 program or associated with it, can be summarized as follows:

### 1. Full-scale Instrumentation

To investigate the structural responses of a high-speed open-deck containership at sea, an extensive full-scale instrumentation system was installed on board the SL-7 class containership S.S. SEA-LAND McLEAN to measure wave heights, ship motions, accelerations and wave-induced stresses in many critical areas. The detailed information for this system is given in Reference [2]. In addition, a micro-wave radar was developed and installed to measure wave elevations. A finite element analysis of the entire ship [3], using the ABS/DAISY system of computer programs [4], and a steel structural model test [5] were carried out at the planning and installation stage of the instrumentation program. The results of these two research projects were utilized to identify critical regions for strain-gage locations.

After the instrumentation installation was completed, a dockside calibration was carried out by Teledyne Materials Research Company and reported in Reference [6]. Subsequently, a large amount of stress data has been acquired for three

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\* Numbers in brackets designate References listed at the end of the report.

consecutive winter seasons, between September 1972 and March 1975. Some sample results are presented in Reference [7]. The wave-meter data was analyzed by Dalzell [8].

## 2. Correlation of Calculated Structural Responses and Steel Model Experimental Data

Although the finite element method utilized in predicting structural responses to quasi-static loads has been well tested and verified with full-scale and model experimental results in recent years, it is still desirable to validate the analysis procedure and modeling techniques in dealing with a special structure, such as an open-deck containership. Accordingly, the American Bureau of Shipping has performed a structural analysis of the SL-7 Steel Model, using the ABS/DAISY system. The calculated results together with a comparison with experimental data were presented by Elbatouti, Jan and Stiansen [9]. The predicted hull-girder responses to both bending and torsional loads were generally found to be in good agreement with the measured data. Consequently, the ABS/DAISY system and the current modeling practice employed at ABS are considered satisfactory in dealing with containerships.

## 3. Correlation of Ship-Motion Calculations and Model Experiments

Another significant phase of the SL-7 program was to verify the wave-load prediction. Under the sponsorship of the Ship Structure Committee, a ship-motion computer program, SCORES, was developed in 1972 based on the two-dimensional strip theory [10]. It was generally understood that the strip theory was valid for full-form vessels, but its application to fine-form ships was questionable.

To assess ship motions and wave loads, an SL-7 model was tested in oblique seas in 1974 [11]. The first comparison of the model test data with the SCORES results [12] showed significant discrepancies between the RAOs (Response Amplitude Operators) of ship motions and wave loads. Subsequently, Oceanics Incorporated introduced a speed correction factor in the existing SCORES program and obtained good agreement of the RAO between the model experimental data and the theoretical predictions [13].

Based on the results of the above three phases of the SL-7 research program, it can be seen that the finite element techniques and wave-load predictions have been generally validated. The final phase in the verification of the analytical procedure in assessing ship strength is a correlation of dynamic stresses of ships at sea.

The objective of this phase of the SL-7 program is to compare the stresses calculated using the ABS/DAISY system to those measured on the SL-7, in corresponding sea and dockside conditions, and to evaluate the results through each of four different and progressively more severe technical conditions. Accordingly, the work was divided into the following four tasks:

TASK I - Comparison of the results of the DAISY stress analysis with the results of the full-scale dockside calibration.

TASK II - Comparison of the stress spectra calculated from the DAISY analysis results with the selected full-scale at-sea stress spectra.

TASK III - Comparison of the DAISY stress analysis results, using measured acceleration and a specific, selected wave profile estimated from the wave radar data in head seas, with the instantaneous measured stresses taken simultaneously while the wave profile was developed.

TASK IV - Investigation of TASK III for oblique sea conditions.

### ANALYSIS PROCEDURE

The procedure utilized in the performance of this study relies heavily on the ABS/DAISY system of computer programs. A brief description of the system and the interface of the associated element programs are presented in Appendix A. The analysis comprises the following four steps:

#### 1. Selection of Record Intervals and Acquisition of the Wave and Full-Scale Stress Data

To compare the measured stress data with theoretical predictions for a ship at sea, it is essential to have reliable information about the actual wave environment. Two wave-measuring meters were installed on-board the SEA-LAND McLEAN, a Tucker wave meter and a micro-wave radar. A correlation and verification study [8] of the wave-meter data shows that the Tucker meter data is in error for the high ship speeds of interest, and the radar wave data appears to be more realistic. Since there was no meaningful stress data recorded with the ship in a "hove-to" condition or at "near zero" speed, the Tucker wave meter data could not be directly utilized for this study. Consequently, the wave environment was solely determined based on recorded signals of the wave radar, as presented by Dalzell [8]. However, it should be noted that the radar wave data has not yet been fully verified. Any possible errors incurred in the wave-measuring system or in the data reduction procedure would be essentially carried through the analysis and might cause stress deviations in the calculation.

The selection of record intervals was based on the availability of the measured stress data, the relative wave angle and the relative directions of the wave and the swell.

Regarding the full-scale stress data, the RMS values, stress spectra and stress-time histories of the midship average bending sensor (LVB) are presented in Reference [8] for all the selected record intervals at sea. For other strain gages, the required stress data was reduced by Teledyne.

For Task I (dockside calibration), all the test conditions were taken into consideration. The detailed information about loading conditions, gage locations and the measured data are presented in Reference [6].

## 2. Approximation of the Wave Environment and Prediction of Wave Loads

For each selected record interval, the wave environment was approximated either by an equivalent regular wave or a wave-spectrum approach, depending upon the methods utilized for each individual task. Details are discussed under Tasks II and III.

Once the wave environment was determined, the ABS/SHIPMOTION and ABS/DYNPRE computer programs were used to predict ship motions, wave loads and hydrodynamic pressures for input to the finite element structural model. The ABS/SHIPMOTION program is a revised version of SCORES. ABS/DYNPRE which is an extended subroutine of the SHIPMOTION program calculates the hydrodynamic pressure distribution on the wet surface of a ship's hull.

As mentioned in the introduction, the revised version of SCORES, modified with a speed correction factor, has been validated with a model test. This modified version is not available to the public. In the present study, the ABS version of the SCORES program (designated ABS/SHIPMOTION) was utilized for wave-load predictions. A speed correction factor similar to that introduced by Oceanics has been incorporated into the ABS version. A sample comparison of the RAO with the model test data is shown in Figures A-2 and A-3 of Appendix A, which shows that the comparison of the RAO for the vertical bending moments at midship is very good. However, the calculated RAO for the lateral bending moment at midship are generally less than those reduced from model experiments.

## 3. Generation of Finite Element Structural Models and Calculation of Structural Responses

In calculating the hull-girder responses, the entire ship was first represented by a three-dimensional coarse-mesh finite element model. Subsequently, fine-mesh three-dimensional models were utilized to determine the stress distribution at selected strain-gage locations. The ABS/DAISY computer program was used for this step of the analysis. The structural models and the boundary conditions used in the analysis are discussed in the following section.

#### 4. Analysis of Calculated Results and Comparison with Measured Data

After the completion of the fine-mesh DAISY runs for each task, a selective output containing strains and stresses for the selected strain gage locations was printed out. The analysis of the calculated results varies task by task. The detailed analysis methods and the comparison with the measured data are discussed separately under each individual task.

#### FINITE ELEMENT STRUCTURAL MODELS

The SL-7 is an 880'-6" x 105'-6" x 68'-6" twin-screw containership with a displacement of 50,315 long tons at 34'-0" draft. A general arrangement is shown in Figure 1.

The structural analysis for the SL-7 Containership was performed using the ABS/DAISY system of finite element computer programs (see Appendix A) for one coarse-mesh model comprising the entire ship and six fine-mesh models representing detailed structures of selected strain gage regions.

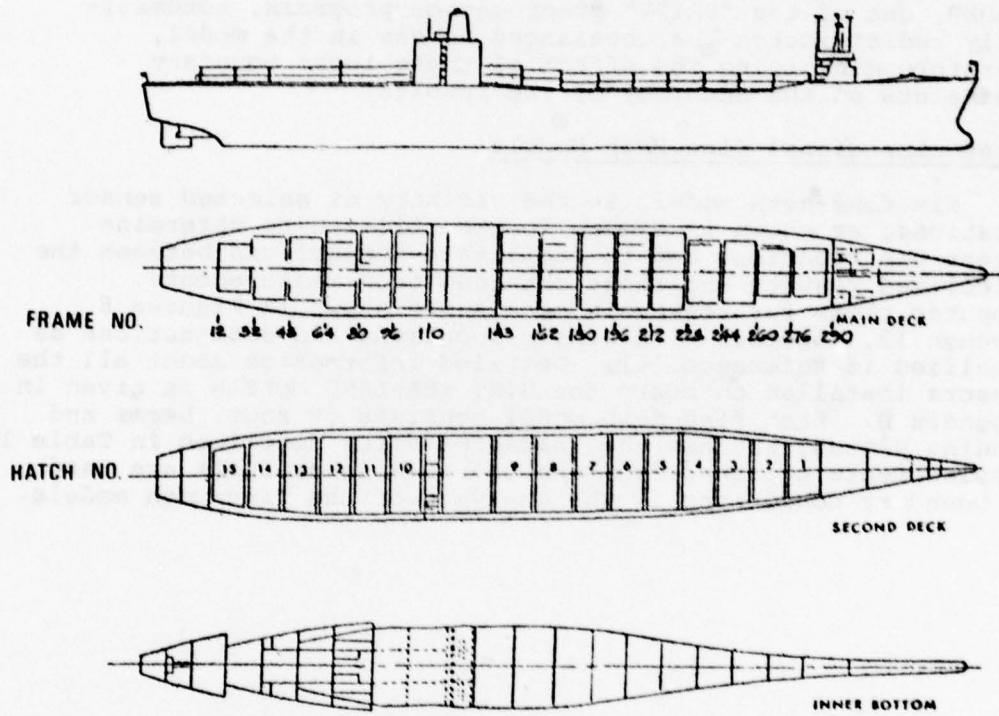


FIGURE 1 - SL-7 GENERAL ARRANGEMENT (FROM REFERENCE [3])

#### Three-dimensional Coarse-Mesh Model

The coarse-mesh model consists of 2602 nodal points and 9866 degrees of freedom. The major structural members are represented by rods, beams, membrane plates and bending plates,

with a total of 7122 elements. The model comprised only the port side of the vessel because of structural symmetry about the centerline plane. A three-dimensional isometric plot of this model is shown in Figure 2. Computer plots of the decks, bottom, side shell and the centerline profile are shown in Figure 3. Typical transverse bulkheads and web frames in the model are shown in Figure 4. For asymmetric loadings, the loads are divided into symmetric and anti-symmetric loading conditions, as shown in Figure 5, with appropriate boundary restraints at the centerline plane, as shown in Figure 6. In order to prevent rigid body movements due to the possible unbalanced forces on the model, certain additional boundary restraints must be imposed on the model, as shown in Figure 6. LOADER, one of the "DAISY" preprocessor programs, automatically redistributes the unbalanced forces in the model, therefore minimizing the effect of these local boundary restraints on the accuracy of the results.

#### Three-dimensional Fine-Mesh Models

Six fine-mesh models in the vicinity of selected sensor locations, as shown in Figure 7, are utilized to determine stress distributions and to facilitate comparisons between the calculated results and those obtained from measurements. Computer plots for these six models are shown in Figures 8 through 13, together with sensor locations and designations as specified in Reference [6]. Detailed information about all the sensors installed on board the S.S. SEA-LAND MCLEAN is given in Appendix B. Each fine-mesh model consists of rods, beams and bending plates, and has the characteristics described in Table 1. Displacements obtained from the coarse-mesh analysis are used as boundary conditions in the analysis of the fine-mesh models.

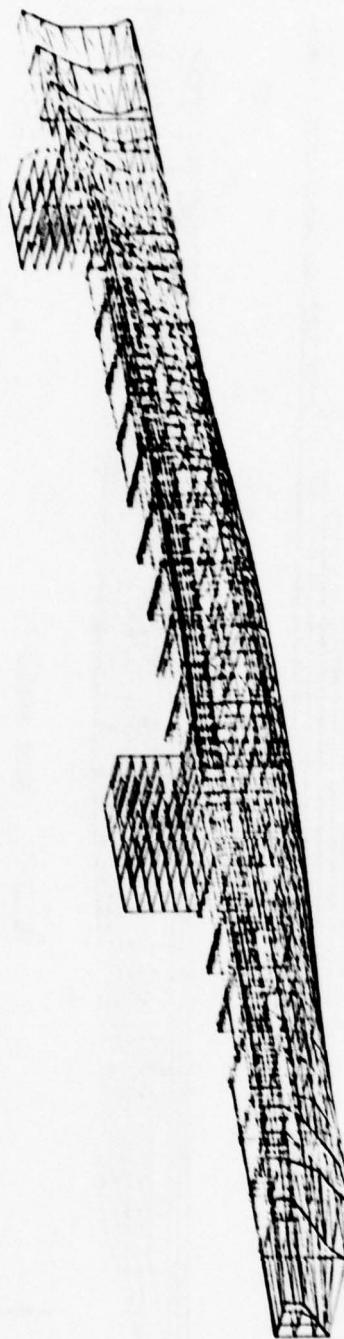


FIGURE 2 - ISOMETRIC PLOT OF FINITE ELEMENT COARSE-MESH MODEL

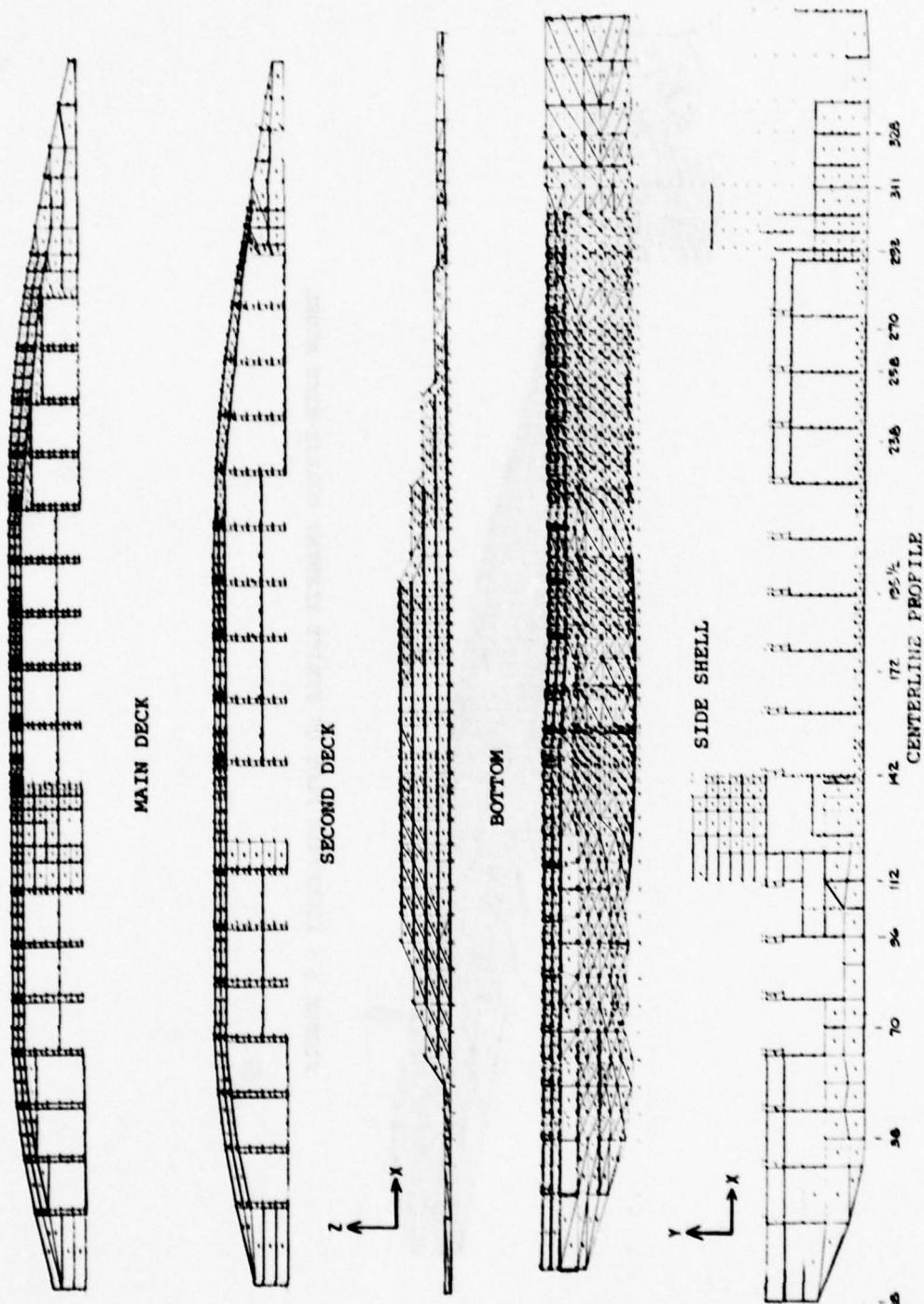


FIGURE 3 - PROJECTIONS OF FINITE ELEMENT COARSE-MESH MODEL

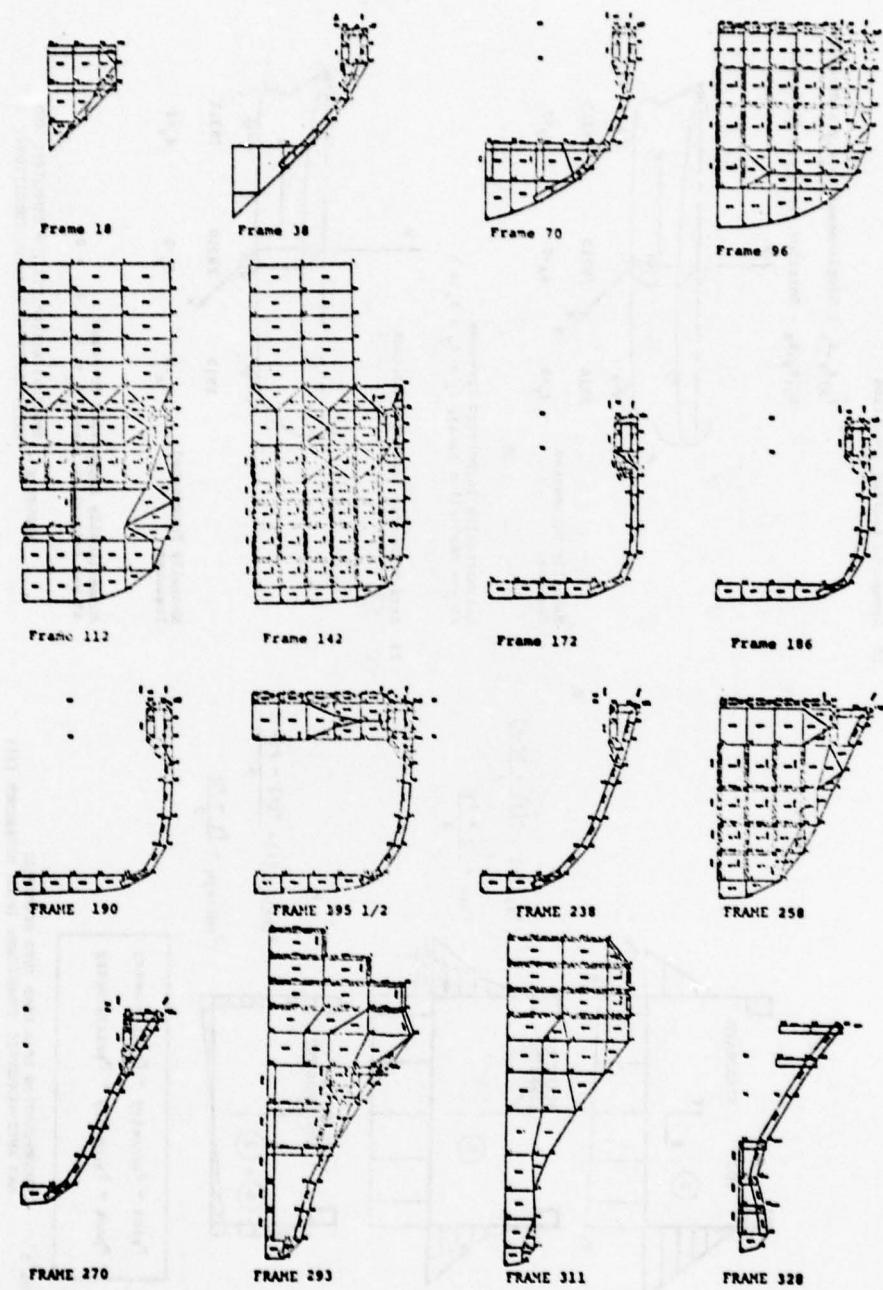
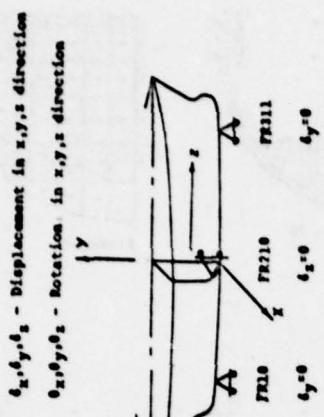
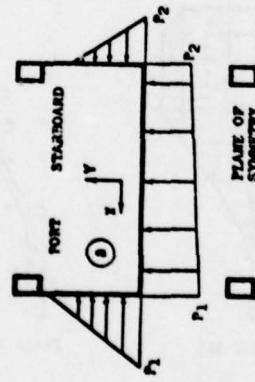


FIGURE 4 - TRANSVERSE FRAMES AND BULKHEADS

1.1 Symmetric Loading Condition



$u_x, u_y, u_z$  - Displacement in x,y,z direction  
 $\theta_x, \theta_y, \theta_z$  - Rotation in x,y,z direction

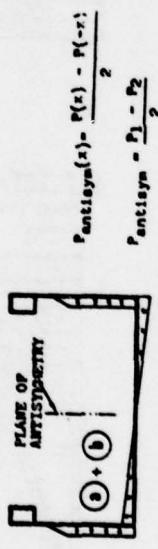


$u_x, u_y, u_z$  - Displacement in x,y,z direction  
 $\theta_x, \theta_y, \theta_z$  - Rotation in x,y,z direction

$$P_{sym}(x) = \frac{P(x) + P(-x)}{2}$$

$$P_{sym} = \frac{P_1 + P_2}{2}$$

2) Anti-Symmetric Loading Condition



$u_x, u_y, u_z$  - Displacement in x,y,z direction  
 $\theta_x, \theta_y, \theta_z$  - Rotation in x,y,z direction

$$P_{antisym}(x) = \frac{P(x) - P(-x)}{2}$$

$$P_{antisym} = \frac{P_1 - P_2}{2}$$

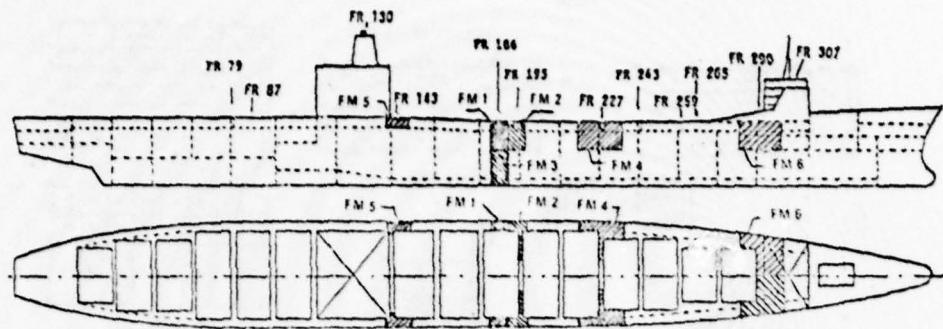
**Plane of Antisymmetry**

$P_{sym} = P_{symtry} - P_{antisymtry}$

$P_{antisym} = P_{symtry} + P_{antisymtry}$

FIGURE 5 - DECOMPOSITION OF A LOAD INTO SYMMETRIC AND ANTI-SYMMETRIC COMPONENTS (FROM REFERENCE (1))

FIGURE 6 - BOUNDARY RESTRAINTS FOR SYMMETRIC AND ANTI-SYMMETRIC LOADING CONDITIONS



Fine-Mesh Models

FM1  
 FM2  
 FM3  
 FM4  
 FM5  
 FM6

FIGURE 7 - LOCATION OF FINE-MESH MODELS

TABLE 1 CHARACTERISTICS OF FINE MESH FINITE ELEMENT MODELS

Model Designation	Model Location	Number of Nodes	Number of Elements	Number of Degrees of Freedom
FM1	FR.182-190 (above str.no.1)	314	602	1507
FM2	FR.190-198	432	802	2030
FM3	FR.182-190 (below str.no.1)	391	536	1565
FM4	FR.218-238	318	680	1764
FM5	FR.140-150	326	522	1350
FM6	FR.282-298	393	541	1405

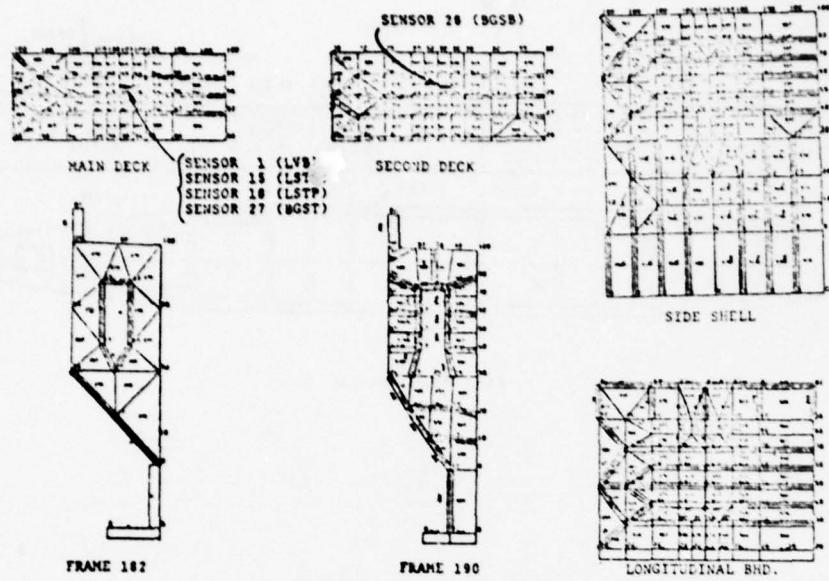


FIGURE 8 - FINE MESH MODEL 1 (FRAMES 182 - 190, ABOVE STRINGER NO. 1)

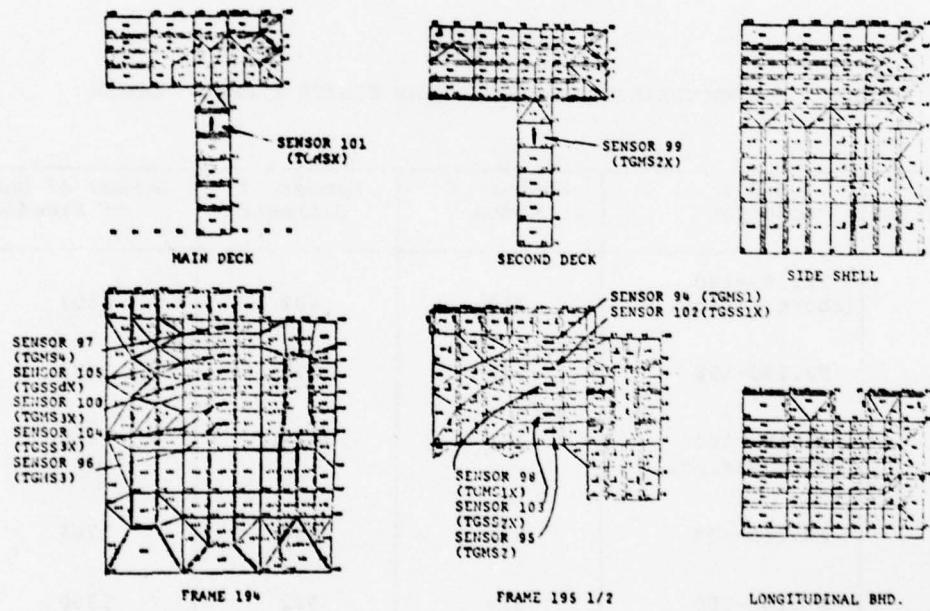


FIGURE 9 - FINE MESH MODELS 2 (FRAMES 190 - 198, ABOVE STRINGER NO. 1)

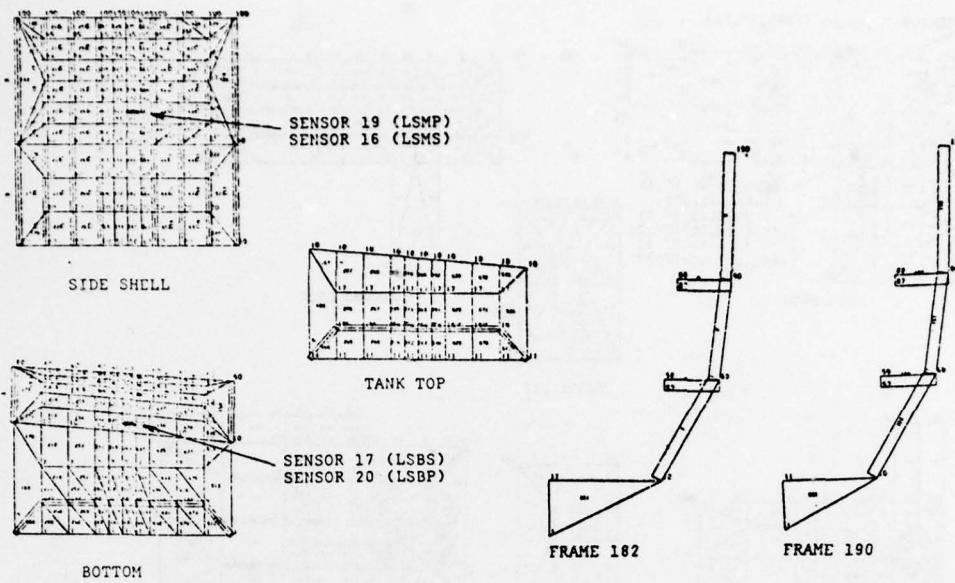


FIGURE 10 - FINE MESH MODEL 3 (FRAMES 182 - 190, BELOW STRINGER NO. 1)

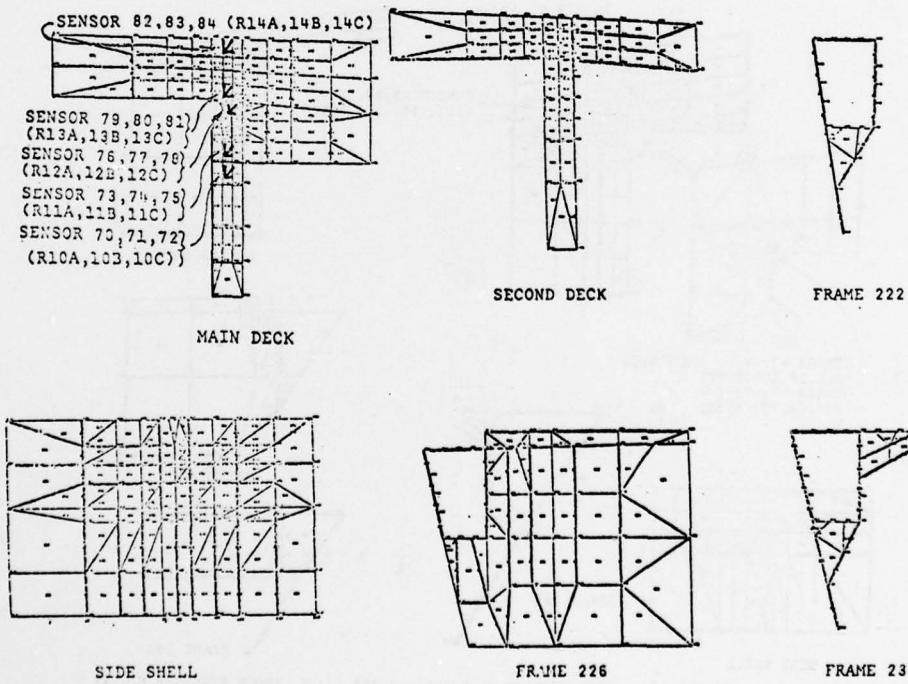


FIGURE 11 - FINE MESH MODEL 4 (FRAMES 218 - 238, ABOVE STRINGER NO. 1)

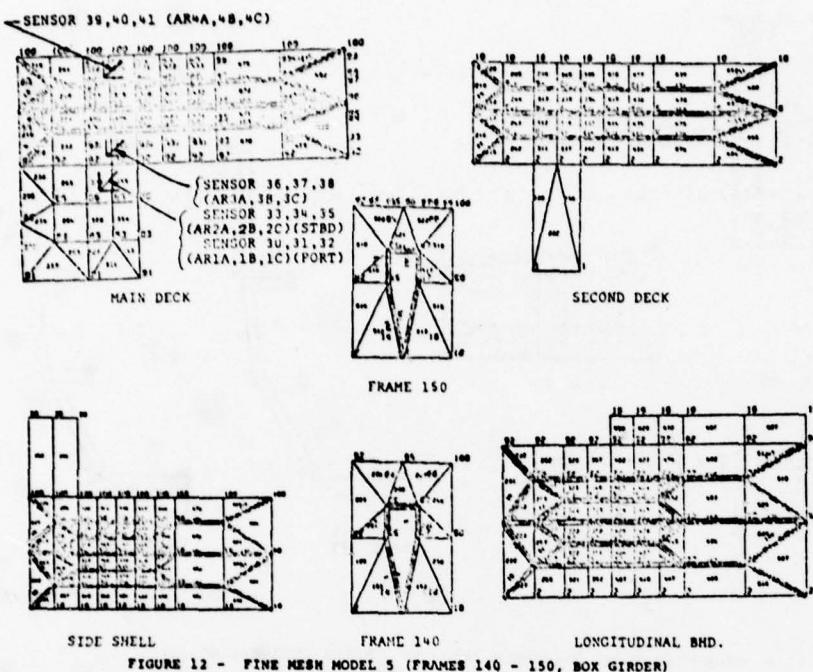


FIGURE 12 - FINE MESH MODEL 5 (FRAMES 140 - 150, BOX GIRDERS)

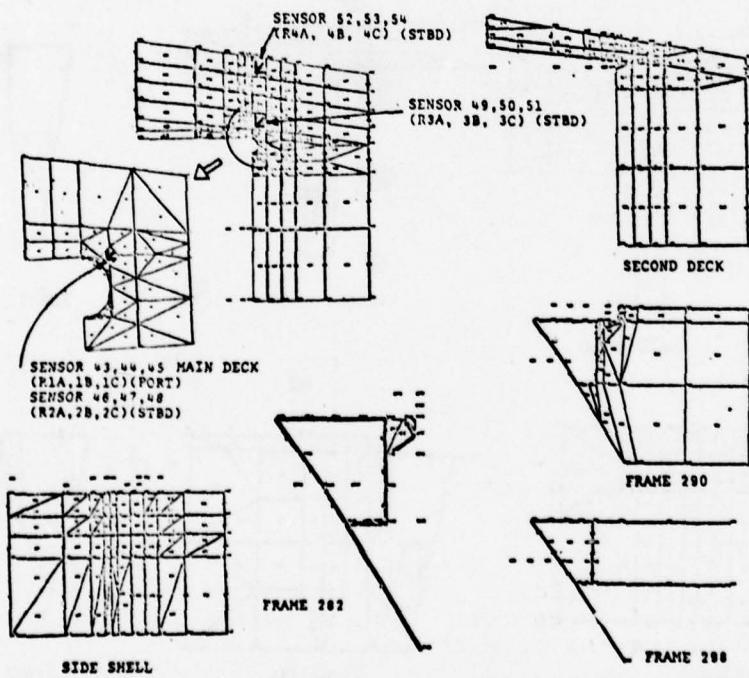


FIGURE 13 - FINE MESH MODEL 6 (FRAMES 282 - 298, ABOVE STRINGER NO. 1)

TASK I - DOCKSIDE CALIBRATION1. Loading Conditions

The following six calibration loading conditions have been analyzed:

Condition 1

Dockside initial loading condition, with all cargo hold and deck containers, except holds beneath Hatches 3, 10 and 14.

Condition 3

Deck containers removed from Hatches 1 through 4 and 12 through 15.

Condition 4

Remaining deck containers on Hatches 5 through 11 removed.

Condition 5

Approximately one-half of containers removed from starboard side of Hatches 1 through 7 and from the port side of Hatches 8 through 15, generating a torsional moment. Hatch covers placed asymmetrically to contribute to the torsional moment.

Condition 6

Completion of unloading described in Condition 5. This represents the maximum torsional load.

Condition 7

Nominally empty ship except for one propeller (47 long tons) loaded into Hatch 3 and one propeller in Hatch 4, all hatch covers on.

Loading conditions 1, 3, 4 and 7 are symmetric about the centerline plane; loading conditions 5 and 6 are asymmetric and include torsional loadings.

Note: The designated loading cases are identical with those used in Reference [6]. Loading condition 2 was not used during the calibration test.

The "SHIPMOM" program was used to calculate static bending moments and to generate hydrostatic pressures for the DAISY model. These SHIPMOM results were compared with the results of a previous less refined SHIPMOM computer run with fewer stations, described in Reference [6]. The comparison indicated very little difference between the two runs.

## 2. Comparison of Calculated Stresses and Measured Data - Longitudinal Stresses

The comparison of stresses measured by strain gages and those calculated at corresponding gage locations is presented in Tables 2 and 3. In Table 2 the comparison was made by subtracting loading condition 1 from each loading condition, taking loading condition 1 as a datum loading. Similarly, Table 3 takes loading condition 4 as the datum loading.

A preliminary investigation of the experimental and analytical stress results indicates some instances of agreement, and some instances of disagreement. The correlation of full-scale measurements and computed results can only be established after carefully verifying the data and the environmental conditions.

The experimental data for midship sensors 1 (LVB), 15 (LSTS) and 18 (LSTP), where LVB measures the average longitudinal vertical bending stress of port and starboard, LSTP and LSTS measure the longitudinal stress at top, port and starboard respectively, show some discrepancies.

Sensor number 1 should average the values of the longitudinal stress components of the LSTS and LSTP sensors. However, differences in the comparative stress values appear with the change in the reference of datum loading condition, or the so-called zero-stress reference case. Table 4 illustrates these differences when referring loading condition 7 to loading conditions 1, 3 or 4.

Loading condition 1 represents no change in mechanical loading of the ship from the initial calibration condition (zero reading for the strain gages) that took place when the ship was navigating through the Maas river, but the measured stresses range between -2677 and 1654 psi (Reference [6]).

Loading condition 7 is predominantly a symmetric loading case, but the measured stresses from the symmetrically located sensors 15 and 18 indicate asymmetric response, especially in loading condition 7-4.

TABLE 2 COMPARISON OF CALCULATED AND MEASURED STRESSES (PSI)  
USING LOADING CONDITION 1 AS DATUM LOADING

Sensor	Loading Cond.	(3-1)		(4-1)		(5-1)		(6-1)		(7-1)	
		Calc.	Meas.								
FM 1	1(LVB)	-1058	-1148	104	-133	890	574	1352	883	2746	2119
	15(LSTS)	-1058	-1756	104	-1486	791	-495	1103	-180	2746	2252
	18(LSTP)	-1058	-1369	104	-88	988	782	1600	1286	2746	1789
	94(TGMS1)	75	467	437	2451	-940	409	-1622	-583	435	2568
	95(TGMS2)	142	993	-112	386	-1399	-3033	-1988	-2041	71	-1600
	96(TGMS3)	119	228	471	-684	1569	-342	2191	114	561	-1483
	97(TGMS4)	-107	396	-9	1016	1803	4230	2795	5245	753	1974
	98(TGMS1X)	237	1322	125	680	-1191	-1421	-1833	-1888	-291	38
	99(TGMS2X)	43	1225	115	-429	88	-621	148	-1532	601	-484
	100(TGMS3X)	170	376	109	-365	1189	-822	1803	-365	-24	-993
FM 2	101(TGMS4X)	-123	-1882	246	248	621	1037	863	1150	984	1094
	17(LSBS)	531	1785	16	1648	-236	961	-267	1099	-1146	183
	20(LSBP)	531	589	16	453	-453	272	-802	589	-1146	226
FM 3	73(R11A)	-85	-115	102	-343	59	-115	-10	399	486	798
	76(R12A)	-133	142	78	-77	402	687	515	687	671	796
	79(R13A)	-525	-1096	341	76	1872	2653	2420	3257	2865	3141
	82(R14A)	-454	-1864	301	-1032	1061	-307	1300	-195	2391	864
FM 4	30(AR1A)	-719	-593	-277	-209	1033	1217	1476	1491	507	-758
	33(AR2A)	-719	152	-277	1592	-975	-592	-1297	-1125	507	-334
	36(AR3A)	-1362	-1084	-471	1882	-1425	-513	-1884	-1939	1003	912
	39(AR4A)	-1212	-1964	-459	-982	171	1363	266	1145	892	600
FM 5	43(R1A)	85	-937	170	-334	-568	-1870	-1136	-2418	710	-1102
	46(R2A)	85	-725	170	-1562	1150	390	1974	948	710	335
	49(R3A)	98	-513	202	-1825	835	0	1386	343	814	199
	52(R4A)	111	-163	230	-382	-57	-1145	-246	-1254	902	-436
FM 6											

TABLE 3 COMPARISON OF CALCULATED AND MEASURED STRESSES (PSI)  
USING LOADING CONDITION 4 AS DATUM LOADING

Sensor	Loading Cond.	(5-4)		(6-4)		(7-4)	
		Calc.	Meas.	Calc.	Meas.	Calc.	Meas.
FM 1	1 (LVB)	786	707	1249	1016	2643	2252
	15 (LSTS)	687	-991	1000	1306	2643	3738
	18 (LSTP)	885	870	1497	1374	2643	1877
FM 2	94 (TGMS1)	-1377	-2042	-2059	-3034	-3	117
	95 (TGMS2)	-1287	-3419	-1876	-2427	183	-1986
	96 (TGMS3)	1098	342	1720	798	89	-799
	97 (TGMS4)	1812	3214	2803	4229	761	958
	98 (TGMS1X)	-1316	-2101	-1958	-2568	-416	-642
	99 (TGMS2X)	-27	-192	33	-1103	486	-55
	100 (TGMS3X)	1079	-457	1694	0	-133	-628
	101 (TGMS4X)	375	789	617	902	738	846
	17 (LSBS)	-251	-687	-283	-549	-1162	-1465
FM 3	20 (LSBP)	-469	-181	-818	136	-1162	-227
	73 (R11A)	-43	228	-112	742	384	1141
	76 (R12A)	324	764	437	764	593	873
FM 4	79 (R13A)	1531	2577	2079	3181	2524	3071
	82 (R14A)	760	-1339	999	837	2089	1896
	30 (AR1A)	1310	1426	1753	1700	784	-549
FM 5	33 (AR2A)	-698	-2184	-1020	-2717	784	-1926
	36 (AR3A)	-953	-2395	-1412	-3821	1475	-970
	39 (AR4A)	623	2345	718	2127	1350	1582
FM 6	43 (R1A)	-738	-1536	-1306	-2084	540	-768
	46 (R2A)	980	1952	1803	2510	540	1898
	49 (R3A)	633	1825	1184	2168	612	2024
	52 (R4A)	-270	-763	-462	-872	672	-54

It can be seen that the degree of agreement between the measured and calculated results varies significantly from Table 2 to Table 3. This indicates some inconsistencies of the measured data between different loading conditions, probably due to changes in ambient temperatures.

According to Table IV of Reference [6], a temperature difference of  $15^{\circ}$  F between port and starboard sides was recorded during the calibration test. A temperature difference of  $21^{\circ}$  F between the deck and sea water was also recorded. With these magnitudes of temperature gradient, the thermal stress could be as high as 1500 psi. However, due to the lack of sufficient information about the temperature distribution during the calibration test, it is impossible to incorporate thermal stresses into the analysis.

To evaluate the effects of these temperature changes, longitudinal stresses measured at 17 selected sensors are shown in Figures 14 through 16. The sensors are described in Table 5.

Figure 14 shows the longitudinal stresses for calibration loading conditions 3 through 7, utilizing loading condition 1 as the datum loading. The deviations of the measured stresses from the calculated values are generally within a bandwidth of  $\pm 1,500$  psi. Based on the temperature differential recorded during the calibration test, shown in Table 6, the maximum thermal stress may be as high as 1,500 psi. With this thermal stress margin in mind, the overall comparison of the measured and computed stresses shown in Figure 14 is reasonably good.

In an attempt to minimize the possible thermal effects, changes in stresses between two loading conditions with the least temperature differential were also examined. The results of loading case 4-3, which represents pure vertical bending, are shown in Figure 15. With a few exceptions, the stress deviations fall within a bandwidth of  $\pm 400$  psi. This magnitude of deviation is regarded as acceptable, considering the sensibility and reliability of strain-gage readings. The results of loading case 7-6, which reflects both vertical bending and torsion, are shown in Figure 16. With the exception of Sensors 30 and 39, the stress comparison is generally good.

To minimize the possible thermal effects, it is advisable, for the future calibration tests, that other means of loading be used to create appreciable mechanical strains and that complete steel temperature data of the deck and side shell (port and starboard) be recorded.

TABLE 4 MEASUREMENTS OF LONGITUDINAL STRESSES (PSI) IN MIDSHIP SENSORS

Sensor No.	Loading Condition	(7-1)	(7-3)	(7-4)
15(LSTS)		2252	4008	3738
18(LSTP)		1789	3158	1877
Average of 15 and 18		2020.5	3583	2807.5
1(LVB)		2119	3267	2252

TABLE 5 SELECTED SENSOR LIST

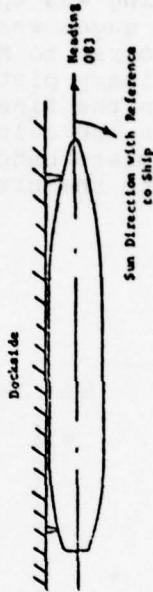
SENSOR NUMBER	SENSOR NOMEN.	SIGNAL NOMENCLATURE
1	LVB	Longitudinal Vertical Bending
15	LSTS	Longitudinal Stress Top Starboard
18	LSTP	Longitudinal Stress Top Port
17	LSBS	Longitudinal Stress Bottom Starboard
20	LSBP	Longitudinal Stress Bottom Port
73	R11A	
76	R12A	R = Foreward Rosettes on the
79	R13A	Main Deck, See Figure 11
82	R14A	
30	AR1A	
33	AR2A	AR = Aft Rosettes on the Main
36	AR3A	Deck, See Figure 12
39	AR4A	
43	R1A	
46	R2A	R = Forward Rosettes on the
49	R3A	Main Deck, See Figure 13
52	R4A	

TABLE 6 ENVIRONMENTAL CONDITIONS AT CALIBRATION  
(FROM REFERENCE [6])

Smd.	Time	Temperatures, °F (a)			Location of Sun, degrees Elevation Azimuth (b)		Wind Speed, mph	Wind Direction
		Air, Dry	Air, Wet	Water	Port Tunnel	Star Tunnel		
1	0300 9 Apr '73	49.5	43	43	51	52	43 (Overcast)	50 Stbd 60° Port
2	1000	58	50.5	43	69	64	60 (Clear)	100 Stbd 60° Port
4	1725	49	44	43	49	61	30 (Clear)	Alt 110° Port
5	2120	38	36	43	45	52	-	90° Port
6	0105 10 Apr '73	36.5	35	43	43	46	-	60° Port
7	0830	40	-	42	40	46	-	90° Port

Notes:

- a Measured on hull plating backside  
b Relative to ship



Another uncertainty in comparing the measured and calculated stresses is the influence of plate unfairness and local plate bendings. So far as the longitudinal stresses in the hull structures are concerned, this kind of influence is considered insignificant. All the longitudinal strain gages were installed on very thick plates where the effect of unfairness, if any, would not be generally noticeable. Where local plate bending was apparent, such as the bottom hull plating, the strain gages were located at the quarter span between frames (floors) to minimize local plate bending. In addition, all the primary plating was represented by bending plate elements in the fine-mesh models. Panel bendings are also included in the calculation. Consequently, the plate unfairness and local plate bending should not be regarded as an influential parameter to the stress discrepancies in this case.

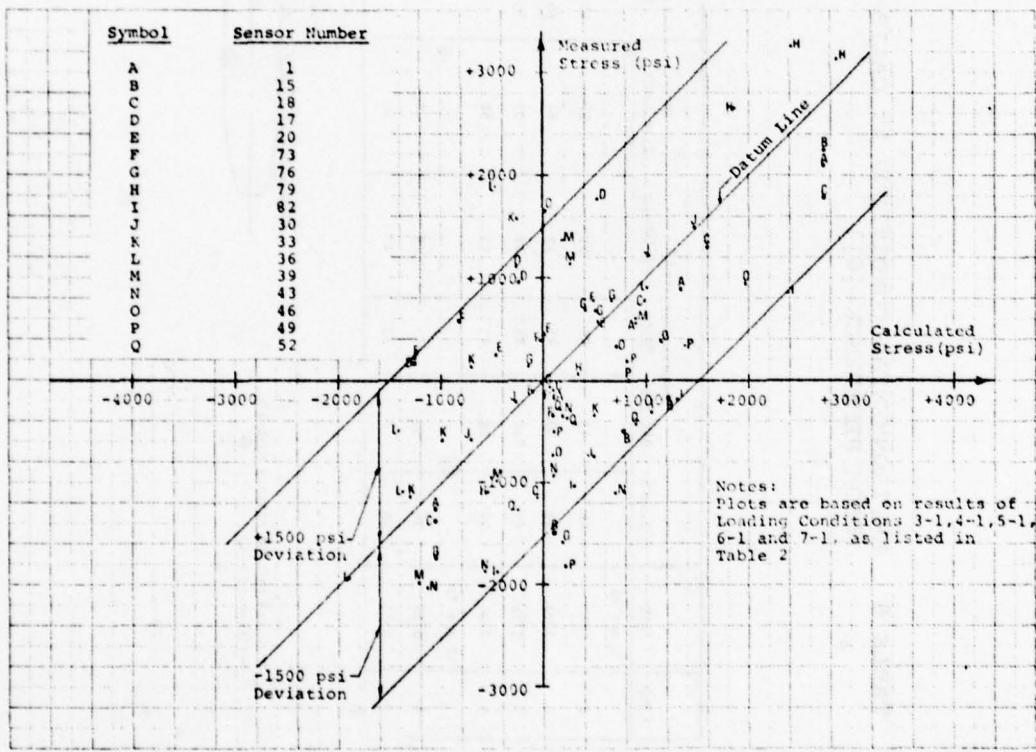


FIGURE 14 - DEVIATION OF MEASURED LONGITUDINAL STRESSES FROM CALCULATED VALUES

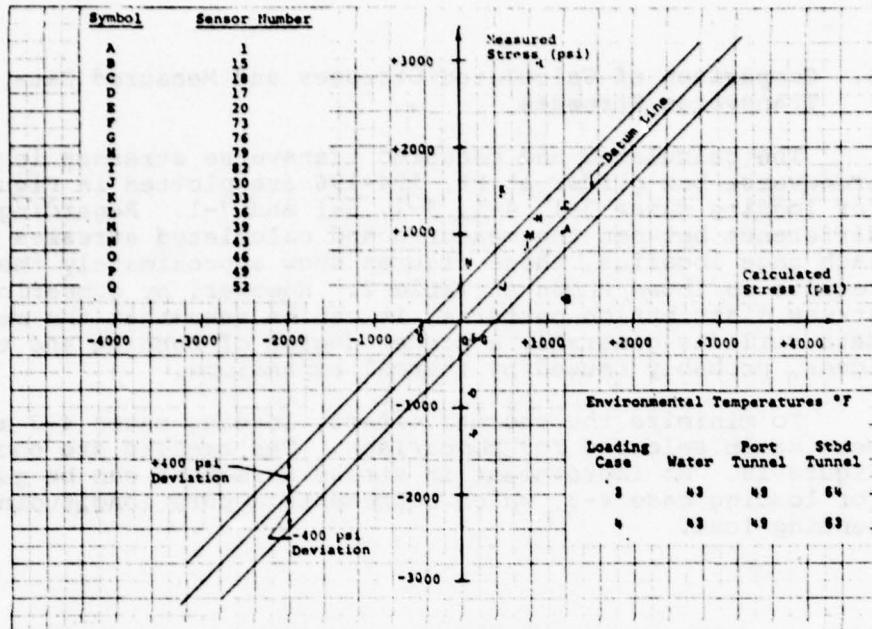


FIGURE 15 - DEVIATION OF MEASURED LONGITUDINAL STRESSES FROM CALCULATED VALUES  
PLOTS ARE BASED ON LOADING CONDITION 4 MINUS 3

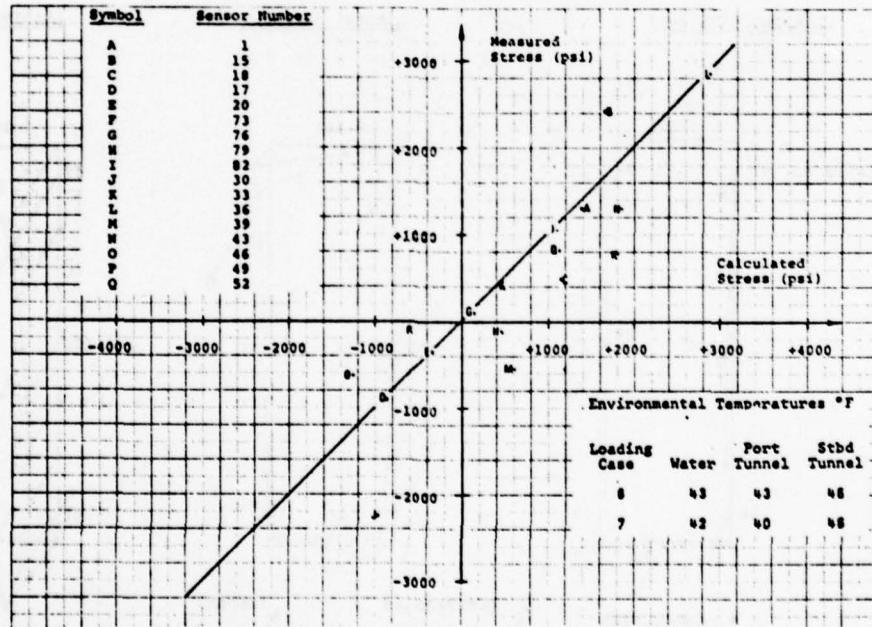


FIGURE 16 - DEVIATION OF MEASURED LONGITUDINAL STRESSES FROM CALCULATED VALUES.  
PLOTS ARE BASED ON LOADING CONDITION 7 MINUS 6

### 3. Comparison of Calculated Stresses and Measured Data - Transverse Stresses

The calculated and measured transverse stresses in the transverse box girder at Fr. 194-196 are plotted in Figure 17 for loading cases 3-1, 4-1, 5-1, 6-1 and 7-1. Regarding the difference between the measured and calculated stresses at each gage location, these figures show approximately the same results as those shown in Table 2. However, by comparing the stress distribution patterns, it can be seen that the measured data reflects generally a higher degree of bending and torsion loads, probably caused by thermal expansions.

To minimize the thermal effect, loading cases 4-3 and 7-6 were again selected for comparison. The results are shown in Figure 18. No improvement in stress agreement can be gained for loading case 4-3, which represents a pure longitudinal bending load.

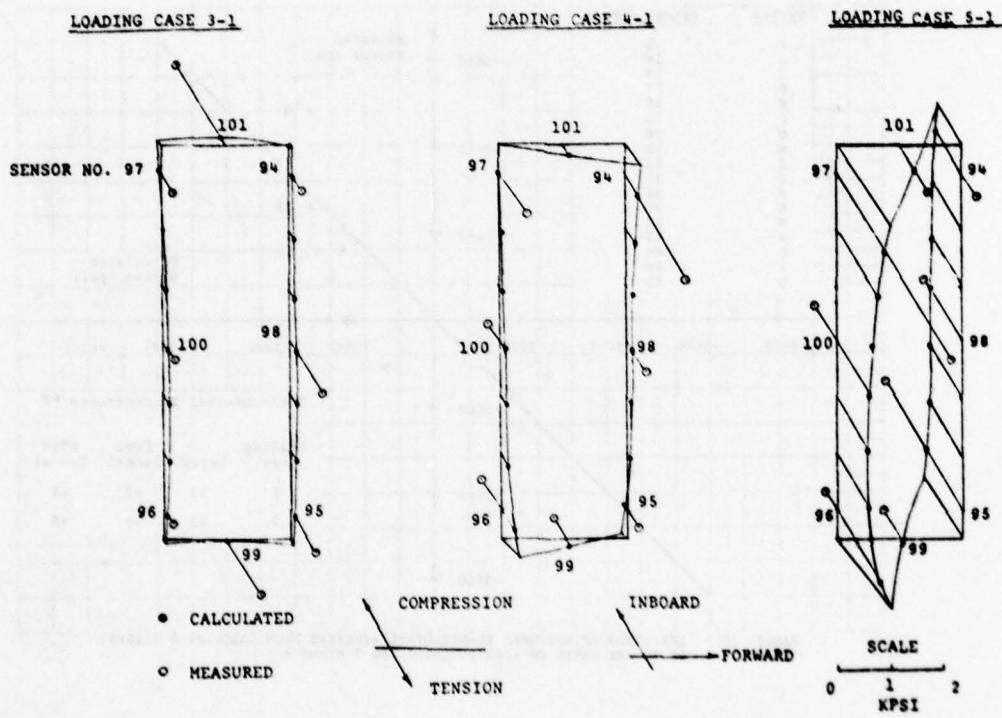


FIGURE 17 - COMPARISON OF CALCULATED AND MEASURED TRANSVERSE STRESSES IN THE TRANSVERSE BOX-GIRDER AT FR. 194-196

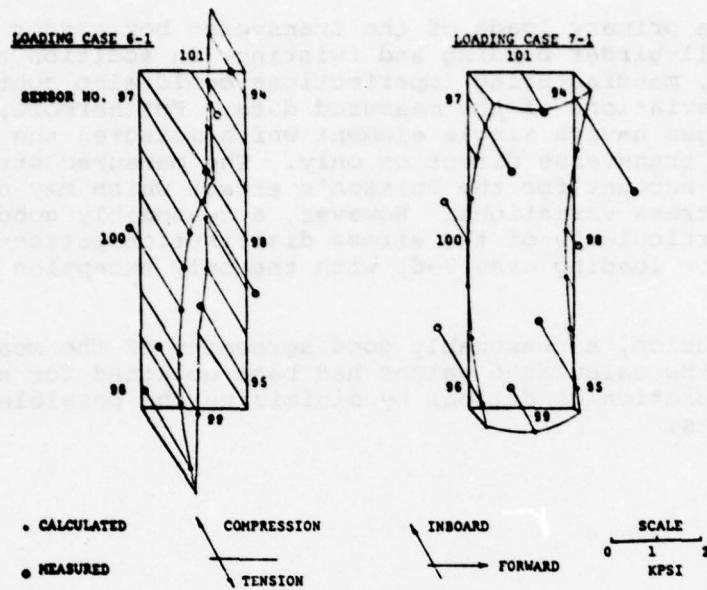


FIGURE 17 - COMPARISON OF CALCULATED AND MEASURED TRANSVERSE STRESSES IN THE TRANSVERSE BOX-GIRDER AT FR. 194-196 (CONT'D)

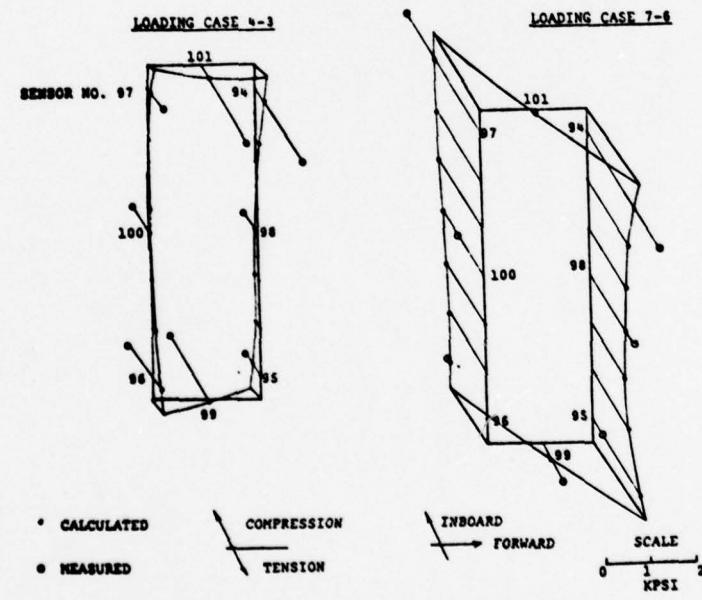


FIGURE 18 - COMPARISON OF CALCULATED AND MEASURED TRANSVERSE STRESSES IN THE TRANSVERSE BOX-GIRDER AT FR. 194-196

Since the primary loads of the transverse box girder are induced by hull-girder bending and twisting, in addition to thermal loads, manufacturing imperfections could also contribute to the deviations of the measured data. Furthermore, the strain gages have a single element which measures the strain in the transverse direction only. The measured stress data does not account for the Poisson's effect which may cause significant stress variations. However, a reasonably good agreement, particularly of the stress distribution patterns, can be seen for loading case 7-6, with the only exception of Sensor 95.

In conclusion, a reasonably good agreement of the measured stresses and the calculated values has been obtained for several selected calibration conditions by minimizing the possible thermal effects.

## TASK II - COMPARISON OF RMS STRESSES IN HEAD SEAS

For Task II, a stress comparison between the calculated and measured RMS values in head seas was carried out by an equivalent regular wave approach and by a spectrum analysis, based on a study of the relationship between computed wave-induced stresses and wave heights.

In order to examine this relationship, a regular wave with a length of 808.5 feet and with its crest at the midship was selected for ship motion and stress calculations. Three different wave heights as shown in Table 7 were taken into consideration, with a ship speed of 10.9 knots in head seas.

TABLE 7 WAVE CHARACTERISTICS AND SHIP MOTION DATA - LINEARITY STUDY

Wave Cond.	(a) Loading Cond.	Ship Speed (knots)	Wave Length (ft.)	Wave Height (Peak-to-Trough) (ft.)	(b) Heave (ft.)	(c) Pitch (deg.)	Location of Wave Crest Forward From A.P. (ft.)
1	8	10.9	808.5	3.28	0.62	0.32	440.0
2	9	10.9	808.5	9.84	1.85	0.97	440.0
3	10	10.9	808.5	20.92	3.93	2.06	440.0

Notes:

- a All conditions consist of full cargo loads and head waves
- b Heave is positive down
- c Pitch is positive bow up

The ship's motions and accelerations which are required as part of the finite element model input were calculated using the SHIPMOTION computer program. A set of quasi-static pressures (still-water plus wave profile) and inertia forces corresponding to the calculated accelerations were applied on the finite element model and a correction was made to ensure that the applied vertical shearing forces and bending moments were generally identical to those obtained directly from the SHIPMOTION program.

The calculated results are shown in Figures 19, 20, and 21 at three selected locations along the length of the ship. As shown in Figure 19, the wave-induced longitudinal and transverse stresses in the main deck plating at the midship section are approximately in direct proportion to the wave height. They depart from linearity at stations remote from the midship section (Figures 20 and 21), even though the hull-girder shearing forces and bending moments vary linearly with wave heights, as predicted by a ship motion calculation.

The non-linearity is due to the combination of the local bending in the transverse direction and the wave-induced forces in the longitudinal direction.

Recent model experiments in towing tanks indicate that the motions and wave loads do not vary linearly with wave heights. A realistic trend cannot yet be defined due to the lack of sufficient experimental data. Based on the results shown in Figures 20 and 21, it can be seen that the hull-girder responses (wave-induced stresses) exhibit a non-linear pattern even with linear shear and bending loads.

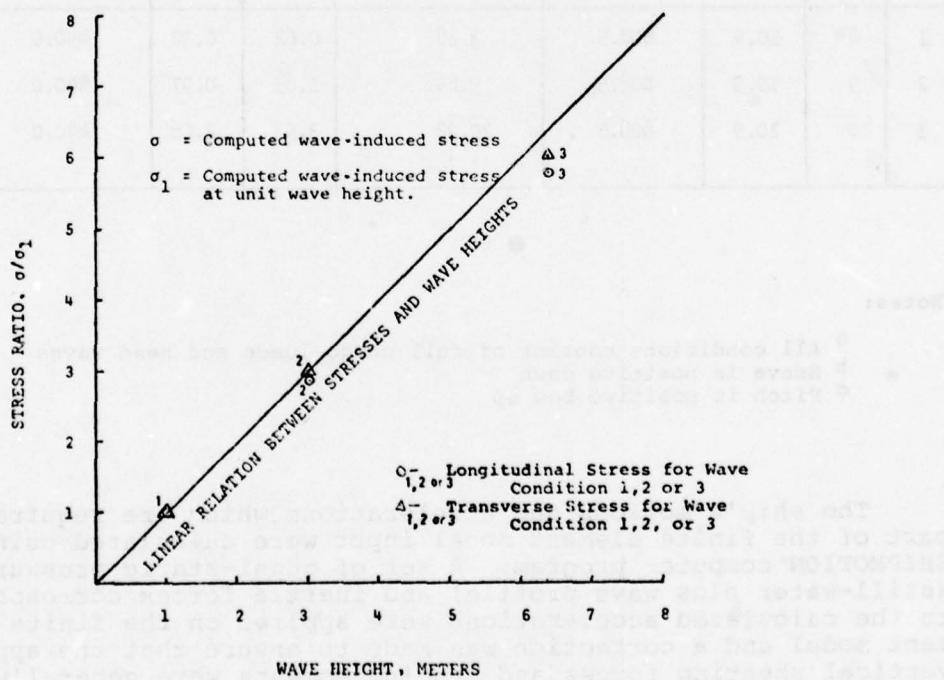


FIGURE 19 - WAVE-INDUCED STRESSES VS WAVE HEIGHTS -  
 MAIN DECK PLATING AT THE MIDSHP SECTION

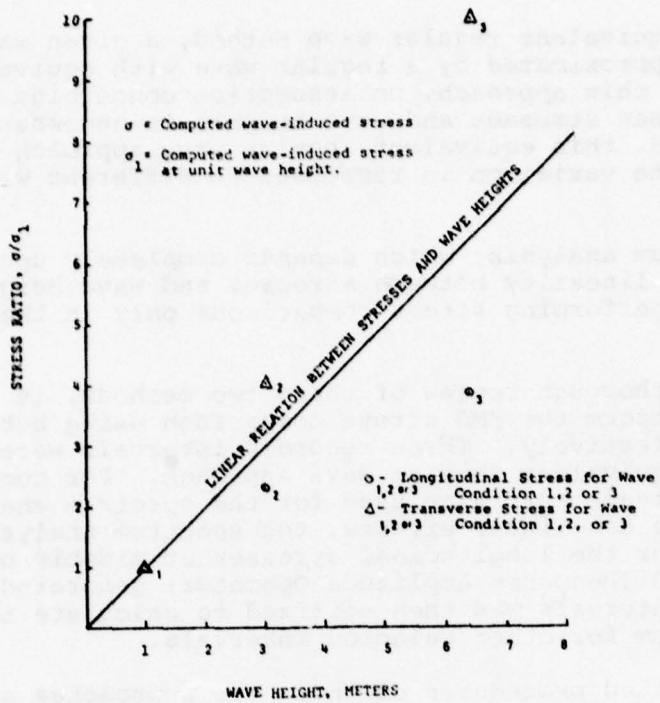


FIGURE 20 - WAVE-INDUCED STRESSES VS WAVE HEIGHTS -  
MAIN DECK PLATING BETWEEN FRAMES 226 and 228

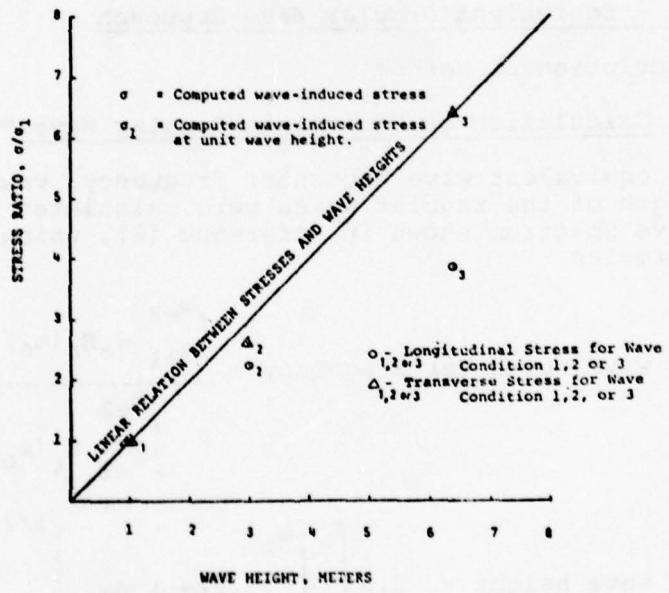


FIGURE 21 - WAVE-INDUCED STRESSES VS WAVE HEIGHTS -  
MAIN DECK PLATING AT FRAME 290

In the equivalent regular wave method, a given wave spectrum can be approximated by a regular wave with equivalent energy. With this approach, no assumption concerning the relation between stresses and wave heights is necessary. On the other hand, this equivalent regular wave approach will not account for the variation in responses to different wave frequencies.

A spectrum analysis, which depends completely upon an assumption of linearity between stresses and wave heights, is valid for performing stress comparisons only in the midship region.

After a thorough review of these two methods, it was decided to perform the RMS stress comparison using both approaches selectively. Three recorded intervals were selected for the equivalent regular wave approach. For comparison, the same intervals were also used for the spectrum analysis. Because of the non-linear effects, the spectrum analysis was carried out for the longitudinal stresses at midship only. The stress RAO (Response Amplitude Operator) generated for these three intervals was then utilized to calculate the stress spectrum for other selected intervals.

The detailed procedures of these two approaches and the computed results are discussed in the following sections.

### 1. Method 1 - Equivalent Regular Wave Approach

#### A. Description of Method

##### a. Calculation of Equivalent Regular Waves

The equivalent wave encounter frequency, wave height and wave length of the regular waves were calculated based on the radar wave spectrum shown in Reference [8], using the following formulas

$$\text{Equivalent wave encounter frequency} = \frac{\int_{\omega_{el}}^{\omega_{e2}} \omega_e S_{\zeta}(\omega_e) d\omega_e}{\int_{\omega_{el}}^{\omega_{e2}} S_{\zeta}(\omega_e) d\omega_e} \quad (1)$$

$$\text{Equivalent wave height} = 2.5 \left[ \int_{\omega_{el}}^{\omega_{e2}} S_{\zeta}(\omega_e) d\omega_e \right]^{1/2} \quad (2)$$

$$\text{Equivalent wave length} = \frac{2\pi g}{\omega^2} \quad (3)$$

$\omega$  is derived from the equation

$$\omega_e = \omega - \frac{u \cos \beta}{g} \omega^2 \quad (4)$$

where  $g$  = gravity constant

$\omega$  = Wave frequency

$\omega_e$  = encounter frequency, measured directly from the time history

$\beta$  = heading angle

$u$  = ship's speed

$s_\zeta(\omega_e)$  = wave spectrum in the frequency domain

$\omega_{e1}, \omega_{e2}$  = lower and upper bounds of frequency within the domain under consideration

Three wave conditions were selected for this task. The detailed information and the equivalent waves for these three wave conditions are shown in Table 8.

TABLE 8 WAVE CONDITIONS SELECTED FROM REFERENCE [8] FOR COMPARISON OF RMS STRESSES IN HEAD SEAS

Wave Condition	Tape	Index	Interval	Run No.	Ship Speed (knots)	Equivalent Wave Length (ft.)	Equivalent Wave Height (ft.)
4	145	18	5	405	10.9	808.5	20.92
5	145	24	29	429	18.7	808.5	21.97
6	145	29	50	450	28.1	561.5	16.47

### b. Calculation of Wave-Induced Loads

A computer subroutine was developed by ABS for the calculation of hydrodynamic pressures (ABS/DYNPRE) acting on the ship in a seaway. However, it was found that the wave-induced vertical bending moments obtained from an integration of the hydrodynamic pressures were not in full agreement with those obtained from the ABS/SHIMPOTION program. In order to eliminate such differences, the following two modifications were utilized to correct the vertical bending moments obtained by integration of the imposed pressures and inertial forces.

#### Method 1A - Quasi-static Pressures

A quasi-static pressure approach was used to calculate the input load for the ABS/DAISY runs. In this approach, the quasi-static pressures and the inertia forces at the instantaneous ship position as determined by the ship motion calculation were first applied to the structural model. Then, the bottom pressures were modified to make the vertical wave bending moments comparable with those obtained from the ship motion program. This was the procedure used in the determination of the relationship between computed wave-induced stresses and wave heights described at the beginning of Task II.

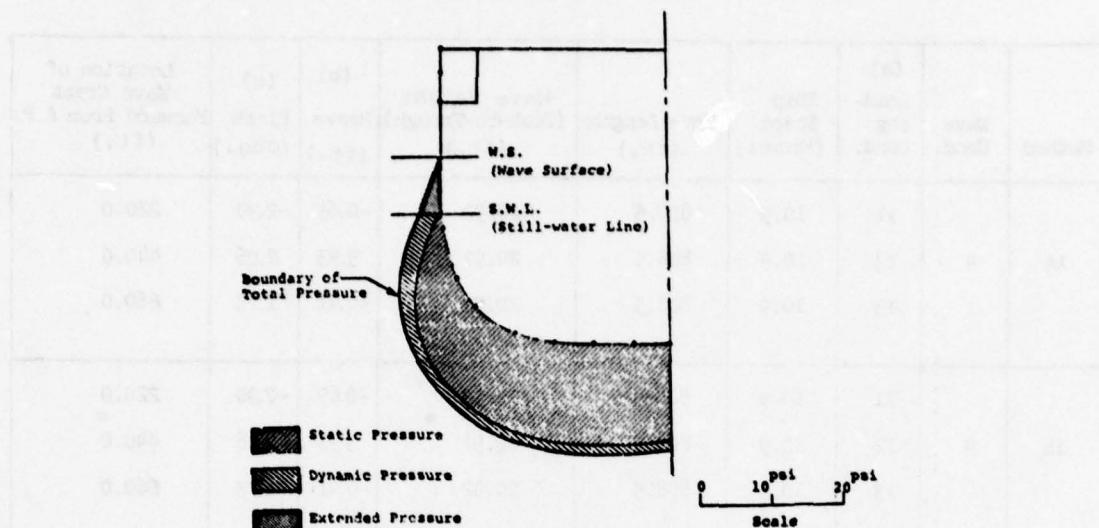
#### Method 1B - Hydrodynamic Pressures

The hydrodynamic pressure coefficients obtained from the ABS External Pressure subroutine (DYNPRE) were modified at each ship station to make the vertical wave bending moments comparable with those obtained from the ship motion calculation.

In both the ship motion and the external pressure calculations, information is obtained only for regions of the ship under the still-water line. To account for the actual wave profile, the external pressures were linearly extended up or deleted down to the wave surface, depending on whether the wave surface was above or below the still-water line. By so doing, the total external pressure, which includes both the hydrodynamic and the static components, becomes zero at the wave surface. The typical pressure distributions for wave surfaces above and below the still-water line are shown in Figures 22 and 23 respectively. The wave characteristics and ship motion data used for Methods 1A and 1B are shown in Table 9.

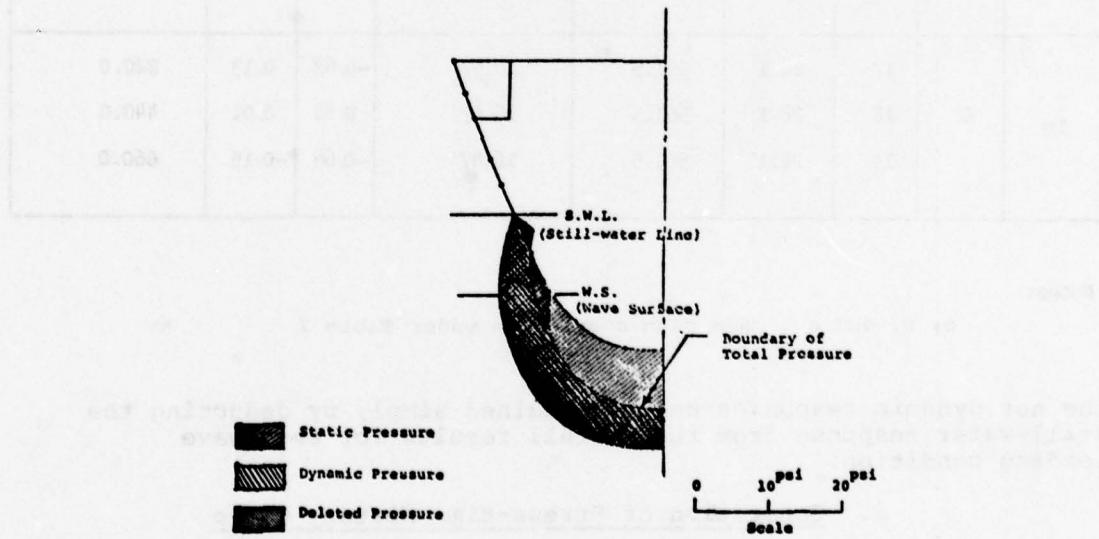
### c. Structural Analysis

After the calculation of dynamic and static loads, the stress analyses for coarse-mesh and subsequently for fine-mesh models were performed by using the ABS/DAISY finite element program. For the comparison of wave-induced stresses,



Frame 14 (Ship Frame 120)

FIGURE 22 - DISTRIBUTION OF STATIC AND WAVE-INDUCED DYNAMIC PRESSURES IN HEAD SEAS, W.S. > S.W.L.



Frame 44 (Ship Frame 250)

FIGURE 23 - DISTRIBUTION OF STATIC AND WAVE-INDUCED DYNAMIC PRESSURES IN HEAD SEAS, W.S. < S.W.L.

TABLE 9 WAVE CHARACTERISTICS AND SHIP MOTION DATA - METHOD 1 - TASK II

Method	Wave Cond.	(a) Load- ing Cond.	Ship Speed (knots)	Wave Length (ft.)	Wave Height (Peak-to-Trough) (ft.)	(b) Heave (ft.)	(c) Pitch (deg.)	Location of Wave Crest Forward From A.P. (ft.)
1A	4	11	10.9	808.5	20.92	-0.69	-2.30	220.0
		12	10.9	808.5	20.92	3.93	2.06	440.0
		13	10.9	808.5	20.92	-0.41	1.73	660.0
1B	4	11	10.9	808.5	20.92	-0.69	-2.30	220.0
		12	10.9	808.5	20.92	3.93	2.06	440.0
		13	10.9	808.5	20.92	-0.41	1.73	660.0
1B	5	14	18.7	808.5	21.97	1.02	-1.93	220.0
		15	18.7	808.5	21.97	7.30	3.27	440.0
		16	18.7	808.5	21.97	-2.65	1.2	660.0
1B	6	17	28.1	561.5	16.47	-0.63	0.13	220.0
		18	28.1	561.5	16.47	0.84	0.01	440.0
		19	28.1	561.5	16.47	-0.68	-0.15	660.0

Notes:

a, b, and c see sign convention under Table 7

the net dynamic responses can be obtained simply by deducting the still-water response from the overall results for each wave loading condition.

#### d. Generation of Stress-time History Curve

In order to determine the amplitude of the wave-induced stresses for each wave condition, a stress-time history of one complete stress cycle, as shown in Figure 24, was plotted based on three different wave-crest positions at the midship and the quarter points, and each wave-crest position was treated as a separate loading condition.

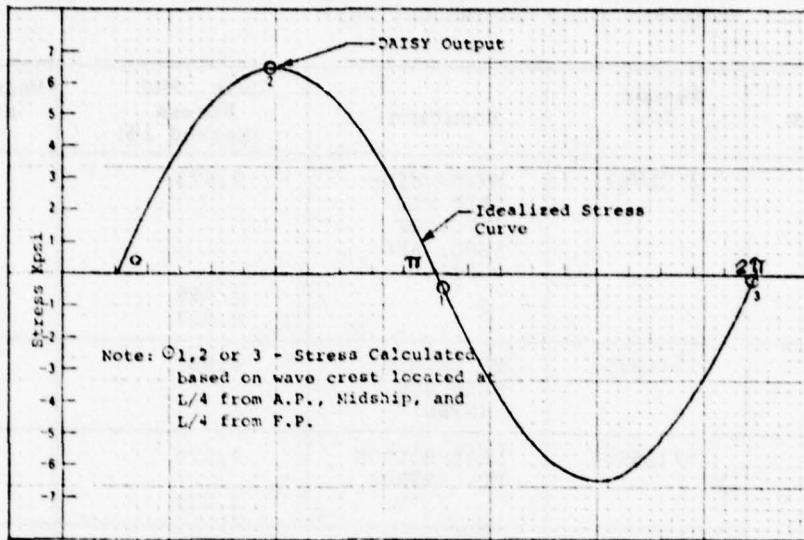


FIGURE 24 - IDEALIZED STRESS-TIME HISTORY CURVE, BASED ON THREE WAVE CREST POSITIONS

Two assumptions were made in generating the stress-time history. First, the time variation of stress was assumed to be sinusoidal. Second, the calculated longitudinal wave-induced stress was assumed to be in phase with the wave-induced vertical bending moment at the ship station under consideration. In the comparison of the calculated and the measured RMS stresses, the double stress amplitude (peak-to-trough) was utilized.

#### B. Calculated Results

For comparison, the calculated and measured RMS stresses for the three selected wave conditions are shown in Tables 10 and 11 and are also plotted in Figures 25, 26 and 27. The measured values were based on data reduction performed by Teledyne. The RMS stresses calculated by Methods 1A and 1B are generally in good agreement, as shown in Table 10. These stresses are generally of the same order of magnitude as the measured values. The calculated and measured RMS stresses for sensors AR1A, AR2A and AR3A in Table 11 show good agreement; however, sensor AR4A consistently shows that the measured data are much lower than the calculated values.

TABLE 10 COMPARISON OF CALCULATED AND MEASURED RMS LONGITUDINAL STRESSES (PEAK-TO-ROUGH, PSI)

Wave Condition	Sensor No.	Location	Calc. RMS Stress (Method 1B)	Meas. RMS Stress
4	1 (LVB)	MN.DK.MID-SHIP SEC. (AVG. OF STBD. AND PORT)	9,673	6,743
			8,469	6,344
			6,887	5,368
6	15 (LSTS)	MN.DK.MID-SHIP SEC. (STBD)	6,887	5,840
4	17 (LSBS)	NEAR-BOTTOM PL. (STBD)	4,570	4,164
			3,039	3,297
4	18 (LSTP)	MN.DK.MID-SHIP SEC. (PORT)	9,673	5,367
			6,887	5,529
4	20 (LSBP)	NEAR-BOTTOM PL. (PORT)	4,570	2,974
			4,623	2,743
			3,039	2,876
4	30 (AR1A)	MN.DK.FR. 143-144 (PORT)	5,285	5,027
			5,850	4,769
			3,433	4,649
4	33 (AR2A)	MN.DK.FR. 143-144 (STBD)	5,285	5,587
			5,850	5,767
			3,433	4,939
4	36 (AR3A)	MN.DK. FR. 143-144 (STBD)	10,505	7,537
			11,657	7,563
			6,844	6,576
4	39 (AR4A)	MN.DK.FR. 143-144 (STBD)	10,499	4,333
			11,450	4,017
			6,930	3,786

TABLE 11 COMPARISON OF CALCULATED AND MEASURED RMS LONGITUDINAL STRESSES (PEAK-TO-ROUGH, PSI)

WAVE CONDITION 4

Sensor No.	Calculated RMS Stress Method 1A	Measured RMS Stress Method 1B
1 (LVB)	9,243	9,673
18 (LSTP)	9,243	9,673
17 (LSBS)	4,320	4,570
20 (LSBP)	4,320	4,570
30 (AR1A)	6,121	5,285
33 (AR2A)	6,121	5,285
36 (AR3A)	10,566	10,505
39 (AR4A)	9,813	10,499

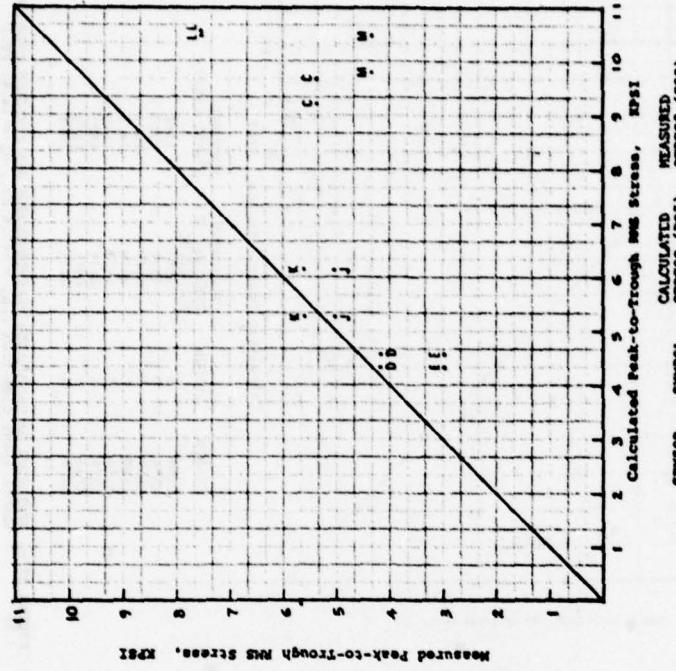


FIGURE 25 - DEVIATION OF CALCULATED PEAK-TO-ROUGH RMS STRESSES FROM MEASURED VALUES (WAVE CONDITION 1,  $L_w = 808.5$  FT.,  $H_w = 20.92$  FT.)

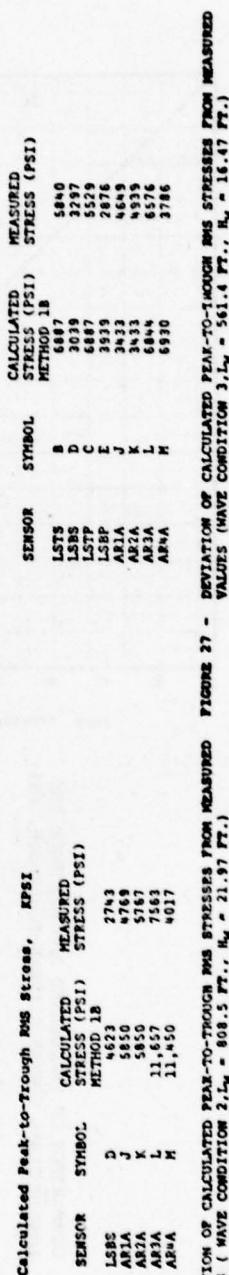
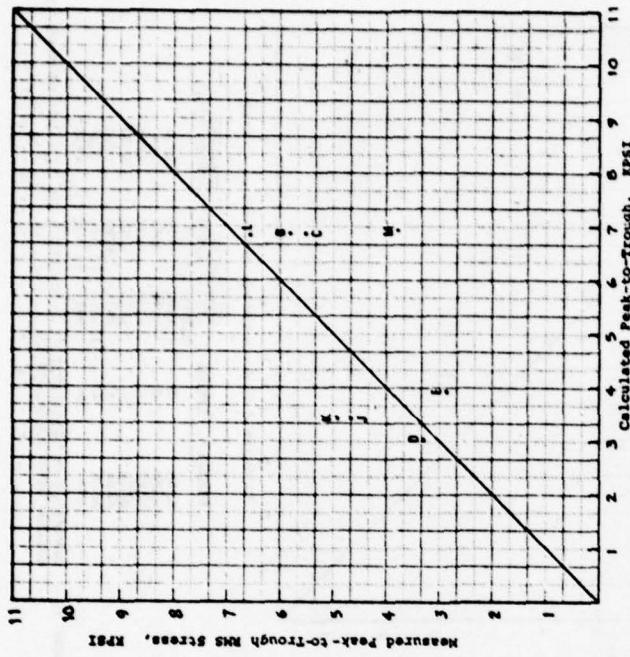


FIGURE 27 - DEVIATION OF CALCULATED PEAK-TO-THROUGH RMS STRESSES FROM MEASURED VALUES (WAVE CONDITION 3.  $\lambda_H = 561.4$  FT.,  $H_0 = 16.47$  FT.)

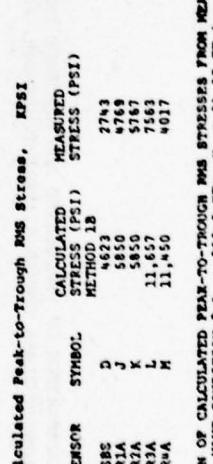
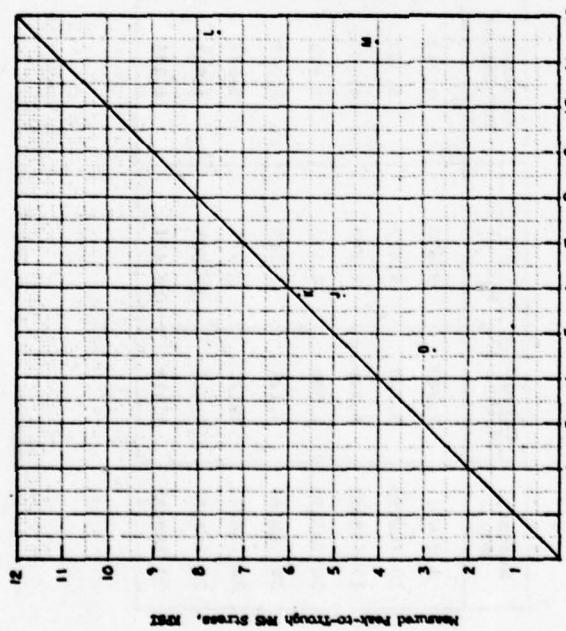


FIGURE 26 - DEVIATION OF CALCULATED PEAK-TO-THROUGH RMS STRESSES FROM MEASURED VALUES (WAVE CONDITION 2.  $\lambda_H = 608.5$  FT.,  $H_0 = 21.97$  FT.)

## 2. Method 2 - Stress Spectrum Approach

### A. Description of Method

The procedures used to calculate wave-induced dynamic loads and to perform the structural analysis are the same as those described in the previous sections. For the external pressures, Method 1B was used in conjunction with wave spectra from Reference [8] for the selected wave conditions. Since the selected three basic intervals have three different ship speeds, ranging from 10.9 to 28.1 knots, a stress RAO curve has to be generated for each ship speed. In order to have sufficient points for each selected wave spectrum a total of 23 wave lengths, as shown in Table 12, was used. The wave lengths were determined in such a way that these three basic wave spectra could be closely represented.

For each wave length the relative wave-crest position was selected to maximize the vertical bending moment at the ship station under consideration. This information was readily available from the ship motion calculations. Assuming that the longitudinal stresses in the deck plating are in phase with the vertical bending moment, the stress amplitude could be directly calculated for each wave frequency.

In an attempt to minimize the possible stress deviations due to the non-linear relation between stresses and wave heights, an average wave height of 19.68 ft was used for all the different waves. The computed stresses were then divided by the wave amplitudes to generate the stress RAO. Once the RAO curve in terms of a unit wave amplitude for the selected stress sensor is constructed, a stress spectrum can be obtained by multiplying the ordinate of the wave spectrum by  $(RAO)^2$ . The measured wave spectrum has a low-frequency cut-off value, as illustrated in Figures 28 through 41, below which the data are unreliable.

### B. Calculated Results

The calculated stress spectra for the three selected basic wave conditions are shown in Figures 28, 29 and 30 for sensor LVB and Figures 31, 32 and 33 for sensors LSBS and LSBP. The measured stress spectra reproduced from Reference [8] are also shown for comparison. The calculated RMS stresses, expressed in terms of  $2\sqrt{2}A$  (where A is the area under the stress spectrum curve) are listed in Table 14.

In order to make a further comparison for sensor LVB, eight additional recorded intervals with nearly head sea conditions and comparable ship speeds were selected from Reference [8]. The particulars for these intervals are listed in Table 13.

TABLE 12 - WAVE CHARACTERISTICS AND SHIP MOTION DATA - METHOD 2 - TASK II  
WAVE HEIGHT (PEAK-TO-TROUGH) = 19.68 FT. (6M)

Wave Cond.	(a) Load- ing Cond.	Ship Speed (knots)	Wave Encounter Frequency (rad./sec.)	Wave Length (ft.)	(b) Heave (ft.)	(c) Pitch (deg.)	Location of Wave Crest Forward From A.P. (ft.)
4	20	10.9	.279	3366.9	9.32	0.19	471.7
	21	10.9	.451	1460.5	7.26	0.81	448.8
	22	10.9	.560	1015.85	5.21	1.42	439.0
	23	10.9	.643	808.5	3.69	1.85	433.8
	24	10.9	.729	682.96	2.76	1.95	432.0
	25	10.9	.816	550.4	1.75	0.96	436.8
	26	10.9	1.088	350.8	-0.39	0.31	564.3
5	27	18.7	.430	1901.8	8.31	0.87	496.8
	28	18.7	.548	1295.4	7.38	1.55	464.3
	29	18.7	.611	1093.14	6.97	2.00	452.0
	30	18.7	.745	821.5	6.58	2.89	436.7
	31	18.7	.866	644.46	2.14	1.97	431.4
	32	18.7	1.019	509.33	1.75	0.02	431.8
	33	18.7	1.138	434.5	1.36	-0.05	426.8
6	34	28.1	.305	3887.9	7.57	0.65	820.3
	35	28.1	.508	1758.7	8.23	1.51	541.2
	36	28.1	.479	1322.0	8.72	2.25	492.4
	37	28.1	.712	1088.0	9.63	3.00	465.6
	38	28.1	.817	880.9	8.27	3.81	438.9
	39	28.1	.967	695.7	-1.52	1.30	433.4
	40	28.1	1.130	561.5	1.00	0.05	427.6
	41	28.1	1.273	478.4	1.03	-0.01	420.7
	42	28.1	1.579	359.3	-14	0.0	578.2

Notes:

a, b, and c see sign convention under Table 7

TABLE 13 PARTICULARS OF WAVES SELECTED FROM REFERENCE [8] FOR COMPARISON OF RMS STRESSES USING THE STRESS SPECTRUM APPROACH

Wave Condition	Ship Speed (knots)	Tape	Index	Interval	Run No.
7	32.6	143	9	36	337
8	32.5	145	27	41	441
9	32.3	153	4	15	815
10	32.4	153	8	29	829
11	31.8	153	12	45	845
12	31.2	155	17	1	901
13	32.6	157	12	45	1045
14	31.3	163	14	5	1305

TABLE 14 COMPARISON OF CALCULATED AND MEASURED RMS VERTICAL BENDING STRESSES (PEAK-TO-ROUGH, PSI) FOR SENSORS AT MIDSHIP

Wave Condition	Sensor	Calculated (Wave Spectrum) Approach	Measured
4	LVB	8343	6743
5		7295	6344
6		7183	5368
7		3547	2990
8		4192	3300
9		4544	4730
10		2542	2000
11		4252	3410
12		4766	4310
13		3188	3530
14		4595	5800
4	LSBS	3605	4101
5		2926	---
6		1954	3161
4	LSBS	3605	3261
5		2926	2729
6		1954	2732

Notes:

1. RMS values of stresses are defined as  $2\sqrt{A}$  where A is the area under the stress spectrum curve (Figures 28 through 41)
2. Stress RAO curve used for above Runs (except for Runs 405 and 429) is the same as that for Run 450 (Figure 30b)

Since these intervals have ship speeds comparable with wave condition 6, it is reasonable to use the stress RAO curve for wave condition 6, (Figure 30b) to calculate the stress spectra for the additional intervals. The calculated stress spectra together with the measured stress and wave spectra are shown in Figures 34 through 41. The calculated RMS stresses for these additional intervals, together with the measured values, are also listed in Table 14.

As shown in Figures 28 through 41, the calculated stress spectra generally exhibit shapes similar to those obtained from the measured data. Some of the calculated peak frequencies and peak amplitudes deviate significantly from the measured curves. However, the calculated RMS stresses as shown in Table 14 generally agree well with the measured values.

For a further comparison of the three different methods to calculate external pressures, some calculated and measured RMS longitudinal stresses in the main deck at midship (Sensor LVB) are shown in Table 15.

TABLE 15 COMPARISON ON CALCULATED AND MEASURED RMS VERTICAL BENDING STRESSES (PEAK-TO-TROUGH, PSI) FOR SENSOR LVB AT MIDSHIP

Wave Condition	Method	Calculated		Measured
		Equivalent Wave Approach	Stress Spectrum Approach (Method 2)	
4		9243	9673	6743
5		7810	8469	6344
6		7800	6887	5368

### 3. Conclusions

Based on the results and comparison discussed above, the following conclusions can be drawn:

A. In general, both Method 1 and Method 2 give a reasonably good agreement between the calculated RMS stresses and the measured data in head seas.

B. As shown in Tables 14 and 15, the RMS stresses calculated by the spectrum analysis (Method 2) agree very well with those obtained from the measured data. On the other hand, at some strain-gage locations (Table 10) the RMS stresses computed by Method 1 are significantly greater than the measured values. This may be attributed to the fact that the equivalent regular wave approach ignores the possible variations in structural responses to different wave frequencies. The spectrum analysis, which accounts for all the significant wave frequencies, gives a better average value. It should be noted that the spectrum analysis is valid only where the relation between the stresses and wave lengths can be approximated as linear.

C. The calculated stress spectra, as shown in Figures 28 through 41, assume similar shapes as those obtained from the measured data. However, the calculated peak frequency and peak amplitude may deviate from the measured ones in some comparisons.

D. The stresses calculated by Method 1A are, as expected, comparable with those obtained by Method 1B for head sea conditions.

Text continued on Page 53

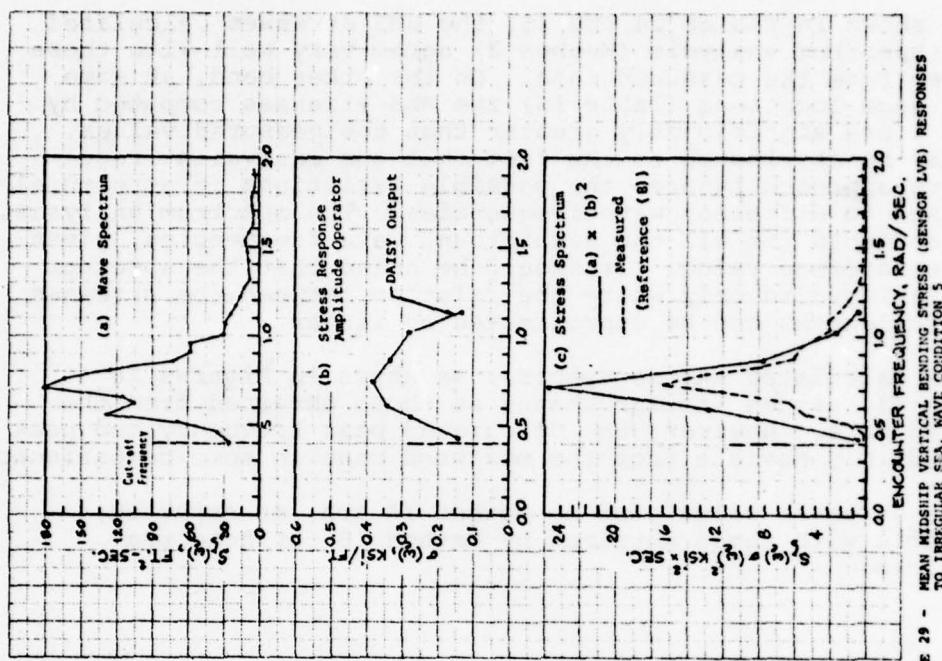


FIGURE 29 - MEAN MIDSHIP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 5

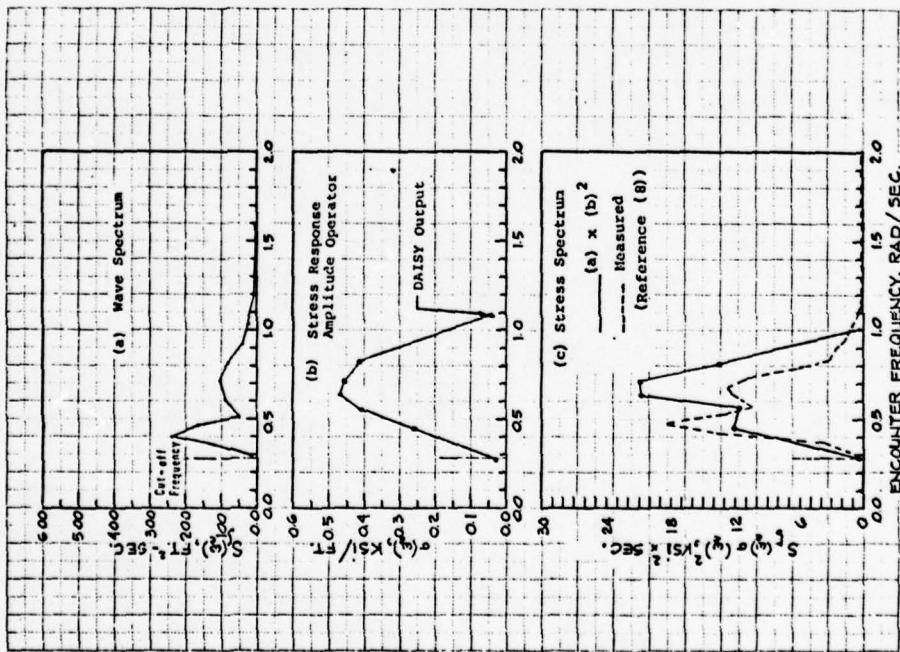


FIGURE 28 - MEAN MIDSHIP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 4

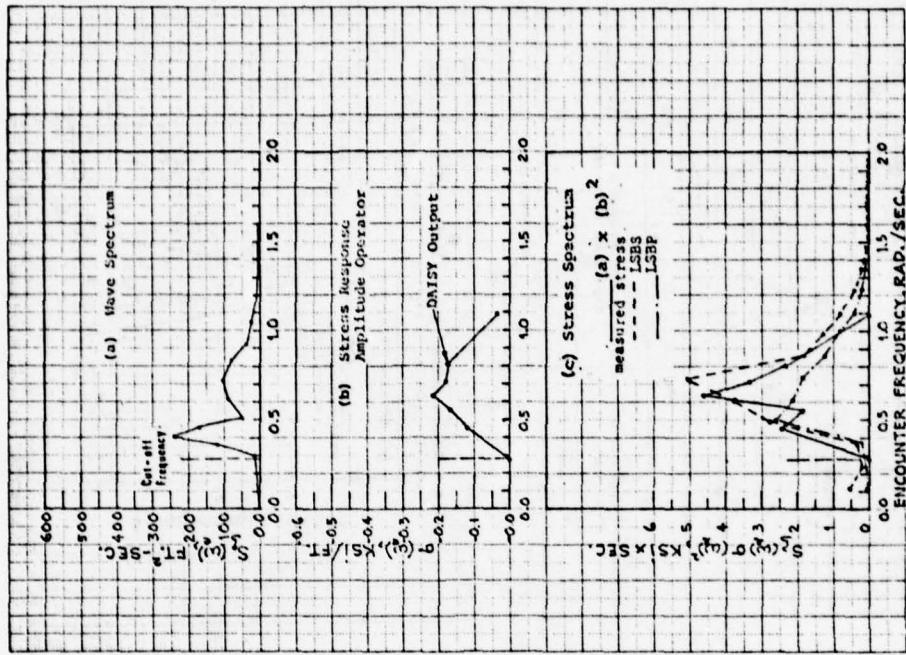


FIGURE 31 - MIDSHIP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 4

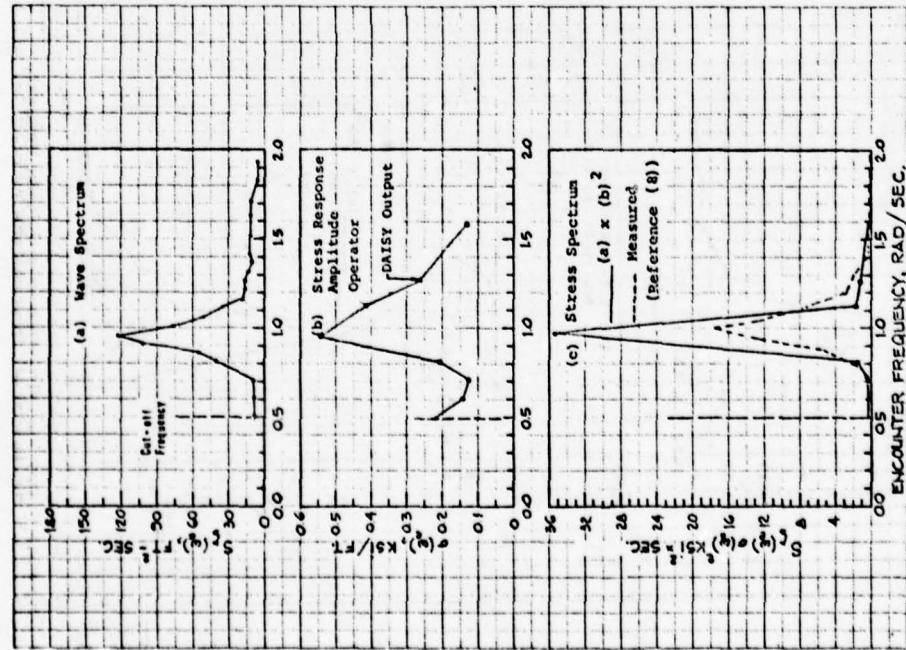


FIGURE 30 - MEAN MIDSHIP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 6

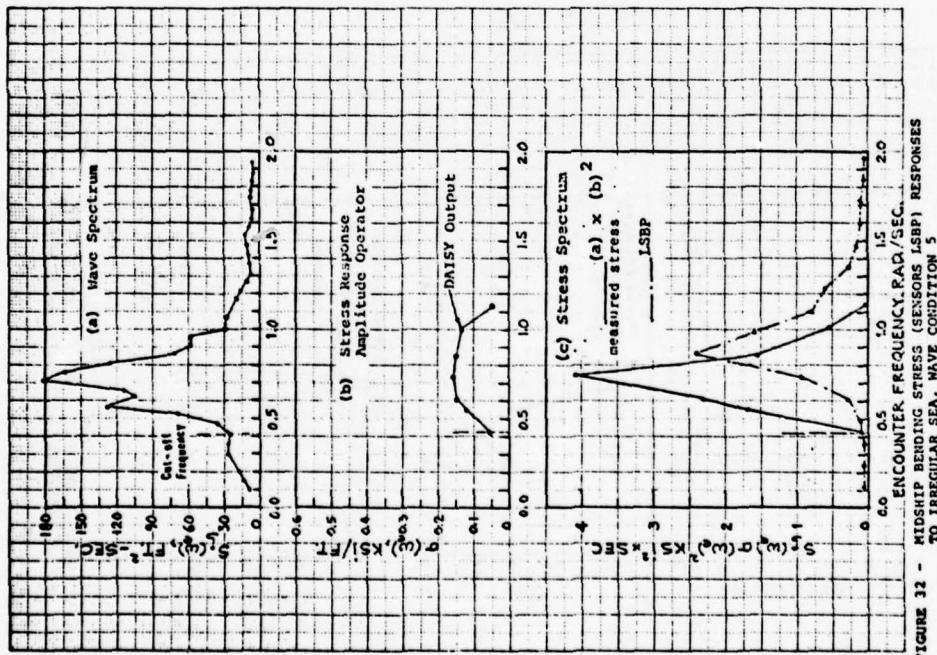


FIGURE 12 - MIDSHIP BENDING STRESS (SENSORS L5BP) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 5

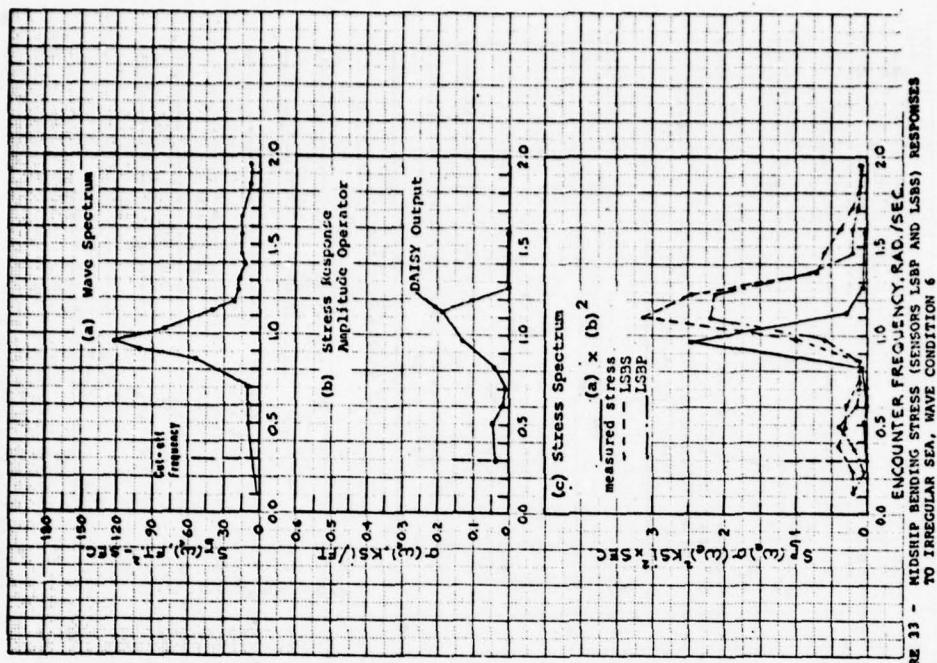


FIGURE 13 - MIDSHIP BENDING STRESS (SENSORS L5BP AND LSBS) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 6

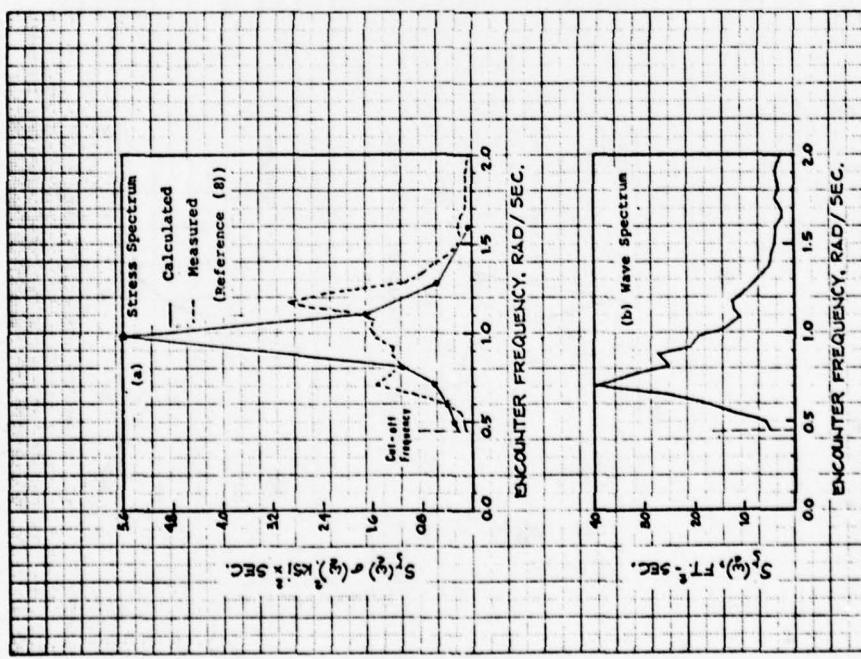
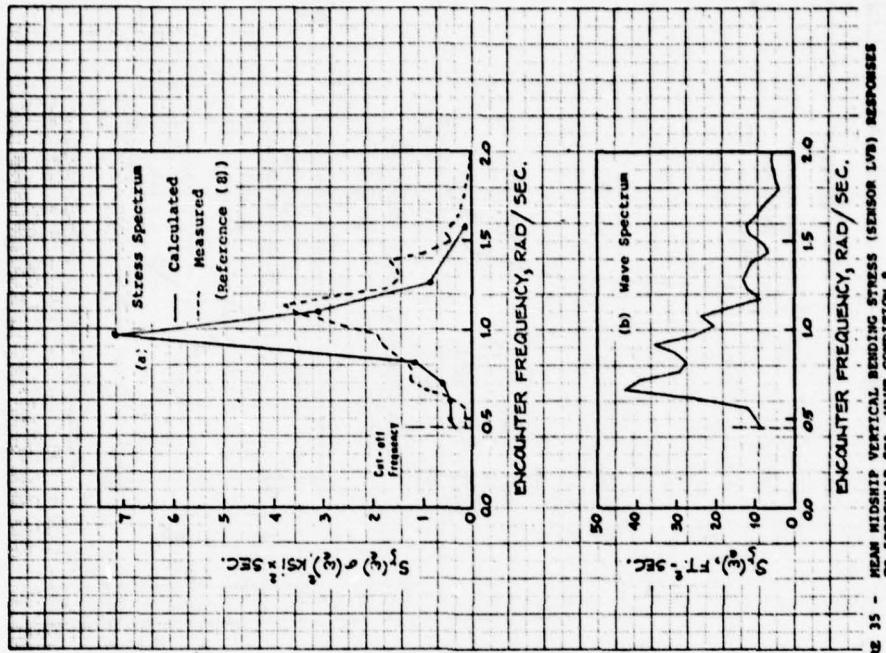


FIGURE 34 - MEAN MIDSHIP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES  
TO IRREGULAR SEA, WAVE CONDITION 7

FIGURE 35 - MEAN MIDSHIP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES  
TO IRREGULAR SEA, WAVE CONDITION 8

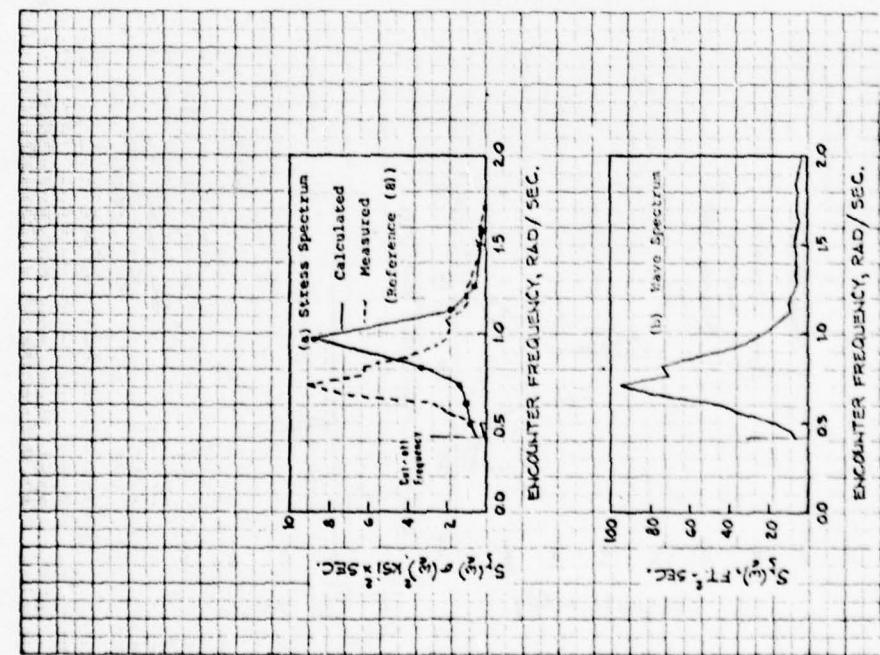


FIGURE 36 - MEAN MIDSHP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES  
TO IRREGULAR SEA, WAVE CONDITION 9

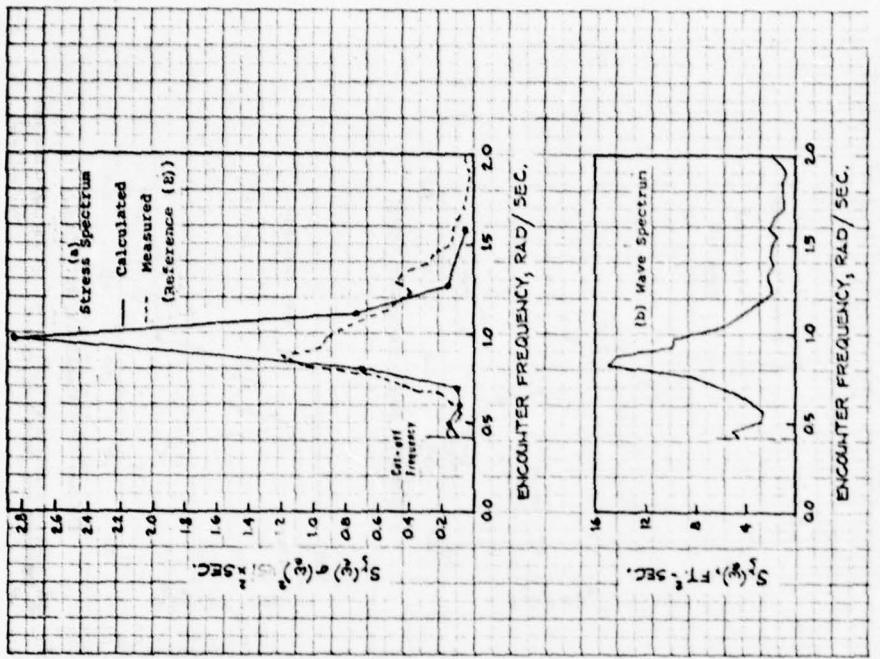


FIGURE 37 - MEAN MIDSHP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES  
TO IRREGULAR SEA, WAVE CONDITION 10

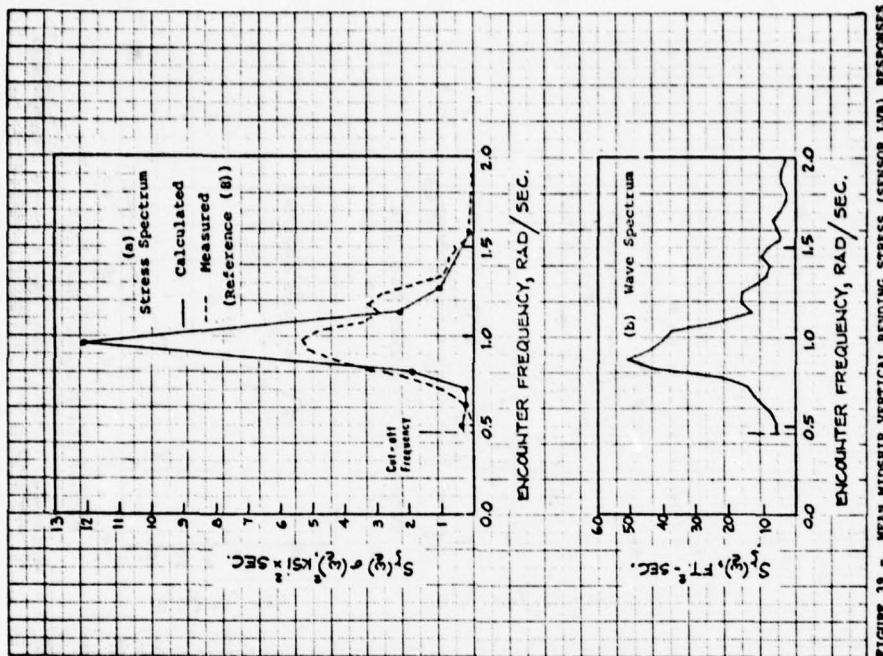


FIGURE 19 - MEAN MIDSHIP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 12

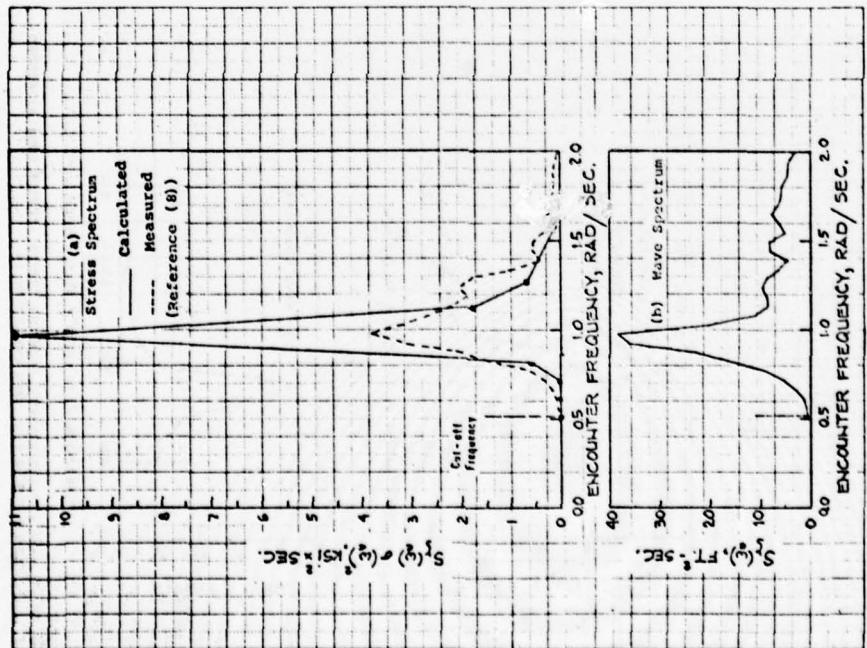


FIGURE 20 - MEAN MIDSHIP VERTICAL BENDING STRESS (SENSOR LVB) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 11

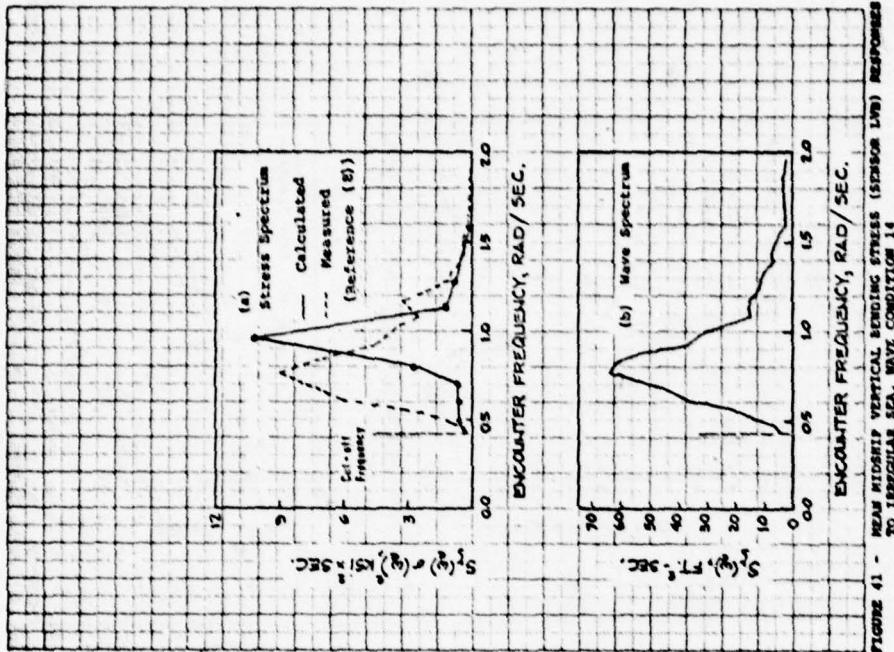


FIGURE 41 - MEAN SHIP VERTICAL BENDING STRESS (SENSOR L19) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 14

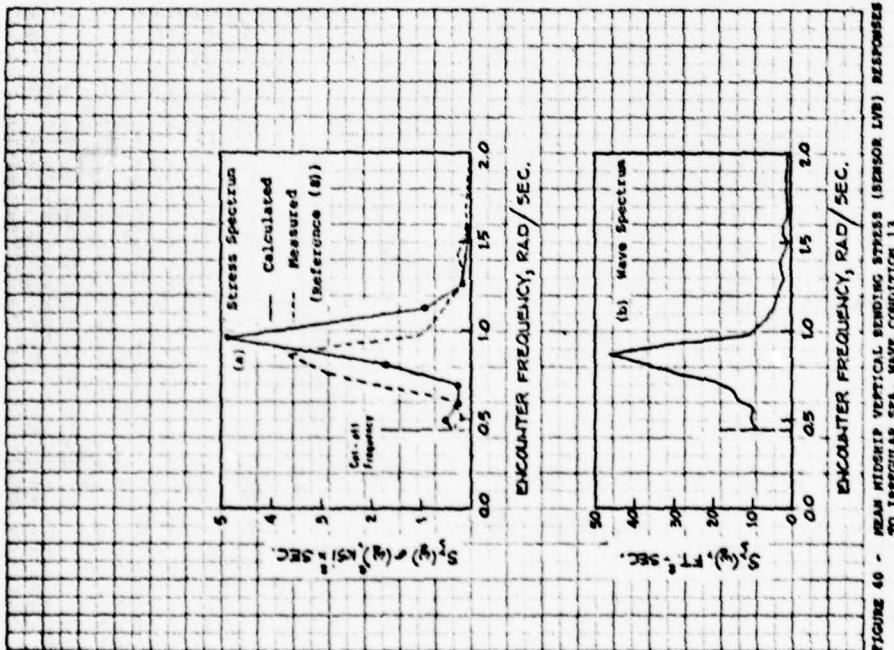


FIGURE 40 - MEAN SHIP VERTICAL BENDING STRESS (SENSOR L19) RESPONSES TO IRREGULAR SEA, WAVE CONDITION 13

### TASK III - INSTANTANEOUS STRESS COMPARISON IN HEAD SEAS

#### 1. Selection of Record Intervals

Two record intervals were selected for this phase of the study. The detailed environmental information on these two selected intervals is shown in Figures 42 and 43, as obtained from Reference [8]. The selected time span for stress correlation covers two complete encounter cycles, shown shaded within a circle on the time history of the radar wave elevation, Figures 42 and 43.

#### 2. Ship Motion Calculation

To facilitate the wave-load calculations, the recorded wave elevation within the time period of two complete cycles was approximated by a regular wave with a mean encounter frequency, an average height and an equivalent wave length (See the enlarged scale in Figures 42 and 43).

The wave particulars for the two selected conditions are shown in Table 16. Once the wave particulars were established, a ship motion calculation was performed using the ABS/SHIPMOTION program to compute motions, accelerations, shearing forces and bending moments for three positions of the wave crest along the length of the vessel, namely, at the midship and two quarter-length stations.

TABLE 16 WAVE CHARACTERISTICS AND SHIP MOTION DATA - TASK III

Wave Cond.	Loading Cond.	Ship Speed (knots)	Wave Length (ft.)	Wave Height (Peak-to-Trough)	Heave (ft.)	Pitch (deg.)	Location of Wave Crest Forward From A.P. (ft.)
15	43	10.9	1651.0	51.2	-15.9	-2.8	220.0
	44	10.9	1651.0	51.2	20.4	1.48	440.0
	45	10.9	1651.0	51.2	11.4	4.79	660.0
16	46	28.1	808.5	30.6	14.35	0.48	220.0
	47	28.1	808.5	30.6	4.64	5.40	440.0
	48	28.1	808.5	30.6	-15.64	-1.99	660.0

LOG BOOK DATA			
DATE AND TIME	01-10-74	2400	
POSITION	43-29 N	24-51 W	
COURSE AND SPEED	250	10.9 KNOTS	
SEA STATE	7		
WAVE HEIGHT	25 FEET		
REL DIR	2 PORT		
SWELL HEIGHT	20 FEET		
REL DIR	2 PORT		
---- VISUAL WEATHER / COMMENTS ----			
OCAST /			
WAVE HEIGHT STATISTICS (FEET)			
TUCKER/DYN. HEAD/RADAR			
P-T SAMPLE SIZE	93	70	121
MAXIMUM HEIGHT	15.9	43.4	57.2
10TH HIGHEST HTS	15.0	39.0	39.7
3RD HIGHEST HTS	13.0	33.8	32.5
4.0 RMS(SPECTRA)	13.3	34.0	33.5

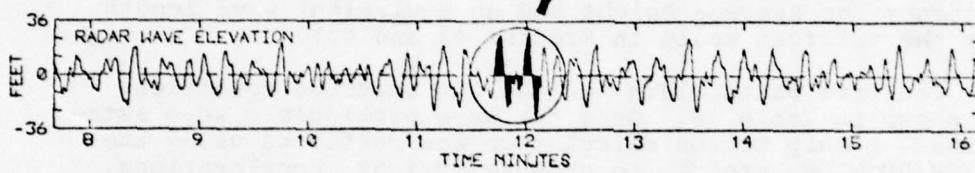
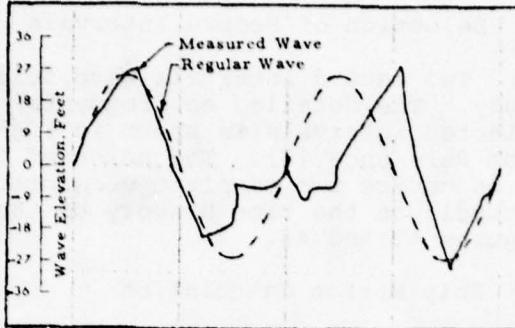


FIGURE 42 - CONDITION 15 - VOYAGE 32W -- TAPE 145 -- INDEX 18 -- INTERVAL 5 - RUN 405

LOG BOOK DATA			
DATE AND TIME	01-12-74	1630	
POSITION	41-07 N	20-28 W	
COURSE AND SPEED	245	28.1 KNOTS	
SEA STATE	9		
WAVE HEIGHT	25 FEET		
REL DIR	2 STBD		
SWELL HEIGHT	20 FEET		
REL DIR	2 STBD		
---- VISUAL WEATHER / COMMENTS ----			
OCAST /			
WAVE HEIGHT STATISTICS (FEET)			
TUCKER/DYN. HEAD/RADAR			
P-T SAMPLE SIZE	239	179	222
MAXIMUM HEIGHT	8.3	8.9	59.0
10TH HIGHEST HTS	5.9	7.0	33.7
3RD HIGHEST HTS	4.6	5.4	26.4
4.0 RMS(SPECTRA)	5.5	6.3	29.4

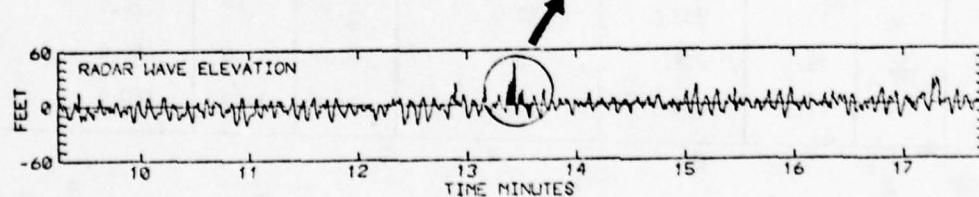
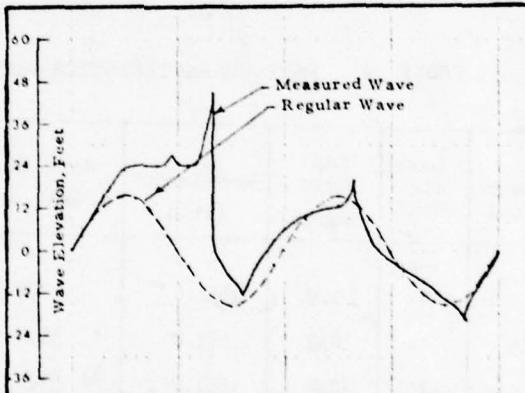


FIGURE 43 - CONDITION 16 - VOYAGE 32W -- TAPE 145 -- INDEX 29 -- INTERVAL 50 - RUN 450

### 3. Input to DAISY Model

A modification of hydrodynamic pressure coefficients, described as Method 1B in Task II, was carried out to make the wave-induced vertical shearing forces and bending moments acting on the DAISY model generally identical to those calculated directly by the SHIPMOTION program. A sample comparison is shown in Figure 44.

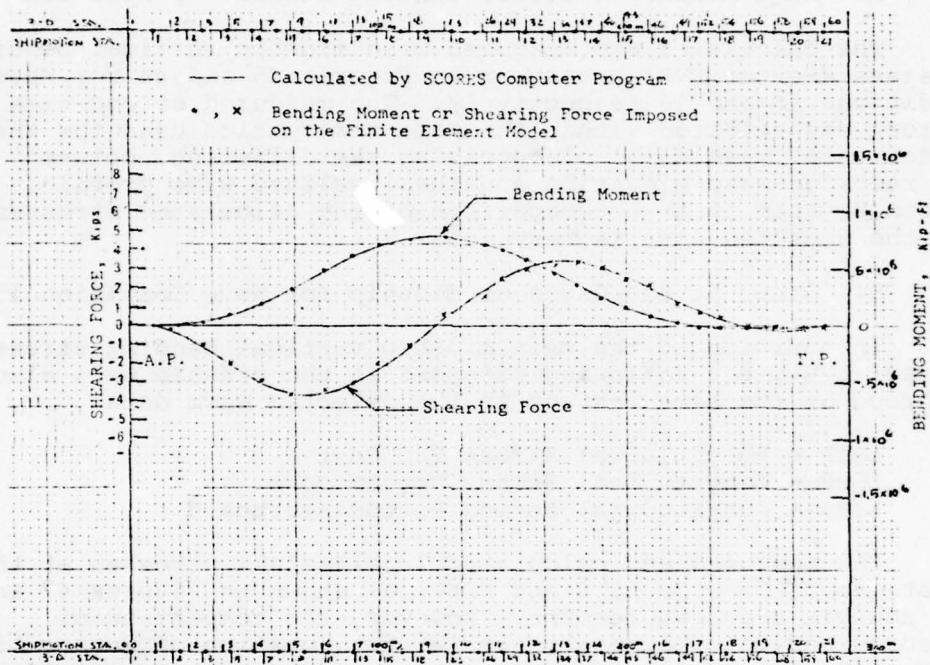


FIGURE 44 - WAVE-INDUCED BENDING MOMENTS AND SHEARING FORCES

### 4. Comparison of Calculated Stresses and Measured Data

For each wave condition, three values of stress were calculated, corresponding to the three wave-crest positions described above. In order to construct a diagram of stress-time history based on the three calculated values, it was assumed that the computed stress follows a sinusoidal curve

with a frequency equal to the wave-encounter frequency. It was further assumed that the stress variation is directly proportional to the dominant wave-load component, the phase angle of which can be obtained from the ship motion calculations. For longitudinal stresses in the deck or bottom regions, the dominant wave-load component was assumed to be the vertical bending moment. A mean stress amplitude obtained from these three calculated stress values, with proper phase angles, was then utilized to plot an idealized stress curve (See Figures 45 and 46).

#### A. Mean Midship Vertical Bending Stress - Sensor LVB

The calculated and measured mean midship vertical bending stresses (Sensor LVB) are shown in Figures 45 and 46 for Wave Conditions 15 and 16 respectively. The measured stress-time history was selected within the same time period used for the selection of wave data. Recognizing the difference between the recorded wave elevation and the idealized wave profile, it can be seen that the comparison of the calculated stresses and the measured data is very good.

#### B. Other Stress Gages at Midship for Wave Condition 15

In addition to the mean midship vertical bending stress (Sensor LVB), the following stresses at the midship were also recorded at the same instant as the selected wave data:

LSTP - Longitudinal Stress Top Port  
LSBP - Longitudinal Stress Bottom Port  
LSBS - Longitudinal Stress Bottom Starboard

The measured and calculated longitudinal stresses at the locations of Sensors LSTP and LSBP, as shown in Figures 47 and 48, exhibit a good agreement. However, the comparison of measured and calculated stresses at the location of Sensor LSBS, Figure 49, is relatively poor. The measured stresses of LSBP and LSBS should be generally comparable in head seas. The difference between the measured stresses of LSBP and LSBS may be attributed to irregularity of the encounter wave and possibly irregularity of gage signals.

It should be noted that all the stress data other than the mean midship vertical bending stress (LVB) was generally recorded on Recorder No. 2, whereas the wave data and LVB were recorded on Recorder No. 1. In addition, all the stress data other than LVB was reduced separately from that of the wave data.

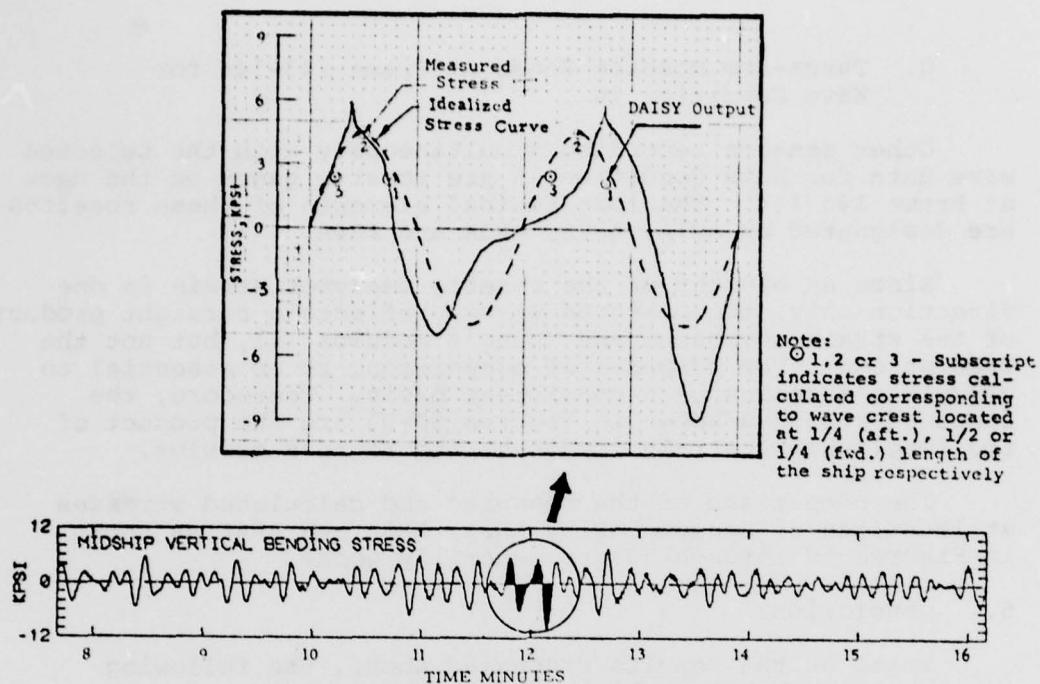


FIGURE 45 - COMPARISON OF THE CALCULATED AND MEASURED MEAN MIDSHIP VERTICAL BENDING STRESSES (SENSOR LVB), WAVE CONDITION 15

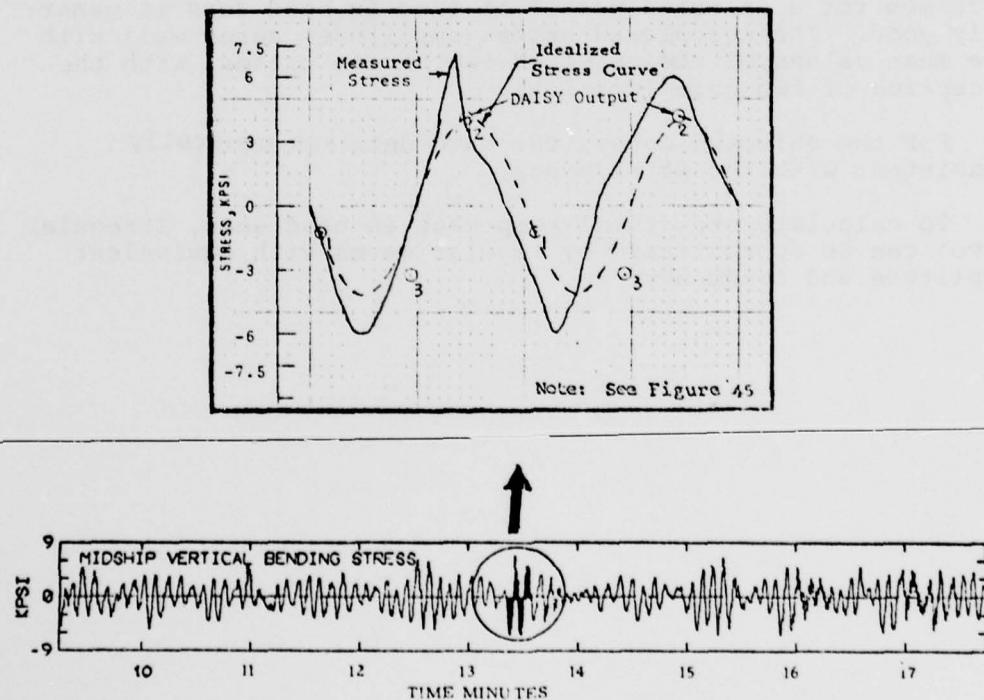


FIGURE 46 - COMPARISON OF THE CALCULATED AND MEASURED MEAN MIDSHIP VERTICAL BENDING STRESSES (SENSOR LVB), WAVE CONDITION 16

C. Three-arm Rosette Gages at Frame 143-144 for Wave Condition 16

Other sensors recording simultaneously with the selected wave data for Wave Condition 16 are rosette gages on the deck at Frame 143-144. The longitudinal elements of these rosettes are designated by AR1A, AR2A, AR3A and AR4A.

Since an element of the rosette measures strain in one direction only, the measured stress reflects a straight product of the strain component and Young's modulus (E), but not the true stress. For purposes of comparison, it is essential to calculate stresses on a consistent basis. Therefore, the computed stresses shown in Figures 50-53 are the product of the longitudinal strain component and Young's modulus.

The comparison of the measured and calculated stresses at locations of Sensors AR1A, AR2A, AR3A and AR4A as shown in Figures 50 through 53 are generally good.

5. Conclusions

Based on the results discussed above, the following conclusions can be established for Task III:

- A. The comparison of the measured and calculated longitudinal stresses for a selected period of time in head seas is generally good. The calculated stress amplitudes agree well with the mean values of the selected two stress cycles, with the exception of two gage locations.
- B. For the selected cases, the wave data are generally consistent with the stress data.
- C. To calculate structural responses in head seas, irregular waves can be approximated by regular waves with equivalent amplitude and frequency.

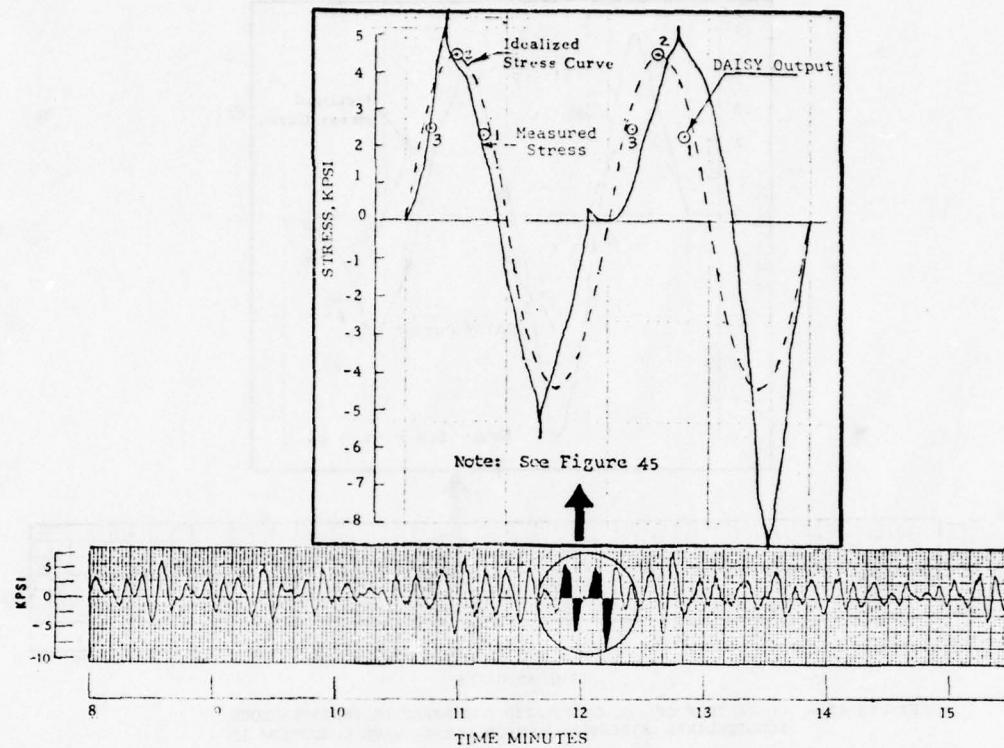


FIGURE 47 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR LSTP), WAVE CONDITION 15

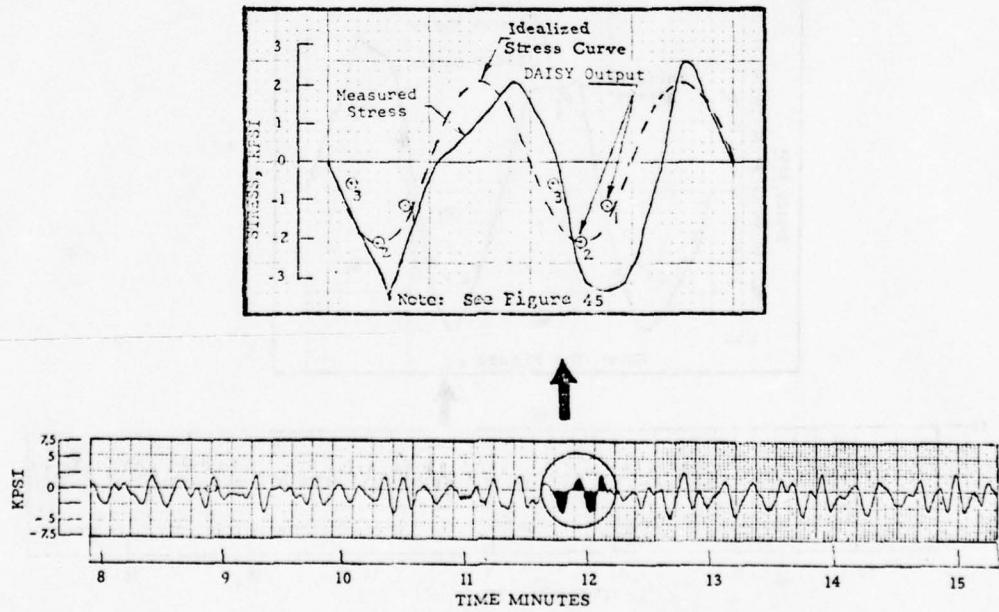


FIGURE 48 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR LSBP), WAVE CONDITION 15

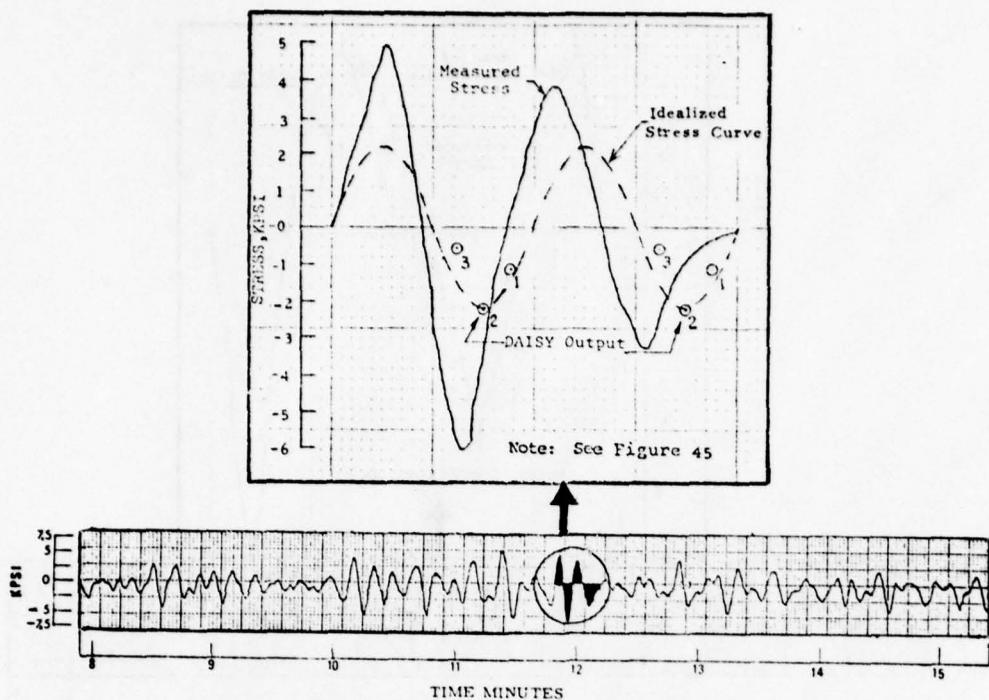


FIGURE 49 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR LSBS), WAVE CONDITION 15

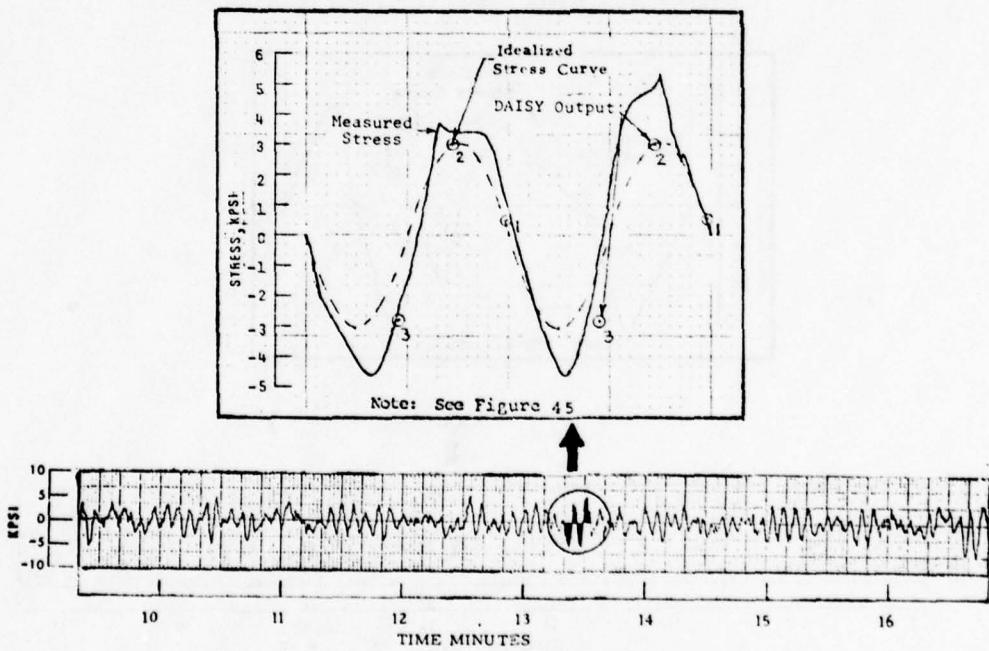


FIGURE 50 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR ARIA), WAVE CONDITION 16

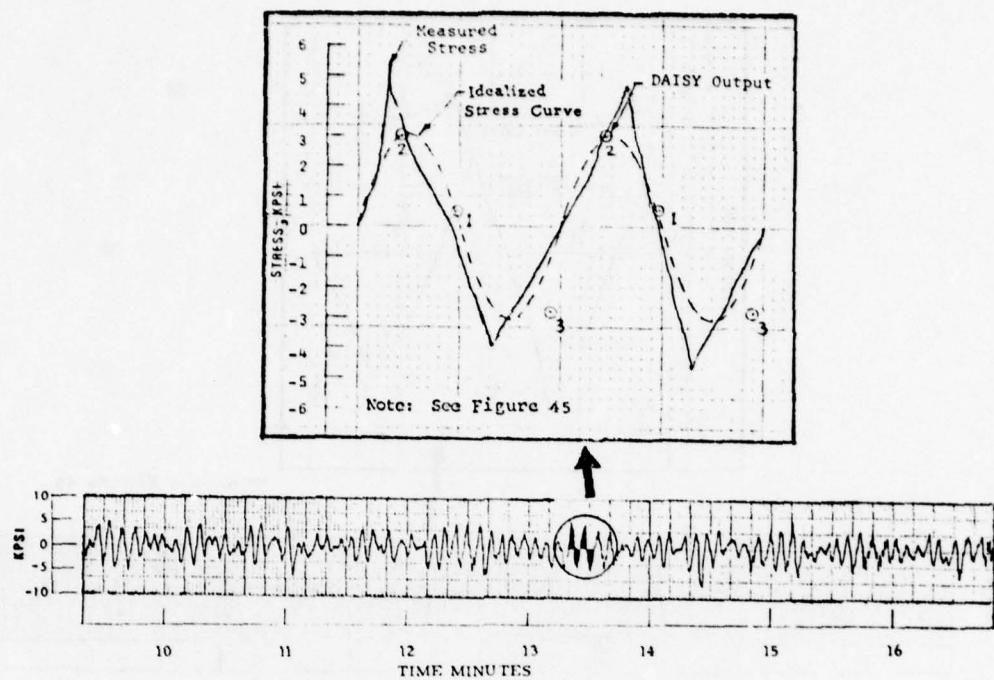


FIGURE 51 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR AR2A), WAVE CONDITION 16

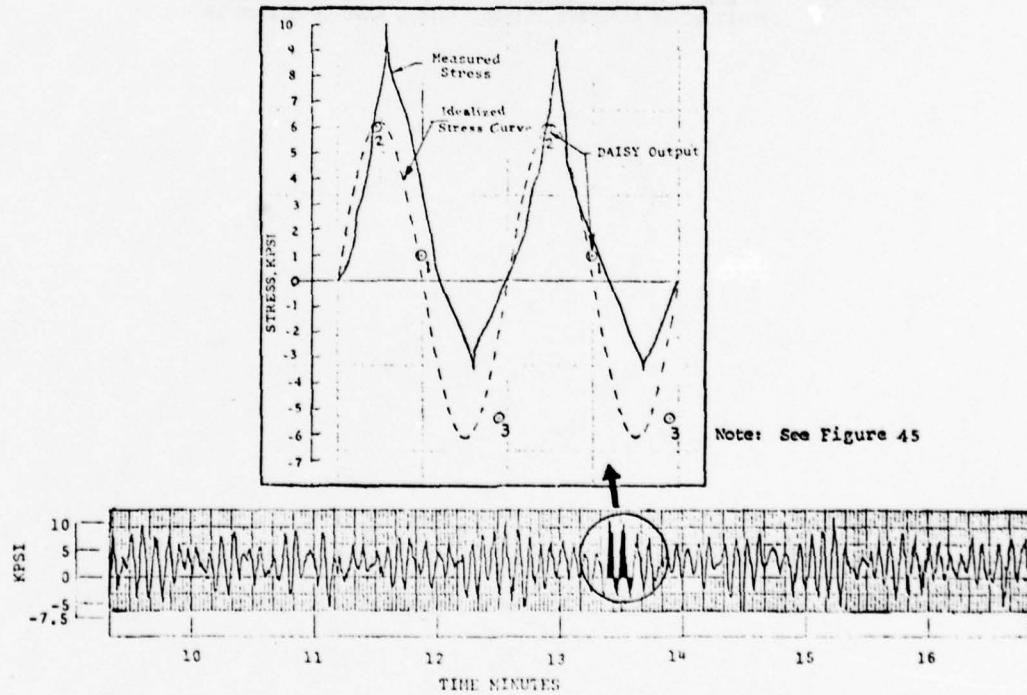


FIGURE 52 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR AR3A), WAVE CONDITION 16

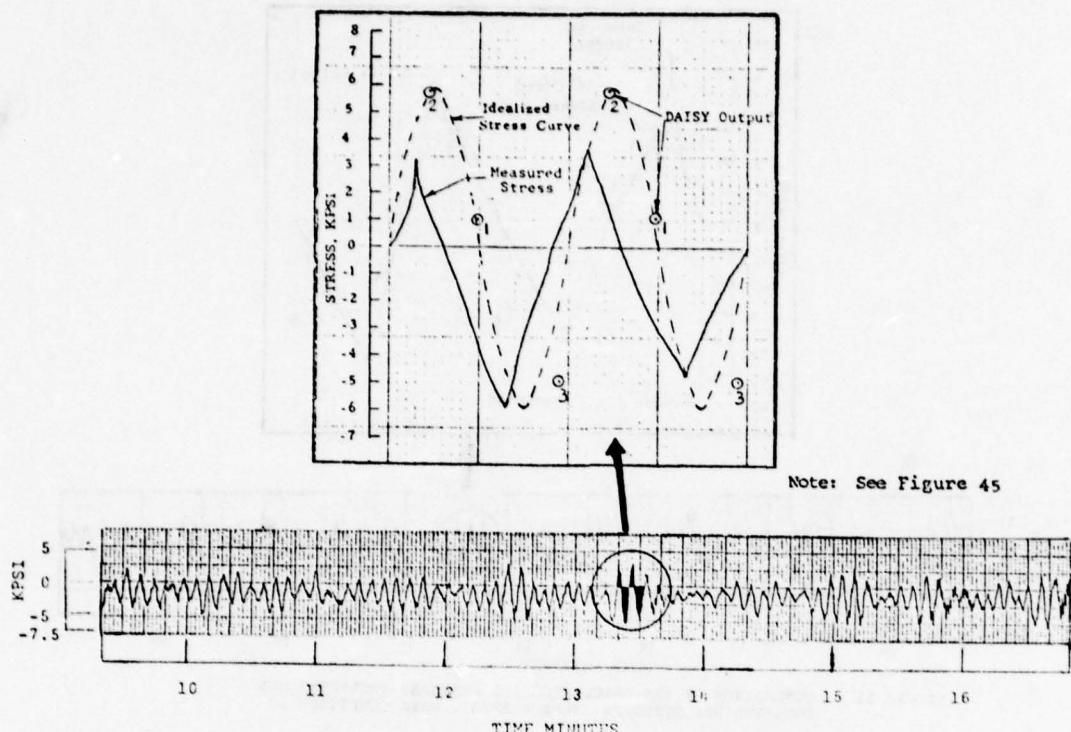


FIGURE 53 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR AR4A), WAVE CONDITION 16

#### TASK IV - INSTANTANEOUS STRESS COMPARISON IN OBLIQUE SEAS

##### 1. Method of Approach

Due to the complexity of the wave load pattern in oblique seas, difficulties were experienced in dealing with the hydrodynamic pressures and balancing of the structural model. To improve the consistency between the external pressures and the ship motion calculation, hydrodynamic pressure coefficients were modified for lateral moments in addition to the modifications for vertical moments described in Task II.

To investigate the feasibility of performing an instantaneous stress comparison in oblique seas, two record intervals were selected for test runs. The detailed information for the selected intervals and the radar wave data as obtained from Reference [8] are shown in Table 17.

TABLE 17 WAVE CONDITION SELECTED FROM REFERENCE [8] FOR THE COMPARISON OF INSTANTANEOUS STRESSES IN OBLIQUE SEAS

Wave Condition	Tape	Index	Interval	Run No.	Ship Speed (knots)	Wave Relative Dir. (deg.)	Heading Angle (deg.)
17	143	11	44	345	32.3	64 Port	116
18	143	12	48	349	31.8	41 Port	139

For instantaneous stress comparison, two complete cycles of the recorded wave elevation were selected and approximated by a regular wave with a mean encounter frequency, an average height and an equivalent wave length, (see the enlarged scale in Figures 54 and 55).

For each wave, three wave-crest positions, at midship and the two quarter-length points, were utilized to calculate wave loads. Each wave-crest position was treated as a separate loading condition. The particulars of these loading conditions are shown in Table 18.

LOG BOOK DATA		
DATE AND TIME	01-09-74	2000
POSITION	45-08 N	13-30 W
COURSE AND SPEED	267	. 32.3 KNOTS
SEA STATE	8	
WAVE HEIGHT	10 FEET	
REL DIR	64 PORT	
SWELL HEIGHT	10 FEET	
REL DIR	64 PORT	
---- VISUAL WEATHER / COMMENTS ----		
CLDY /		
WAVE HEIGHT STATISTICS (FEET)		
TUCKER/DYN. HEAD/RADAR		
P-T SAMPLE SIZE	201	128 235
MAXIMUM HEIGHT	8.4	13.0 36.4
10TH HIGHEST HTS	6.0	9.4 28.3
3RD HIGHEST HTS	4.5	7.5 21.9
4.0 RMS(SPECTRA)	5.2	7.9 22.9

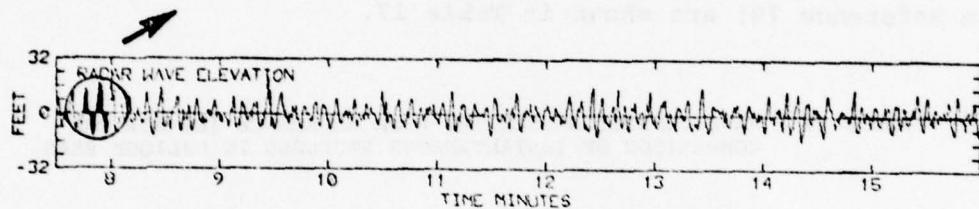
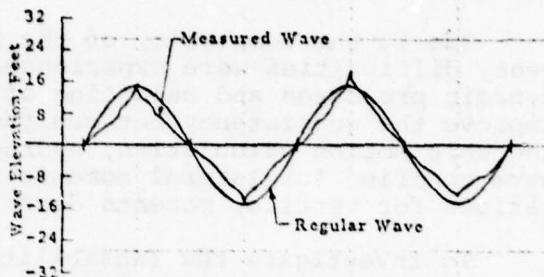


FIGURE 54 - CONDITION 17 - VOYAGE 32W - TAPE 143 - INDEX 11 - INTERVAL 44 - RUN 345

LOG BOOK DATA		
DATE AND TIME	01-09-74	2400
POSITION	45-08 N	13-30 W
COURSE AND SPEED	266	. 31.8 KNOTS
SEA STATE	9	
WAVE HEIGHT	15 FEET	
REL DIR	41 PORT	
SWELL HEIGHT	10 FEET	
REL DIR	41 PORT	
---- VISUAL WEATHER / COMMENTS ----		
OCAST /		
WAVE HEIGHT STATISTICS (FEET)		
TUCKER/DYN. HEAD/RADAR		
P-T SAMPLE SIZE	137	103 146
MAXIMUM HEIGHT	9.9	19.1 40.7
10TH HIGHEST HTS	7.8	15.1 37.8
3RD HIGHEST HTS	6.3	12.5 33.0
4.0 RMS(SPECTRA)	6.8	13.2 33.5

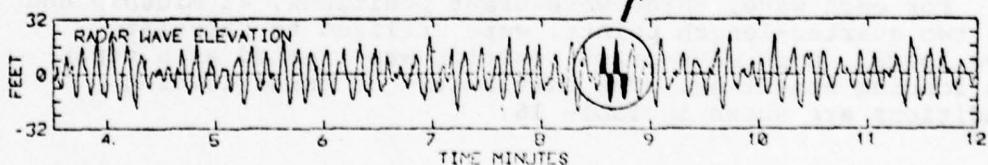
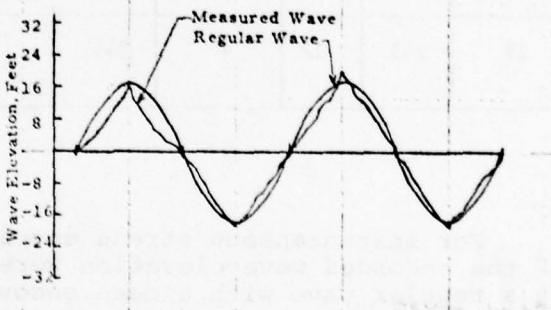


FIGURE 55 - CONDITION 18 - VOYAGE 32W - TAPE 143 - INDEX 12 - INTERVAL 48 - RUN 349

TABLE 18 WAVE CHARACTERISTICS AND SHIP MOTION DATA - TASK IV

Wave Cond.	(a) Load- ing Cond.	Ship Speed (knots)	Wave Length (ft.)	Wave Height (Peak-to-Trough) (ft.)	(b) Heading Angle (deg.)	(c) Roll (deg.)	(d) Heave (ft.)	(e) Pitch (deg.)	Location of Wave Crest Forward From A.P. (ft.)
17	49	32.3	540.15	30.18	116	-0.27	21.45	3.15	220.0
	50	32.3	540.15	30.18	116	1.40	3.97	5.76	440.0
	51	32.3	540.15	30.18	116	1.495	-18.0	1.85	660.0
18	52	31.8	888.29	33.46	139	-0.457	17.5	-0.95	220.0
	53	31.8	888.29	33.46	139	1.252	20.4	5.317	440.0
	54	31.8	888.29	33.46	139	1.40	-1.8	5.05	660.0

## Notes:

a, b, and c see sign convention under Table 7

d Heading angle is measured counterclockwise from ship centerline to wave direction

e Roll is positive starboard deck edge down

## 2. Calculation of Wave-induced Loads

To determine the wave-induced loads for input to DAISY, the SHIPMOTION and DYNPRE programs were used. In oblique seas, the wave-induced moments consist generally of three components, namely vertical, lateral and torsional moments. Since the wave-induced loads obtained by integration of a set of hydrodynamic pressures and the corresponding inertia forces would generally not agree with those obtained from the ship motion calculation, a correction coefficient for the hydrodynamic pressure was introduced to modify the vertical and lateral moments for the external pressures. After this modification, the torsional moments may not be in agreement with those obtained from the ship motion calculation.

Because of the complexity of the wave-load pattern in oblique seas and the difficulty in balancing the DAISY model in the lateral direction, the external pressures were not extended up or deleted down to the wave surface for this task.

### 3. Structural Analysis

The procedures utilized for calculating structural responses for this task are generally the same as those in Tasks II and III, except that the asymmetric wave load was divided into symmetric and anti-symmetric components which were then treated as separate loading conditions in DAISY. The combination of the symmetric and anti-symmetric cases gives the total response for both the port and starboard sides (See Figure 5).

For each wave, in addition to the three dynamic loading conditions corresponding to three wave-crest positions, a static case reflecting the still-water condition was also included in the analysis. The wave-induced stresses were then obtained by subtracting the still-water stresses from the total responses for each dynamic loading condition.

### 4. Calculated Results

In order to generate a stress-time history curve based on the calculated stresses at three different wave-crest positions, the following two assumptions were made:

a. The curve was assumed to be sinusoidal.

b. The calculated longitudinal wave-induced stress was assumed to be in phase with the effective moment ( $M_e$ ) combining the wave-induced vertical and lateral bending moments, expressed in terms of

$$M_e = \left[ M_V + \frac{M_L}{(SM_L/SM_V)} \right] \quad (5)$$

where  $M_V$ ,  $M_L$  = wave-induced vertical and lateral bending moments respectively, at the location under consideration.

$SM_V$ ,  $SM_L$  = hull-girder section modulus with respect to vertical and lateral bending moments respectively, at the location under consideration.

The phase angles of  $M_V$  and  $M_L$  were obtained directly from the ship motion calculation, and the effective moment,  $M_e$ , was then determined graphically. A sample plot is shown in Figure 56.

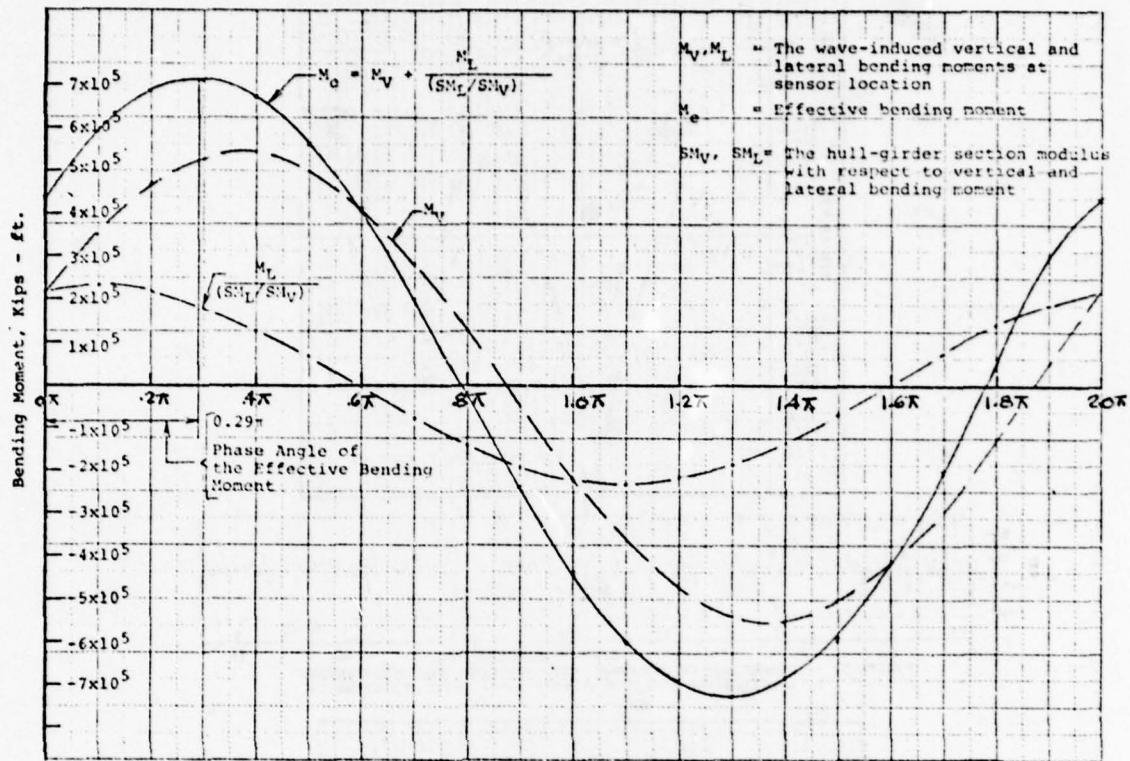


FIGURE 56 - GRAPHIC SOLUTION FOR THE PHASE ANGLE OF THE EFFECTIVE BENDING MOMENT

##### 5. Comparison of Calculated Stresses and Measured Data

For each wave condition, three values of stress were calculated, corresponding to the three wave-crest positions described above. In order to construct a diagram of stress-time history based on the three calculated values, it was assumed that the computed stress follows a sinusoidal curve with a frequency equal to the wave-encounter frequency. It was further assumed that the stress variation is in direct proportion with the dominant wave-load component, and that the phase angle for combined wave-induced vertical and lateral bending moments can be obtained using the method described in the previous section. A mean stress amplitude obtained from these three calculated stress values with proper phase angles was then utilized to plot idealized stress curves as shown in Figures 57 through 67.

Based on the results shown in Figures 57 through 67, it can be seen that the calculated stress-time histories generally agree with the measured data, except for Sensor AR4A in Figure 67. Text contained on page 74.

[Text continued on Page 74]

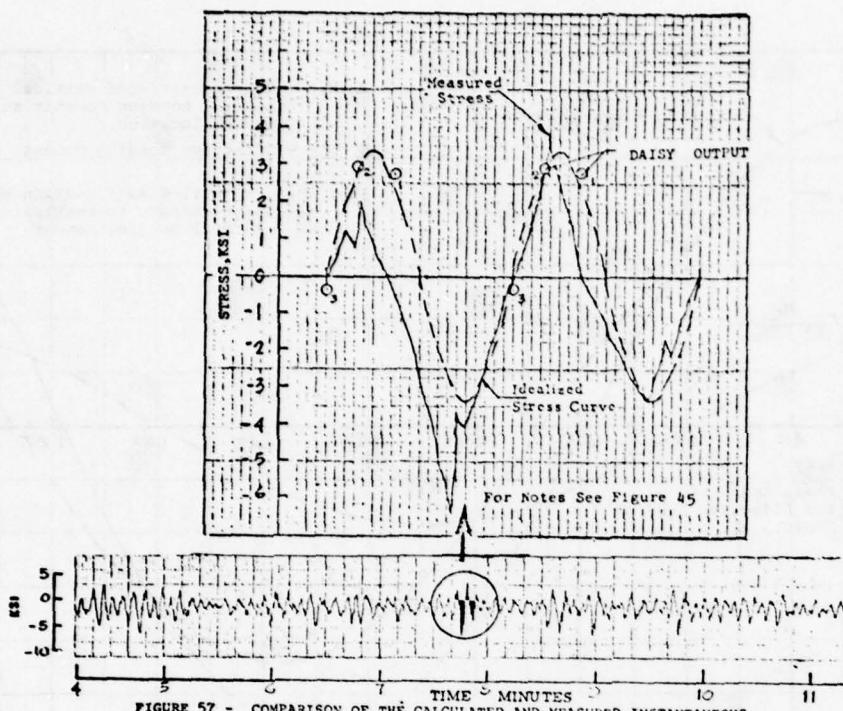


FIGURE 57 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR LS1P), WAVE CONDITION 17

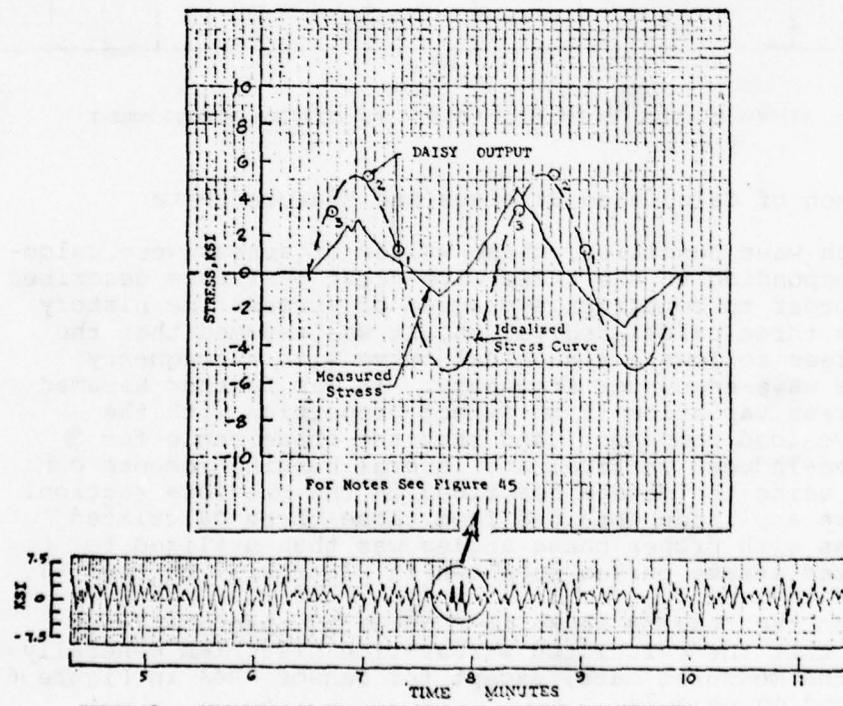


FIGURE 58 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL (SENSOR LS1S), WAVE CONDITION 17

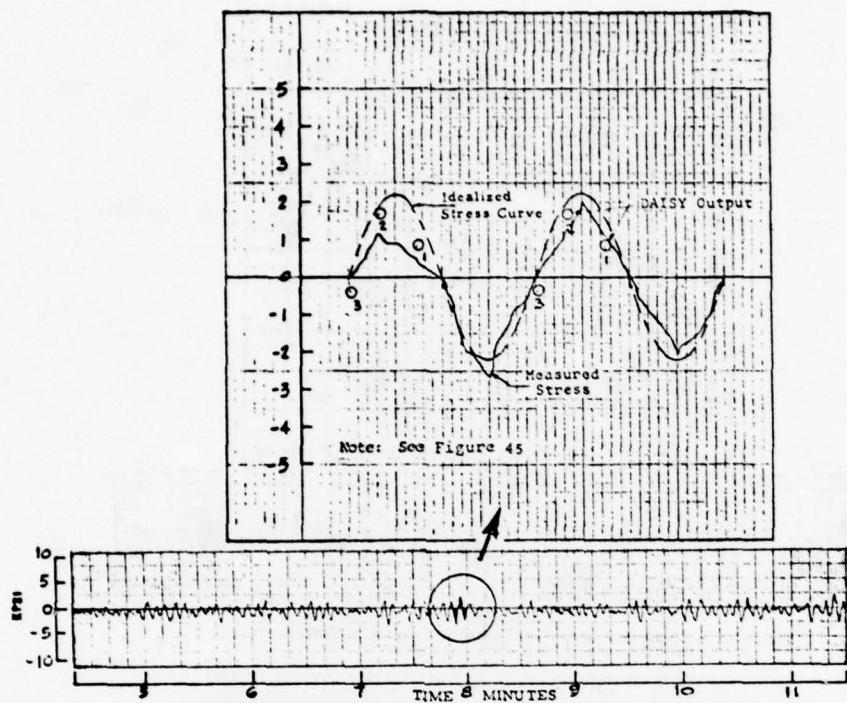


FIGURE 59 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR LSBP), WAVE CONDITION 17

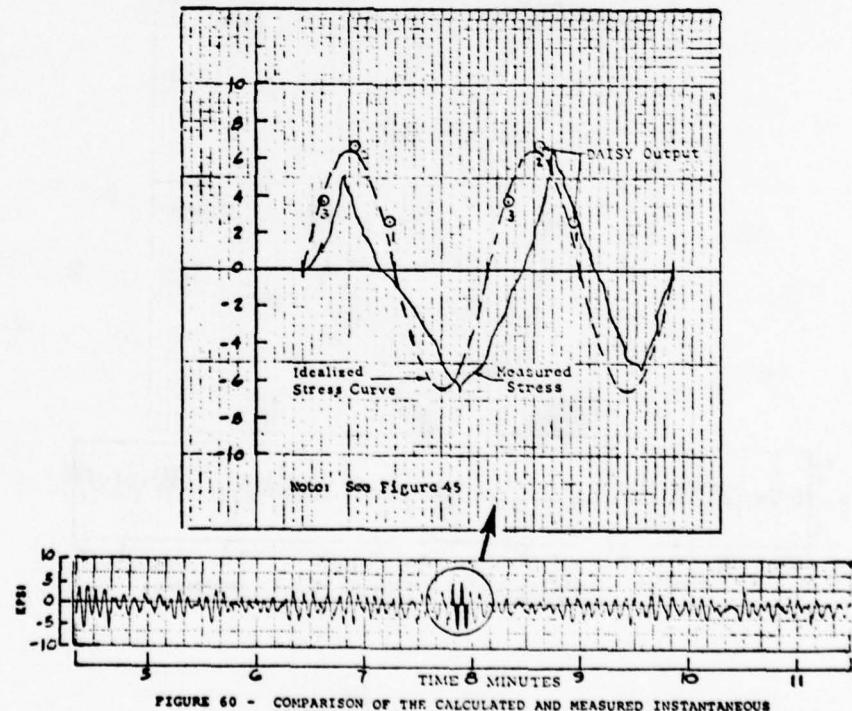


FIGURE 60 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR ARIA), WAVE CONDITION 17

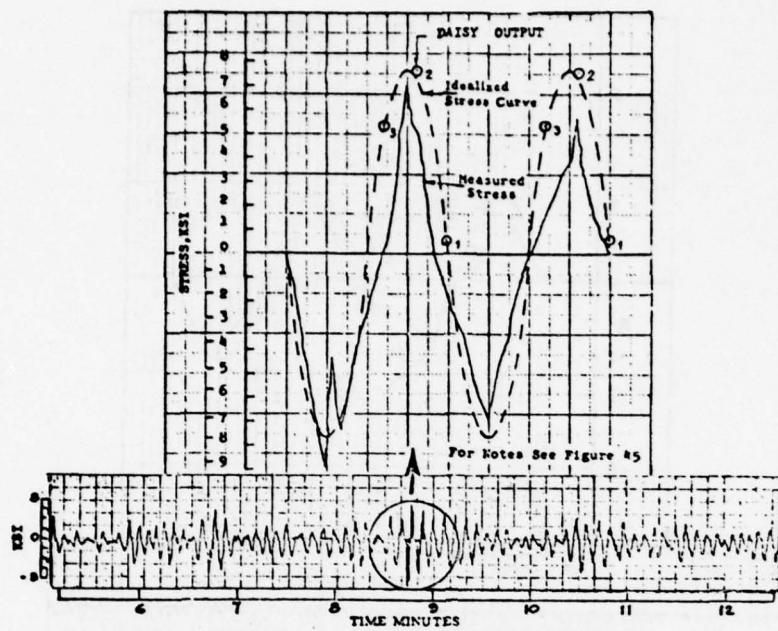


FIGURE 61 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL (SENSOR LSTP), WAVE CONDITION 18

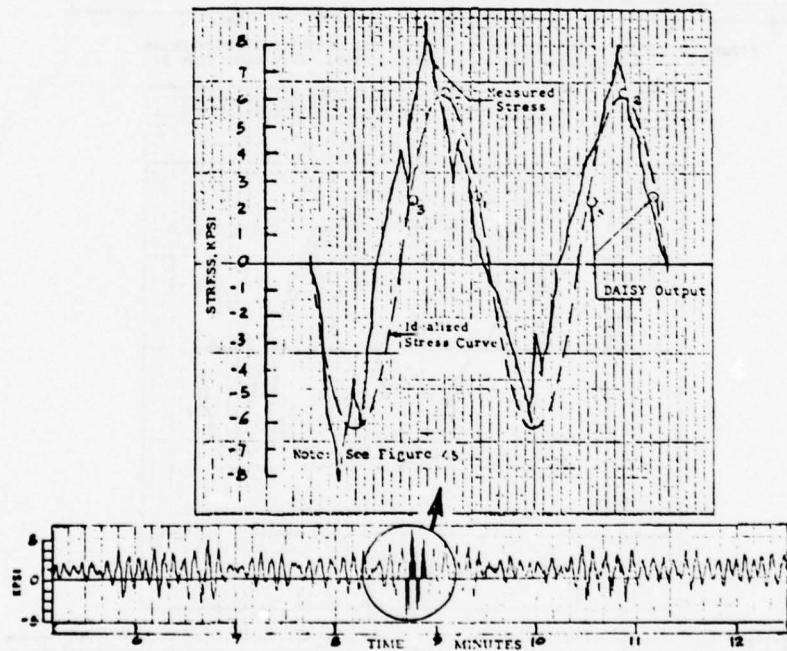


FIGURE 62 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR LSTS), WAVE CONDITION 18

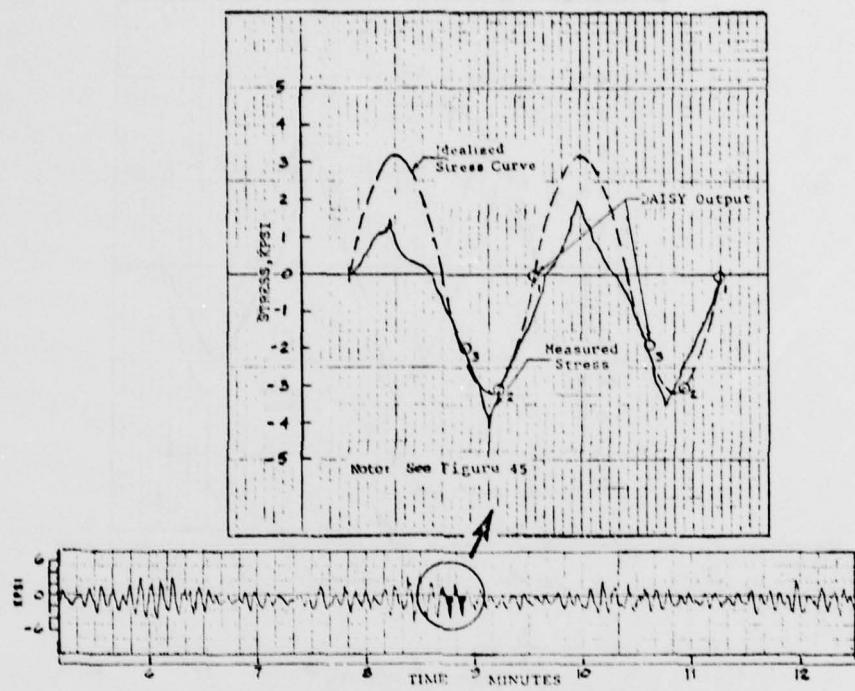


FIGURE 63 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR LSBP), WAVE CONDITION 18

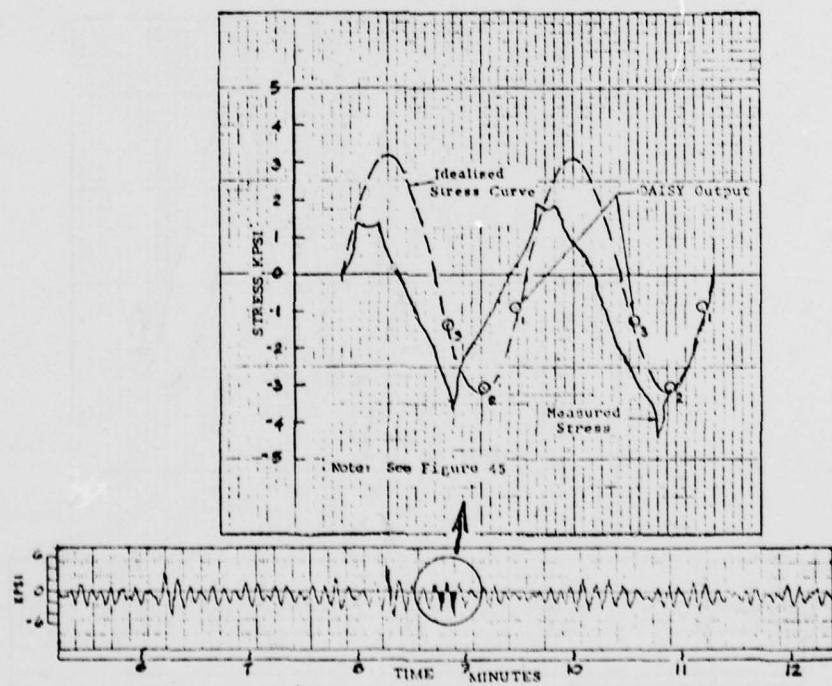


FIGURE 64 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR LSBS), WAVE CONDITION 18

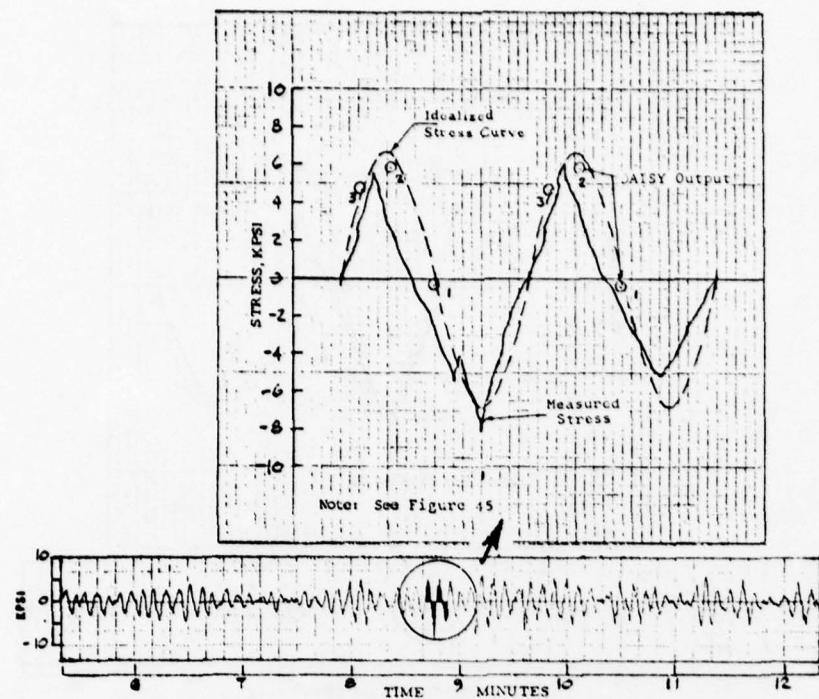


FIGURE 65 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR ARJA), WAVE CONDITION 18

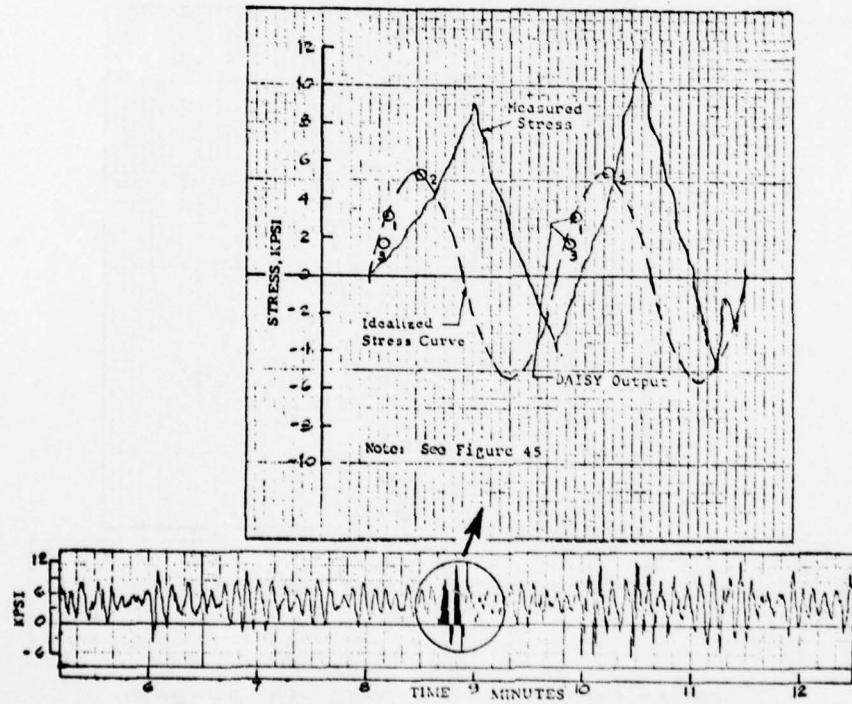


FIGURE 66 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR ARJA), WAVE CONDITION 18

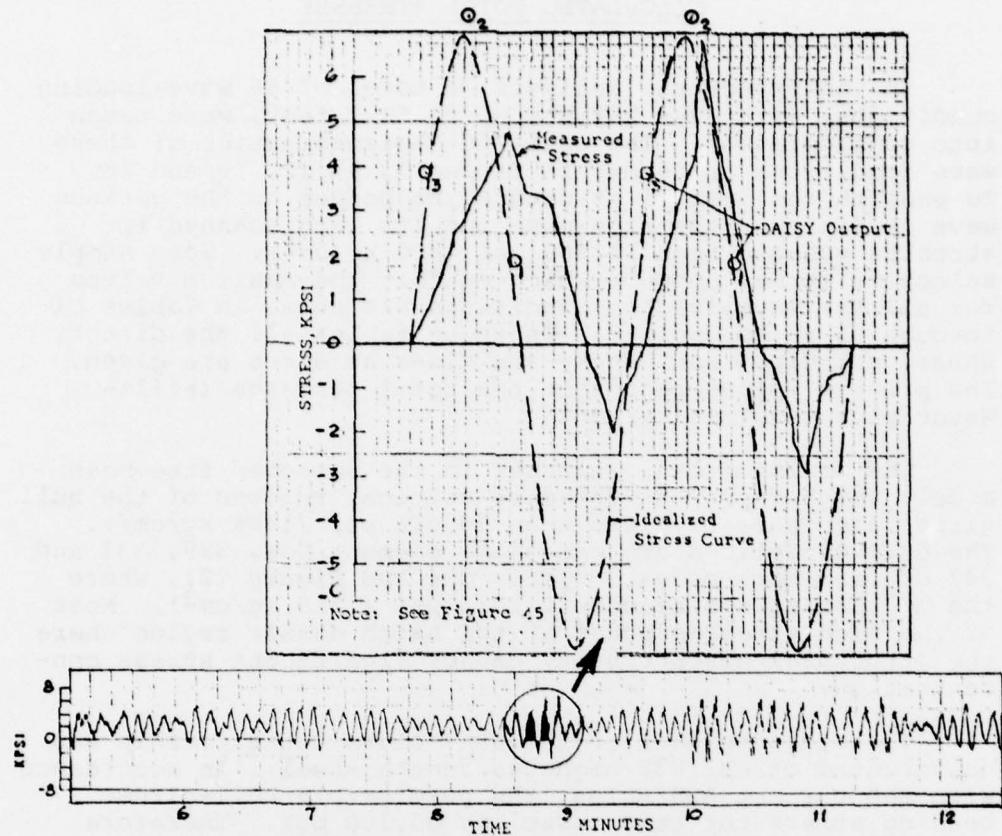


FIGURE 67 - COMPARISON OF THE CALCULATED AND MEASURED INSTANTANEOUS LONGITUDINAL STRESSES (SENSOR AR4A), WAVE CONDITION 18

#### CALCULATED TOTAL STRESSES

In the structural analysis, a total of 46 wave-loading conditions, with wave heights up to 51.2 feet, were taken into consideration. The detailed characteristics of these wave conditions are shown in Tables 7, 9, 12, 16 and 18. To examine the overall structural responses to the various wave conditions, the fine-mesh results were scanned for stresses greater than 11,360 psi (800 kg/cm<sup>2</sup>). Some sample selective stress outputs which exhibit the maximum values for all the wave-loading conditions are shown in Tables C1 through C5 in Appendix C. In those tables all the direct, shear, principal and Hencky-von Mises stresses are given. The printed values represent the total stresses (still-water plus wave) in kg/cm<sup>2</sup>.

The maximum total stresses in the selected fine-mesh models which represent the most critical regions of the hull girder are generally less than 20,000 psi (1408 kg/cm<sup>2</sup>). The only exceptions are the plate element Nos. 329, 333 and 347 of fine-mesh model 5 (Table C-4 and Figure 12), where the highest stress reaches 21,513 psi (1515 kg/cm<sup>2</sup>). Most of the high stresses occur at the hatch corner region where the structural discontinuity causes significant stress concentration.

The strength deck of the SL-7 class containership was constructed of ABS H33 higher-strength steel. In accordance with the current ABS Rules, the permissible hull-girder bending stress for this vessel is 30,100 psi. Therefore, the highest calculated total stress, which includes stress concentration, is approximately 71% of the permissible hull-girder bending stress.

Based on these calculated stresses and the full-scale stress data measured to date, no stresses greater than the permissible limit have been found.

## CONCLUSIONS

1. The overall comparison between calculated and measured stresses for the dockside calibration is generally inconclusive because of significant temperature differentials during the test and the low magnitudes of the applied loads. However, good agreement between the calculated and measured stresses was obtained when thermal effects were small. For future calibration tests, it is recommended that complete temperature data be recorded and that appreciable mechanical strains be generated in the structure.
2. The comparison of calculated and measured RMS stresses in head seas is generally satisfactory, using both the spectrum analysis approach and the equivalent regular wave approach. The correlation using the former approach which takes into account the variation in response to different wave frequencies, shows better agreement than that obtained from the latter approach. However, it should be noted that the spectrum analysis approach is valid only near the midship section where the wave-induced stresses are generally in direct proportion to wave heights.
3. The agreement of calculated and measured instantaneous stresses in head seas is generally good; the calculated stress amplitudes agree well with the mean values of the measured stresses over a time-span of two complete encounter cycles.
4. The agreement of calculated and measured instantaneous stresses in oblique seas is also generally good for the wave conditions considered.
5. Based on the results obtained from this project, it can be concluded that the existing analytical tools for predicting wave loads and structural responses are suitable to assess the overall strength of the hull girder. All the measured stress data reduced to date from the SL-7 instrumentation program and all the calculated hull-girder stresses from the present study were found to be of low magnitude. Consequently, no modifications to the present hull girder strength standard are deemed necessary.
6. Regarding the structural responses, the ABS/DAISY system in its present form is considered satisfactory for performing static analyses for either quasi-static or dynamic loads.
7. In order to assess the strength of local structures and also to improve the current methodology, further research is deemed necessary to improve the calculation method for external pressures (possibly verified by model experiments) and to improve the ship motion program (SCORES) to account for three-dimensional effects and possibly for non-linear effects with respect to wave heights.

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## APPENDIX A - ABS/DAISY COMPUTER PROGRAM SYSTEM

The primary components of the DAISY computer program system are the preprocessor programs, the DAISY finite element program itself and the postprocessor programs. The flow chart of the system is shown in Figure A-1.

### Preprocessor System

Seven principal computer programs form the nucleus of the three-dimensional ship structure preprocessing system. Each program performs a specific function and is used in a particular sequence so that the output of one program can be used as input to another. The seven programs are listed below in the order of their use:

1. **SHIPMOM** (SHIP MOMENT) calculates hull-girder shear forces, bending moments and vertical deflections for a vessel in still water or statically poised on a wave.

With the vessel's hull geometry and a description of lightship weight and cargo weights as inputs, the equilibrium draft and trim are calculated for a ship in a still-water condition or in any sinusoidal or trochoidal wave. The sinusoidal waves are directly applicable to ship dynamics and their effect on the static response of the vessel differs very little from that of the trochoidal waves traditionally used. The program calculates quasi-static lateral bending moments and torsional moments if the vessel is poised in an oblique wave. Computer line plots of the shear force diagrams, bending moment diagrams and deflections are automatically generated.

2. **SHIPMOTION** (ABS version of SCORES program) based upon the two-dimensional ship theory for six degrees of freedom predicts a vessel's motions (velocity and acceleration) in a seaway, longitudinal and lateral wave-induced bending moments and shear forces, as well as torsional moments due to waves. The seaway can be regular sinusoidal waves, irregular long-crested waves, or regular and irregular short-crested seas. The sea condition can also be represented by measured wave data. The program computes the ship motions and the dynamic components of bending moment and shear force resulting from the sea state chosen, as well as long-term values, using probability theory based on the intended service of the vessel. A sample comparison of the RAO for the vertical and lateral bending moments with model experimental data are shown in Figures A-2 and A-3.

3. DYNPRE (DYNAMIC PRESSURE) converts the data generated by SHIPI MOTION into dynamic load inputs to the DAISY finite element program. DYNPRE converts the hydrodynamic pressure distribution generated by SHIPI MOTION into a pressure distribution at the nodal points of the finite element structural model and modifies the pressures to compensate for certain discrepancies in the SHIPI MOTION hydrodynamic pressure calculation.
4. EXAM generates the ship structural finite element model for either a portion of or for the entire hull structure. Basic data generated by the program consists of nodal points and loadings applied to the elements and nodes. The program can generate a finite element model of any type of vessel.

With the vessel's hull geometry finite element nodal point locations are generated with a minimum of user input. Structural properties of bending or membrane plates, beams and/or rod members are input in a convenient tabular form from which the program automatically generates appropriate elements to represent the ship structure. The elements are generated and connected in a way that minimizes the bandwidth of the structural stiffness matrix.

EXAM automatically calculates the symmetric and anti-symmetric components of a general seaway loading. Loads in the form of pressure heads are automatically calculated and applied to the elements according to the vessel's hydrostatic cargo and sea loadings described in SHIPI MOM. The user also has the capability to input additional loads on elements and nodes.

5. EXPLOT (EXAM PLOT) provides plots of the finite element structural model generated by EXAM. The plots are isometric or two-dimensional projections. They can indicate nodal points, elements and freedom patterns. Plots of any or all of the structural portions of the vessel can be made.
6. LOADER calculates the statically consistent nodal point loads from the element pressures provided by EXAM and from the structure's own weight. Additional dynamic forces due to ship motion are calculated within the program. LOADER also calculates the out-of-balance forces for all loading conditions, and

the coefficients necessary to compute a set of pseudo-inertia forces necessary to balance the structural model. The out-of-balance forces are small in magnitude, since the vessel has been balanced in SHIPMOM, but some small re-balancing is needed to avoid artificial reaction forces from developing at the points of rigid body support.

LOADER also takes the EXAM output and rearranges it in a manner suitable for input to the DAISY program.

7. CATCH edits the data, allowing the user to make modifications to the LOADER output before input into the DAISY program. The user is able to make changes to any previously defined input data by means of additional deletions or modifications.

#### DAISY

DAISY (Displacement Automated Integrated SYstem) is a general purpose program which performs linear elastic structural analyses of two- or three-dimensional structures of almost any degree of complexity under statically applied generalized forces and thermal loads.

The DAISY program has a library of fourteen elements which can be divided into six categories: bars, beams, membrane plates, bending plates, solids and substructures.

Some special features of DAISY are:

1. Inputs to DAISY can be in standard fixed formats or in any format desired by the user. The user may also interface his own preprocessor and postprocessor programs with DAISY.
2. Intermediate results such as element stiffnesses, assembled stiffness matrices and loads, can be output on magnetic tape or disk file. Similarly, computed results such as displacements and stresses can be output on magnetic tape or disk file for further processing.
3. DAISY automatically generates the total number of degrees of freedom at each nodal point, exclusive of rigid body supports, based on the elements connected at each node.
4. DAISY is capable of generating planes of symmetry and/or antisymmetry.

5. DAISY has a RESTART capability designed to break down long computations into a series of shorter runs.

#### POSTPROCESSOR SYSTEM

The standard postprocessor system consists of DAISYOUT, a general purpose output program which prints the results calculated by DAISY with the selective output of element stresses and strains, and DAISYPLOT, which produces plots of nodal displacements and stress contours. The output data file produced by DAISY may print all or part of the results. Selective output may be printed listing all elements which are stressed above certain user-prescribed limits. Normal stresses, shear stresses and stress concentrations (Hencky-von Mises stresses) are calculated and printed.

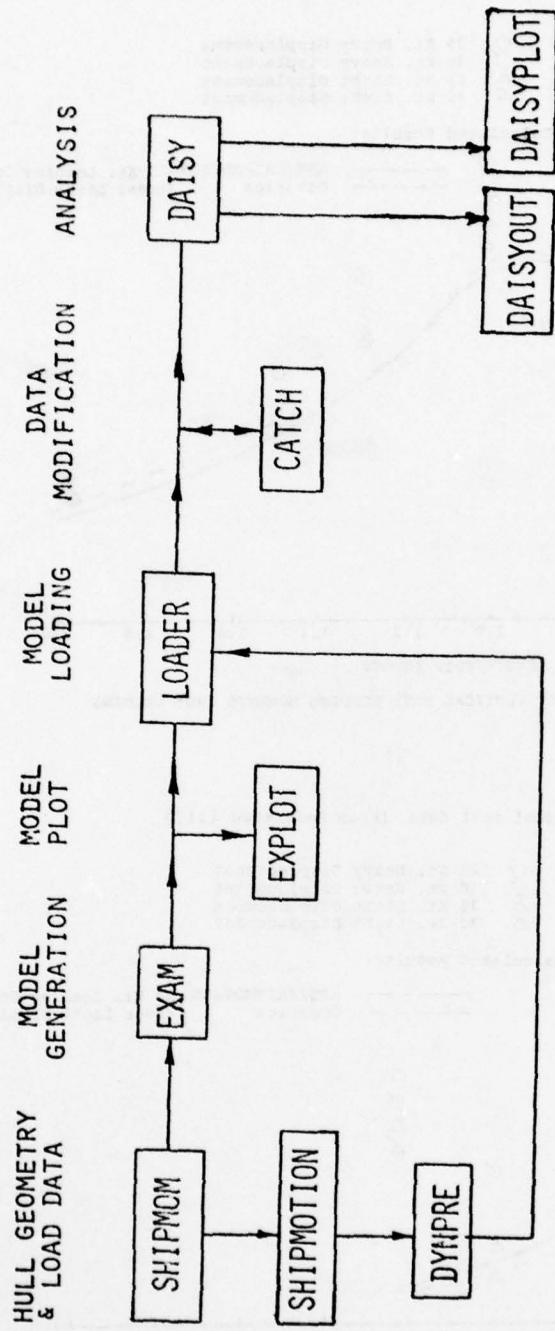


FIGURE A-1 FLOW CHART OF ABS/DAISY FINITE ELEMENT STRUCTURAL ANALYSIS SYSTEM

Model test data (From Reference [11])

○ 25 Kt. Heavy Displacement  
 ○ 30 Kt. Heavy Displacement  
 △ 25 Kt. Light Displacement  
 △ 30 Kt. Light Displacement

Calculated Results:

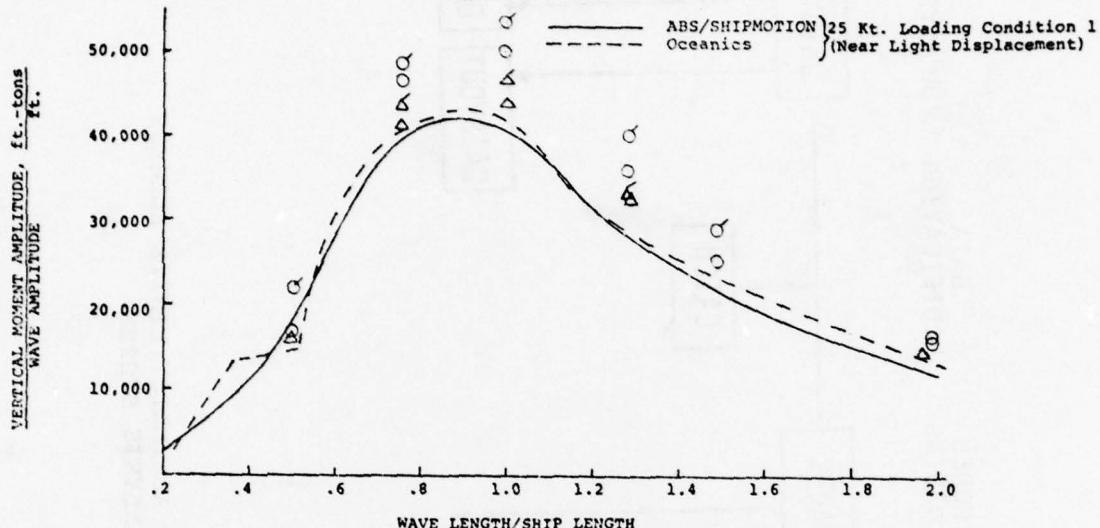


FIGURE A-2 RAO OF MIDSHIP VERTICAL WAVE BENDING MOMENTS 180° HEADING

Model test data (From Reference [11])

○ 25 Kt. Heavy Displacement  
 ○ 30 Kt. Heavy Displacement  
 △ 25 Kt. Light Displacement  
 △ 30 Kt. Light Displacement

Calculated Results:

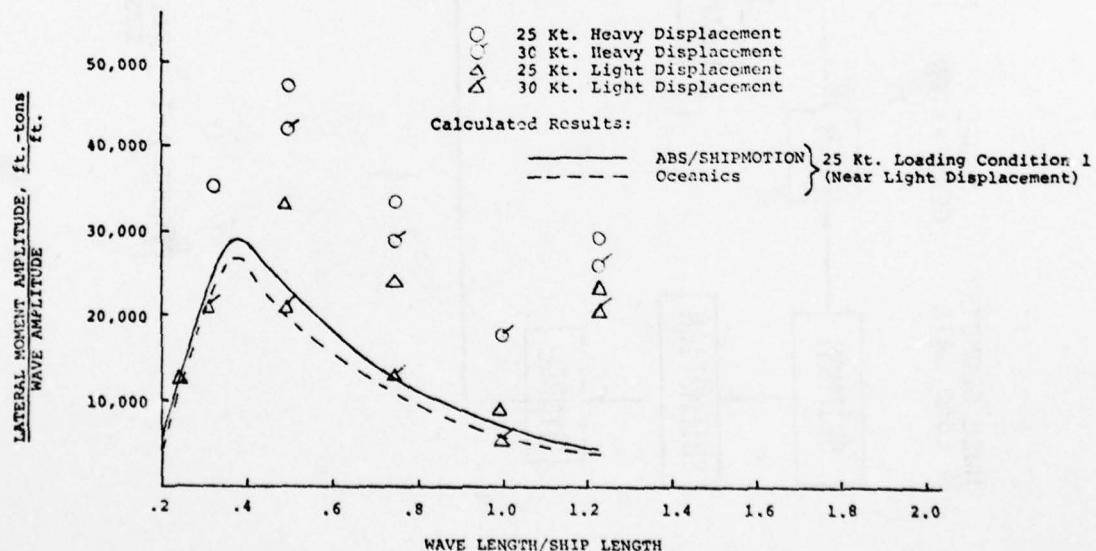


FIGURE A-3 RAO OF MIDSHIP LATERAL WAVE BENDING MOMENTS 60° HEADING

APPENDIX B - STRAIN GAGE SENSORS INSTALLED ON THE SL-7  
CONTAINERSHIP

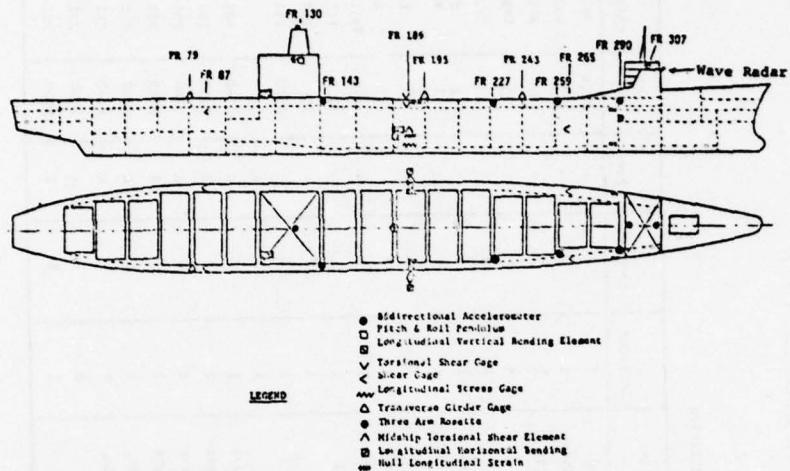


FIGURE B-1 GENERAL SENSORS LAYOUT (FROM REFERENCE [6])

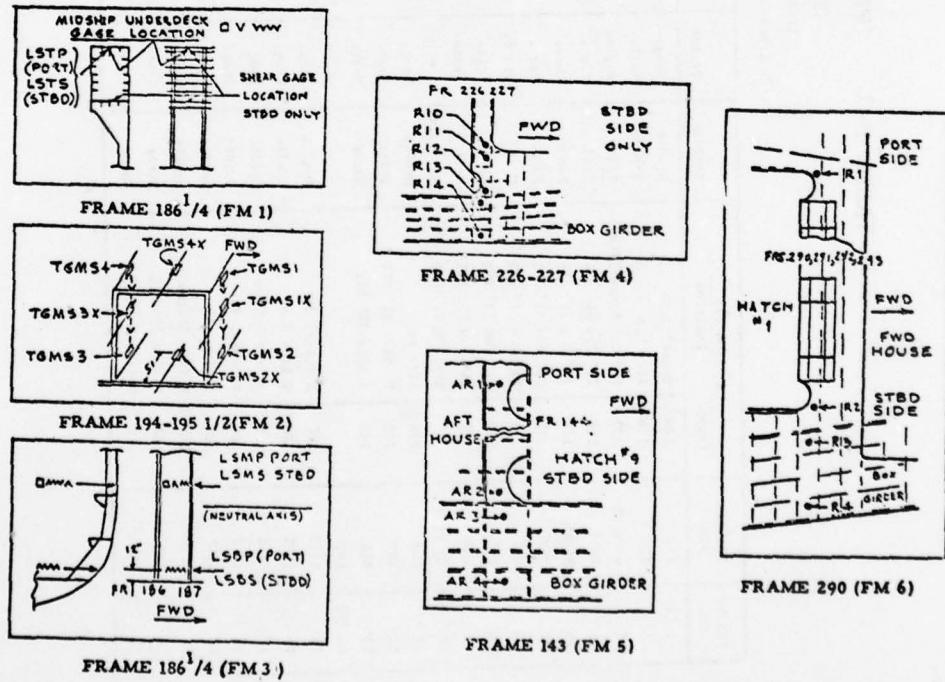


FIGURE B-2 DETAILS OF STRAIN GAGE LAYOUT  
(FROM REFERENCE [2])

AD-A069 031      AMERICAN BUREAU OF SHIPPING      NEW YORK OCEAN ENGINEERI--ETC      F/G 13/10  
COMPARISON OF STRESSES CALCULATED USING THE DAISY SYSTEM TO THO--ETC(U)  
JAN 79      H JAN, K CHANG, M E WOJNAROWSKI      DOT-CG-63176-A  
UNCLASSIFIED      RD-78005      SSC-282      NI

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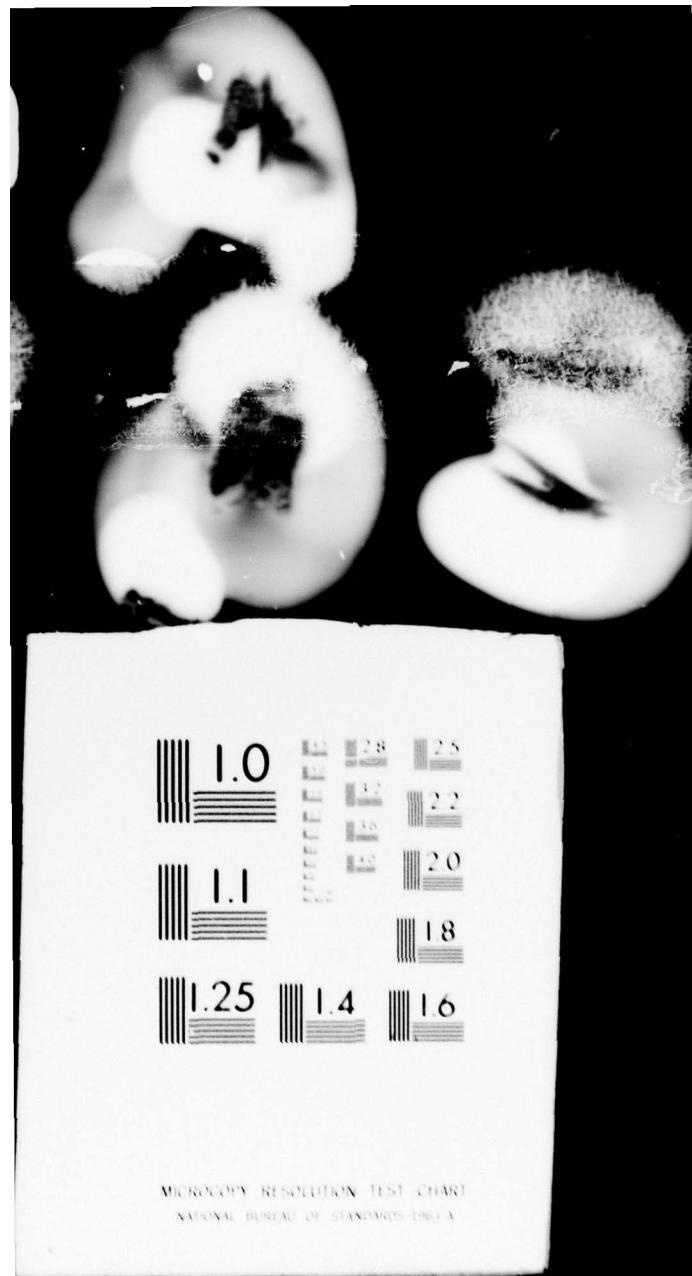


TABLE B-1 (FROM REFERENCE [6])

TABLE I  
SENSOR LIST  
72/73 Session and Calibration

Sensor No.	Sensor No.	Frame	Location (2) Position	Config.	Orient.	Sensitive to	Recorder	Channel	Node	Full Cal	Units	Circuit No.
1 (1)	LVB	186 1	Tunnel Tops	Dyadic	Long.	V. Bend.	1	1	-	8214	PSI	1
2	TSV	186 4	Side WA	Shear	Vert.	H.T. Shear	1	2	-	4991	PSI	3
1	Wave Ht.	300	Pod Deckhouse (Stbd)	Radar	Angled	Range (1)	1	3	-	3.6	Volt	-
4	Roll	178	26" Pod 31° ATT	Pend.	Trans.	Roll	1	4	-	20	Deg.	-
5	Pitch	178	26" Pod 31° ATT	Pend.	Long.	Pitch	1	5	-	20	Deg.	-
6	MAV	178	23" Pod 31° ATT	Hass	Vert.	V. Accel.	1	6	-	1	G	-
7	TAI	178	23" Pod 31° ATT	Hass	Trans.	T. Accel.	1	7	-	1	G	-
8	FAV	290	14" Pod 59° ATT	Hass	Vert.	V. Accel.	1	8	-	1	G	-
2	FAT	290	14" Pod 59° ATT	Hass	Trans.	T. Accel.	1	9	-	1	G	-
10	Op. Para.	-	RPM, Sud, Wind SBD	Multiplexer	-	Transmitter	1	10	-	3.6	Volt	-
11	LHR	186 1	Side MA	Dyadic	Long.	H. Bend	1	11	-	8214	PSI	2
12	SPP	265	P Side 32° ATT	Shear	Vert.	Shear	1	12	-	5000	PSI	4
13	STS	265	S Side 32° ATT	Shear	Vert.	Shear	1	13	-	5000	PSI	4
14 (1)	LVB	186	S Tunnel Top	Dyadic	Long.	N. Stress	2	1	4	-	-	-
15	LST5	186	S Tunnel Top	Dyadic	Long.	N. Stress	2	2	4	8240	PSI	5
16	LSS5	186	S Side N.A.	Dyadic	Long.	N. Stress	2	3	4	8240	PSI	5
17	LSS5	186	S Side Bottom	Dyadic	Long.	N. Stress	2	4	4	8240	PSI	5
13	LST7	186	P Tunnel Top	Dyadic	Long.	N. Stress	2	5	4	8240	PSI	5
19	LSS7	186	P Side WA	Dyadic	Long.	N. Stress	2	6	4	8240	PSI	5
22	LST7	186	P Side Bottom	Dyadic	Long.	N. Stress	2	7	4	8240	PSI	5
21	SAS2	87	P Side 26° ATT	Shear	Vert.	Shear	2	8	4	5000	PSI	4
22	SAS5	87	S Side 26° ATT	Shear	Vert.	Shear	2	9	4	5000	PSI	4

TABLE B-1 (CONT'D)

SENSOR LIST  
72/73 Season and Calibration

Sensor No.	Sensor No.	Location (2) Frame	Location (2) Position	Config.	Orient.	Sensitive to	Recorder	Channel	Mode	Palt Cal	Units	Circuit No.
23	F217	307	Level 04 CL	Mass	Vert.	V. Accel.	2	10	A	+1 (4)	6	-
24	F217	307	Level 04 CL	Mass	Trans.	T. Accel.	2	11	A	+1	6	-
25	A211	139	Level 05 1" P	Mass	Long.	Lo. Accel.	2	10 (a)	A	+1	6	-
26	A211T	130	Level 05 1" P	Mass	Trans	T. Accel.	2	11 (a)	A	+1	6	-
27	B217	186 1/4	S Tunnel Top	Shear	Long.	Shear	2	12	A	5000	PSI	4
28	B219	185 1/4	S Tunnel Bot	Shear	Long.	Shear	2	13	A	5000	PSI	4
29 (1)	L12						2	1	B			
30	A2-1A	143	Port Side Girder	Single	Long.	N. Strain	2	2	B	334.6	psi	6
31	A2-13	143	Star Deck Cutout	Single	Diag.	N. Strain	2	3	B	334.6	psi	6
32	A2-1C	143	Under Deck	Single	Trans.	N. Strain	2	4	B	334.6	psi	6
33	A2-2A	143	Stbd Side Girder	Single	Long.	N. Strain	2	5	B	334.6	psi	6
34	A2-2B	143	Star Deck Cutout	Single	Diag.	N. Strain	2	6	B	334.6	psi	6
35	A2-2C	143	Under Deck	Single	Trans.	N. Strain	2	7	B	334.6	psi	6
36	A2-3A	143	Stbd Tunnel	Single	Long.	N. Strain	2	8	B	334.6	psi	6
37	A2-3B	143	In Board	Single	Diag.	N. Strain	2	9	B	334.6	psi	6
38	A2-3C	143	Under Deck	Single	Trans.	N. Strain	2	10	B	334.6	psi	6
39	A2-4A	143	Stbd Tunnel	Single	Long.	N. Strain	2	11	B	334.6	psi	6
40	A2-4B	143	Out Board	Single	Diag.	N. Strain	2	12	B	334.6	psi	6
41	A2-4C	143	Under Deck	Single	Trans.	N. Strain	2	13	B	334.6	psi	6

TABLE B-1 (CONT'D)

## SENSOR LIST

## 72/73 Session and Calibration

Sensor No.	Sensor No.	Location(s) Frame	Position	Config.	Ornt.	Sensitive to	Recorder	Channel	Mode	Poll Cnt	Units	Circuit No.
42 (1)	178	291	Port Side Gird	Single	Long.	N. Strain	2	1	C	234.6	μv/v	6
43	86A	291	Year Deck Cutout	Single	Diag.	N. Strain	2	2-13	C	234.6	μv/v	
44	87B	291	Under Deck	Single	Trans.	N. Strain	2	VIA	C	234.6	μv/v	
45	87C	291	Stbd Side Gird	Single	Long.	N. Strain	2	858	C	234.6	μv/v	
46	87A	291	Year Deck Cutout	Single	Long.	N. Strain	2	2-13	C	234.6	μv/v	
47	828	291	Under Deck	Single	Diag.	N. Strain	2	VIA	C	234.6	μv/v	
48	83C	291	Stbd Tunnel	Single	Trans.	N. Strain	2	859	C	234.6	μv/v	
49	132A	231	In Board	Single	Long.	N. Strain	2	2-13	C	234.6	μv/v	
50	132B	291	Under Deck	Single	Diag.	N. Strain	2	VIA	C	234.6	μv/v	
51	33C	291	Stbd Tunnel	Single	Trans.	N. Strain	2	858	C	234.6	μv/v	
52	133A	231	Out Board	Single	Long.	N. Strain	2	2-13	C	234.6	μv/v	
53	133B	291	Under Deck	Single	Diag.	N. Strain	2	VIA	C	234.6	μv/v	
54	133C	291	Stbd Side Gird	Single	Trans.	N. Strain	2	858	C	234.6	μv/v	
55	133A	230	Out Corn. Rat 2	Single	Long.	N. Strain	2	2-13	C	234.6	μv/v	
56	133B	230	Under Deck	Single	Diag.	N. Strain	2	VIA	C	234.6	μv/v	
57	133C	230	Stbd Side Gird	Single	Trans.	N. Strain	2	859	C	234.6	μv/v	
58	134A	230	Out Corn. Rat 2	Single	Long.	N. Strain	2	2-13	C	234.6	μv/v	
59	134B	230	Under Deck	Single	Diag.	N. Strain	2	VIA	C	234.6	μv/v	
60	134C	230	Stbd Side Gird	Single	Trans.	N. Strain	2	858	C	234.6	μv/v	
61	137A	230	Year Deck Cutout	Single	Long.	N. Strain	2	2-12	C	234.6	μv/v	
62	137B	230	Under Deck	Single	Diag.	N. Strain	2	VIA	C	234.6	μv/v	
63	137C	230	Stbd Tunnel	Single	Long.	N. Strain	2	858	C	234.6	μv/v	
64	133A	230	In Board	Single	Diag.	N. Strain	2	VIA	C	234.6	μv/v	
65	133C	75A	Under Deck	Single	Trans.	N. Strain	2	854	C	234.6	μv/v	

TABLE B-1 (CONT'D)  
SENSOR LIST  
72/73 Season and Calibration

Sensor No.	Sensor Ssn.	Location (2) Frame Position	Config.	Orient.	Sensitive to	Recorder	Channel	Mode	Full Cal.	Units	Circuit No.
67	R9A	258 Stbd Tunnel	Single	Long.	H. Strain	2	2-13	C	334.4	g/cm <sup>2</sup>	-----
68	R9B	258 Out Board	Single	Diag.	H. Strain	2	-----	VIA	334.4	g/cm <sup>2</sup>	-----
69	R9C	258 Under Deck	Single	Trans.	H. Strain	2	-----	RSS	334.4	g/cm <sup>2</sup>	-----
70	R10A	226 Stbd Side Gird	Single	Long.	H. Strain	2	2-13	C	334.4	g/cm <sup>2</sup>	-----
71	R10B	226 In Cntr. Mat 4	Single	Diag.	H. Strain	2	-----	VIA	334.4	g/cm <sup>2</sup>	-----
72	R10C	226 Under Deck	Single	Trans.	H. Strain	2	-----	RSS	334.4	g/cm <sup>2</sup>	-----
73	R11A	226 Stbd Side Gird	Single	Long.	H. Strain	2	2-13	C	334.4	g/cm <sup>2</sup>	-----
74	R11B	226 Out Cntr Mat 4	Single	Diag.	H. Strain	2	-----	VIA	334.4	g/cm <sup>2</sup>	-----
75	R11C	226 Underdeck	Single	Trans.	H. Strain	2	-----	RSS	334.4	g/cm <sup>2</sup>	-----
76	R11A	226 Stbd Side Gird	Single	Long.	H. Strain	2	2-13	C	334.4	g/cm <sup>2</sup>	-----
77	R11B	226 Star Deck Cutout	Single	Diag.	H. Strain	2	-----	VIA	334.4	g/cm <sup>2</sup>	-----
78	R11C	226 Underdeck	Single	Trans.	H. Strain	2	-----	RSS	334.4	g/cm <sup>2</sup>	-----
79	R12A	226 Stbd Tunnel	Single	Long.	H. Strain	2	2-13	C	334.4	g/cm <sup>2</sup>	-----
80	R12B	226 In Board	Single	Diag.	H. Strain	2	-----	VIA	334.4	g/cm <sup>2</sup>	-----
81	R12C	226 Under Deck	Single	Trans.	H. Strain	2	-----	RSS	334.4	g/cm <sup>2</sup>	-----
82	R13A	226 Stbd Tunnel	Single	Long.	H. Strain	2	2-13	C	334.4	g/cm <sup>2</sup>	-----
83	R13B	226 Out Board	Single	Diag.	H. Strain	2	-----	VIA	334.4	g/cm <sup>2</sup>	-----
84	R13C	226 Under Deck	Single	Trans.	H. Strain	2	-----	RSS	334.4	g/cm <sup>2</sup>	-----
85 (1)	UX9	244 Top	Single	Trans.	H. Stress	2	1	-----	-----	-----	-----
85	TC791	244 Top	Single	Trans.	H. Stress	2	2	-----	10038	PSI	-----

TABLE B-1 (CONT'D)

SENSORS LIST  
72/73 Session and Calibration

Marker No.	Sensor No.	Frame	Location (2) Position	Config.	Orient.	Sensitive to	Recorder	Channel	Mode	Full Cal.	Units	Circuit No.
87	ULSST	289	S Side 1' BT	Single	Long.	N. Stress	2	2 (a)	2	10038	PSI	6
83	TGFS2	264	Pad Bot.	Single	Trans.	N. Stress	2	3	3	10038	PSI	6
87	ULSS3	289	S Side 1' ATT	Single	Long.	N. Stress	2	3 (a)	3	10038	PSI	6
27	TGFS3	262	Aft Bot.	Single	Trans.	N. Stress	2	4	4	10038	PSI	6
91	ULSP7	259	P Side 1' BT	Single	Long.	N. Stress	2	4 (a)	4	10038	PSI	6
52	TGFS4	262	Aft Top	Single	Trans.	N. Stress	2	5	5	10038	PSI	6
23	ULSP8	259	P Side 1' ATT	Single	Long.	N. Stress	2	5 (a)	5	10038	PSI	6
94	TGFS1	196	Pad Gird. Top	Single	Trans.	N. Stress	2	6	6	10038	PSI	6
35	TGFS2	196	Pad Gird Bot.	Single	Trans.	N. Stress	2	7	7	10038	PSI	6
95	TGFS3	194	Aft Gird Bot.	Single	Trans.	N. Stress	2	8	8	10038	PSI	6
97	TGFS4	194	Aft Gird Top	Single	Trans.	N. Stress	2	9	9	10038	PSI	6
94	TGFS1X	194	Pad Gird Mid	Single	Trans.	N. Stress	2	6 (a)	6	10038	PSI	6
99	TGFS2X	195	Bot Gird Mid	Single	Trans.	N. Stress	2	7 (a)	7	10038	PSI	6
123	TGFS3X	194	Aft Gird Mid	Single	Trans.	N. Stress	2	8 (a)	8	10038	PSI	6
121	TGFS4X	195	Top Gird Mid	Single	Trans.	N. Stress	2	9 (a)	9	10038	PSI	6
192	TGFS1X	196	Pad Gid Q Top	Shear	Trans.	Shear	2	6 (a)	6	5000	PSI	6
193	TGFS2X	196	Pad Gid Q Bot	Shear	Trans.	Shear	2	7 (a)	7	5000	PSI	6
194	TGFS3X	194	Aft Gid Q Bot	Shear	Trans.	Shear	2	8 (a)	8	5000	PSI	6
195	TGFS4X	194	Aft Gid Q Top	Shear	Trans.	Shear	2	9 (a)	9	5000	PSI	6
165	TGAS1	80	Pad Top	Single	Trans.	N. Stress	2	10	10	10038	PSI	6
127	TGAS2	80	Pad Bot	Single	Trans.	N. Stress	2	11	11	10038	PSI	6
128	TGAS3	79	Aft Bot	Single	Trans.	N. Stress	2	12	12	10038	PSI	6
169	TGAS4	79	Aft Top	Single	Trans.	N. Stress	2	13	13	10038	PSI	6

APPENDIX C - SELECTIVE STRESS OUTPUT

TABLE C-1 PLATE NO. 01, PCAT  
SELECTIVE STRESS OUTPUT FOR LOAD CASE 17

ELIN	CONNER POINTS	EL. 4TH	SIGMA X	SIGMA Y	SIGMA Z	TAU XY	SIGMA 1	SIGMA 2	TAU ZX	MICROSTRESS	SIGMA 3
STRESSES IN QUADRILATERAL BENDING PLATE (1000000 ELEM)											
416 1949 2098 2099 2100 2101 7	99.8	871.4	6.6	-165.7	677.9	93.7	392.1	846.9	LCL X-05.0		
	53.3	944.9	6.6	-165.8	950.2	45.6	451.1	927.1			
417 2000 1999 2099 2100 2101 7	26.9	1038.4	0.0	-155.2	1643.0	26.7	508.2	1629.9	LCL X-06.3		
	60.9	1567.6	0.0	-167.9	1681.1	45.5	515.8	1651.2			
422 1980 2000 2100 2098 2101 7	0.0	-71.2	668.4	-156.6	971.6	-34.3	563.6	669.2	LCL Y-04.6		
	90.6	925.0	0.0	-92.4	994.6	81.2	456.6	926.4			
423 2086 2094 1996 1986 2101 7	0.0	-162.1	997.9	1.1	997.9	-146.1	548.9	1677.9	LCL Y-02.0		
	181.6	1070.3	0.0	-5.8	1050.4	181.6	456.4	1611.9			
STRESSES IN BENDING TRIANGULAR PLATES (1000000 ELEM)											
430 1986 2076 2096 2106 2107 7	9.0	-25.0	953.4	-7.1	958.6	720.6	470.3	989.7	LCL Y-06.9		
	0.0	-13.2	953.4	-5.4	955.6	-16.2	466.9	964.3			
431 1996 2096 2081 2107 7	0.0	-86.8	1087.3	-75.8	1087.3	-16.9	587.1	1132.3	LCL Y-07.7		
	0.0	-51.0	1091.4	-32.3	1092.3	-51.9	572.2	1139.3			
STRESSES IN QUADRILATERAL BENDING PLATE (1000000 ELEM)											
432 2226 2234 2237 2237 217 7	22.0	-2123.2	0.0	215.3	433.0	-424.2	520.6	720.5	LCL X-18.6		
	51.4	-2265.2	0.0	211.3	455.8	-598.7	527.3	916.0			
439 2256 2269 2276 2260 217 7	246.8	-2115.3	0.0	445.5	545.2	-415.8	460.5	834.8	LCL X-34.0		
	273.4	-2119.4	0.0	440.0	560.3	-404.3	412.3	839.0			
440 2264 2274 2248 2247 217 7	695.3	-75.2	0.0	-893.9	1001.8	-391.7	676.8	1246.8	LCL X-28.3		
	719.9	-12.7	0.0	-899.1	1047.3	-340.1	639.7	1252.5			
441 2069 2068 2080 2270 217 7	697.5	64.2	0.0	393.6	1031.7	-200.0	665.8	1259.9	LCL X-31.3		
	151.6	151.6	0.0	393.7	1076.3	-262.2	664.3	1232.2			
457 2097 2094 2196 2197 217 7	1026.5	0.0	227.3	1027.3	181.6	642.8	980.5	LCL X-08.3			
	1067.3	0.0	-46.7	1069.3	41.8	463.9	987.3				
458 2098 2097 2197 2198 217 7	77.0	913.3	0.0	-311.1	854.5	75.8	359.4	817.3	LCL X-07.7		
	86.9	929.0	0.0	-38.4	930.7	65.2	432.8	879.9			
459 2074 2068 2174 2169 217 7	98.0	846.3	0.0	-259.7	869.9	93.4	347.4	816.2	LCL X-12.4		
	41.7	935.1	0.0	-50.4	937.9	45.4	446.1	935.9			

- a. Each model point is represented by a four-digit number, the first two digits indicate the station number of the finite element model and the last two digits are the local model number.  
b. Under the local coordinate system is used, SIGMA X is the direct stress in the direction from first to second nodal point. SIGMA Y is the direct stress orthogonal with SIGMA X.  
c. Under the global coordinate system is used, SIGMA X, SIGMA Y and SIGMA Z are stress in the directions as shown in Figure 2.  
d. If the global angle is printed, the coordinate stresses are in global system. If not, they are in element local coordinate system the first line indicates stresses and angles at the top of the plate, the second line indicates the results at the plate bottom.

TABLE C-1 FINE MESH #1 (Continued)

## TABLE I (Continued)

SELECTIVE STRATEGIC OUTSOURCING FOR LEAN MANUFACTURING

### SELECTIVE STRESS OUTPUT FOR LOAD CASE 53

STRESS: KG/CM<sup>2</sup>

STRESSES IN BENDING TRIMARAN PLATES (72193 ELM)

473	2070	2180	2080	17	7	0.0	27.1	786.5	95.3	798.3	15.4	391.5	790.7	GL8	Y	83.0	
474	2070	2180	2080	17	7	0.0	72.1	814.0	115.7	831.6	54.4	382.6	805.8	GL8	Y	81.3	
475	2076	2186	2086	17	7	0.0	-221.7	950.7	72.5	956.1	-27.0	491.5	969.9	GL8	Y	85.8	
476	2076	2186	2086	17	7	0.0	-222.2	944.1	85.2	951.5	-29.7	490.6	966.7	GL8	Y	85.0	
477	2096	2196	2081	17	7	0.0	-91.8	1072.0	94.7	1079.7	-99.5	589.6	1132.7	GL8	Y	85.4	
485	2196	2281	2081	17	7	0.0	-55.2	1076.1	125.3	1089.9	-68.9	579.4	1125.9	GL8	Y	83.8	
486	2196	2281	2081	17	7	0.0	-38.8	1197.2	74.0	1201.6	-43.3	622.4	1223.8	GL8	Y	86.6	
487	2196	2281	2081	17	7	0.0	-25.6	1139.0	92.1	1146.2	-32.9	569.5	1163.0	GL8	Y	85.5	
<hr/>																	
STRESSES IN QUADRILATERAL BENDING PLATE (CUABA ELM)																	
489	2197	2196	2296	2297	17	7	20.2	1014.9	0.0	1015.0	20.1	497.5	1065.1	LCL	X	89.4	
490	2198	2197	2297	2298	17	7	28.0	1025.6	0.0	1026.0	27.6	499.2	1022.5	-	-	-88.8	
491	2199	2196	2296	2298	17	7	77.5	972.9	0.0	973.8	76.6	448.6	937.9	LCL	X	-88.1	
492	2200	2199	2299	2300	17	7	25.4	952.7	0.0	954.3	23.6	465.2	942.6	-	-	-87.6	
493	2199	2196	2296	2299	17	7	104.1	993.2	0.0	998.3	99.1	449.6	952.6	LCL	X	-95.7	
494	2199	2196	2296	2299	17	7	-9.7	945.1	0.0	948.7	-13.3	401.0	955.5	-	-	-86.5	
495	2197	2200	2300	2280	17	7	-20.9	1033.5	0.0	-113.2	1045.5	-32.9	539.2	1062.4	LCL	X	-83.9
496	2197	2198	2298	2300	17	7	64.8	1053.3	0.0	-75.8	1059.1	59.1	500.0	1030.9	-	-	-85.6
497	2190	2200	2300	2280	17	7	0.0	-25.0	970.1	-102.3	920.5	-35.5	508.0	998.7	GL8	Y	-64.2
498	2286	2296	2196	2186	17	7	0.0	75.8	993.8	-150.5	1017.9	51.8	483.1	993.0	-	-	-80.9
499	2286	2296	2196	2186	17	7	0.0	-140.9	969.3	45.0	971.2	-141.8	256.5	1049.3	GL8	Y	87.7
500	2286	2296	2196	2186	17	7	0.0	125.6	1036.9	58.8	1040.7	121.8	459.4	985.4	-	-	-86.3

TABLE C-1  
FINE MESH #1  
(Cont'd)  
SELECTIVE STRESS OUTPUT FOR LOAD CASE 53  
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ELEM	CORNER POINTS NO 1ST 2ND 3RD 4TH DAISY BOT	SIGMA X			SIGMA Y			SIGMA Z			TAU			SIGMA 1			SIGMA 2			TAU MAX			HENCKY-YOUNG MIS. STRESS		ANG SIG1 .....AXIS				
		1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25			
STRESSES IN BENDING TRIANGULAR PLATES (TR153 ELM)																													
503 2180 2270 2260 17 T																													
503	2180	2270	2260	17	T	0.0	-47.8	624.0	102.0	835.8	-61.6	446.7	869.2	GLB	Y	83.4													
						0.0	140.7	709.7	93.5	920.9	129.5	392.7	663.5																
505 2186 2276 2266 17 T																													
505	2186	2276	2266	17	T	0.0	-12.8	940.4	30.6	941.4	-13.8	477.6	948.4	GLB	Y	88.2													
						0.0	26.2	956.2	51.3	959.0	25.4	466.8	946.6																
506 2156 2296 2281 17 T																													
506	2156	2296	2281	17	T	0.0	-43.7	1054.0	102.4	1063.5	-53.2	558.3	1091.1	GLB	Y	84.7													
						0.0	-2.8	1057.3	113.2	1069.3	-14.8	562.0	1076.7																
STRESSES IN QUADRILATERAL BENDING PLATE (QUAB4 ELM)																													
508 2297 2296 2396 2397 17 T																													
508	2297	2296	2396	2397	17	T	-16.6	1019.5	0.0	23.5	1011.0	-17.2	514.1	1019.7	LCL	X	88.7												
							13.5	1045.0	0.0	-5.1	1045.0	13.4	515.8	1039.3															
509 2298 2297 2397 2398 17 T																													
509	2298	2297	2397	2398	17	T	45.5	1075.1	0.0	-25.2	1075.7	44.9	515.4	1056.0	LCL	X	88.6												
							-53.3	1023.0	0.0	-30.8	1023.9	-54.2	539.0	1052.1															
510 2299 2298 2398 2399 17 T																													
510	2299	2298	2398	2399	17	T	75.8	1044.5	0.0	-71.3	1099.4	70.8	514.3	1065.8	LCL	X	86.0												
							-84.2	1025.9	0.0	-69.9	1030.3	-68.6	559.5	1077.3															
511 2290 2299 2399 2400 17 T																													
511	2290	2299	2399	2400	17	T	-62.7	1035.7	0.0	-124.7	1049.7	-76.7	563.2	1060.1	LCL	X	83.6												
							54.6	1100.9	0.0	-68.1	1108.3	47.3	530.5	1085.4															
517 2289 2300 2400 2380 17 T																													
517	2289	2300	2400	2380	17	T	0.0	-50.6	994.0	-86.9	1001.2	-57.8	529.5	1031.3	GLB	Y	85.3												
							72.0	1026.0	-1123.3	1049.2	59.7	494.7	1020.6																
518 2386 2396 2296 2286 17 T																													
518	2386	2396	2296	2286	17	T	0.0	-173.3	928.5	72.5	933.2	-178.1	555.7	1033.8	GLB	Y	86.2												
							176.3	1076.7	44.8	1080.9	174.1	493.4	1005.2																
525 2250 2350 2360 2260 17 T																													
525	2250	2350	2360	2260	17	T	0.0	-0.3	780.9	107.3	795.3	-14.7	405.0	802.8	GLB	Y	82.3												
							92.1	849.8	115.0	266.9	75.0	395.9	831.9																
526 2240 2340 2350 2250 17 T																													
526	2240	2340	2350	2250	17	T	0.0	-31.7	712.9	108.0	728.2	-47.1	387.7	752.9	GLB	Y	81.9												
							130.8	798.9	107.7	815.8	113.9	351.0	765.2																
527 2270 2370 2380 2280 17 T																													
527	2270	2370	2380	2280	17	T	0.0	-56.1	879.7	80.2	886.5	-62.9	474.7	919.6	GLB	Y	85.1												
							121.5	985.8	92.2	995.5	111.7	441.9	944.6																
529 2246 2346 2356 2256 17 T																													
529	2246	2346	2356	2256	17	T	0.0	43.1	819.9	63.8	825.1	37.9	393.6	806.8	GLB	Y	85.3												
							742.7	742.7	60.9	753.3	-58.1	405.7	764.0																

TABLE C-2 FINE MESH #3, PORT

## SELECTIVE STRESS OUTPUT FOR LOAD CASE 53

STRESS : KC/CM<sup>2</sup>

ELEM	CORNER POINTS	L.C.	TOP	SIGMA X	SIGMA Y	SIGMA Z	TAU	SIGMA 1	SIGMA 2	TAU MAX	HENCKY-YOUNG MIS.S.TRESS	ANG SIGI ***.AXIS
251	1701 1704 1601 17	1	0.0	-214.5	-879.9	0.0	-342.3	-69.8	-1024.5	477.3	991.4 LCL X-22.9	

## STRESSES IN BENDING TRIANGULAR PLATES (TRI63 ELM)

251	1701 1704 1601 17	1	0.0	-214.5	-879.9	0.0	-342.3	-69.8	-1024.5	477.3	991.4 LCL X-22.9	
				-123.7	-871.6	0.0	-326.6	0.5	-971.6	486.2	972.1 -20.9	

## STRESSES IN QUADRILATERAL BENDING PLATE (QUA64 ELM)

272	1201 1206 1706 1701	17	0.0	-161.6	-806.9	0.2	-161.6	-807.0	322.8	739.7 GLB Y 0.2		
				-163.6	-805.2	0.3	-163.3	-805.3	321.0	737.3 0.7		

## STRESSES IN BENDING TRIANGULAR PLATES (TRI63 ELM)

291	1601 1801 1901 17	1	0.0	-123.8	-829.3	0.0	-224.3	-53.9	-909.2	427.7	693.5 LCL X-16.6	
				-101.5	-853.3	0.0	-222.9	-40.3	-914.4	457.0	694.9 -15.3	

## STRESSES IN QUADRILATERAL BENDING PLATE (QUA64 ELM)

310	1901 1906 1906 1601	17	1	0.0	-155.6	-802.0	21.6	-154.7	-602.7	326.0	737.6 GLB Y 1.9	
				-149.3	-800.9	26.5	-148.2	-601.9	325.9	739.1 2.3		

## STRESSES IN BENDING TRIANGULAR PLATES (TRI63 ELM)

310	1601 1904 2001 17	1	0.0	-124.8	-855.7	0.0	-167.9	-79.3	-601.2	411.0	664.3 LCL X-13.6	
				-114.2	-836.1	0.0	-192.6	-86.1	-871.1	410.5	655.9 -14.0	

TABLE C-3 FINE MESH #4

SELECTIVE STRESS OUTPUT FOR LOAD CASE 15								STRESS : KG/CM <sup>2</sup>								
ELEM	1	2	3	4	5	6	7	SIGMA X	SIGMA Y	SIGMA Z	TAU	SIGMA 1	SIGMA 2	TAU MAX	HENCKY-YOUNG SIGI	
NO 1ST 2ND 3RD 4TH DAISY BOT																
STRESSES IN QUADRILATERAL BENDING PLATE (QUAD4 ELM)																
320	3197	3196	3296	3297	5	7	10.0	798.8	0.0	-26.0	799.7	9.1	395.3	795.2	LCL X-88.1	
								824.5	0.0	-32.0	825.6	12.3	406.8	819.7	-87.7	
342	3170	3196	3206	3278	5	7	0.0	131.1	906.7	43.4	400.1	120.7	390.2	852.1	GLB Y 86.8	
								-130.0	849.6	54.6	852.6	-141.8	497.2	931.7	26.8	
343	3297	3296	3396	2397	5	7	42.7	841.7	0.0	-24.1	842.5	42.0	400.2	822.3	LCL X-88.3	
								-46.5	785.6	0.0	-15.0	785.9	-46.8	416.3	610.3	-88.9
362	3278	3296	3396	2378	5	7	0.0	21.7	724.0	174.7	765.1	-119.3	392.2	774.9	GLB Y 76.8	
								-180.5	620.3	132.4	700.2	-200.4	450.3	819.0	81.5	
363	2397	2266	3496	3497	5	7	70.4	879.1	0.0	-10.3	879.2	70.3	406.5	846.3	LCL X-89.3	
								5.5	856.3	0.0	12.5	858.5	5.3	426.6	855.0	80.1
427	3497	3496	3596	3597	5	7	98.5	862.6	0.0	48.3	865.7	95.4	385.1	822.1	LCL X 86.4	
								844.3	0.0	46.5	847.2	96.7	375.2	803.2	86.4	
STRESSES IN BENDING TRIANGULAR PLATES (TRI163 ELM)																
439	3596	3496	3567	5	7	1020.9	230.7	0.0	-471.2	1247.2	12.3	617.5	1241.1	LCL X-24.9		
								1024.6	231.7	0.0	-474.6	1246.7	9.6	618.6	1242.0	-25.1
461	3506	3406	3486	5	7	1029.4	105.8	0.0	-20.4	1029.9	105.3	422.3	950.9	LCL X -1.4		
								963.4	174.4	0.0	-14.8	943.7	174.1	409.8	919.1	-1.0
STRESSES IN QUADRILATERAL BENDING PLATE (QUAD4 ELM)																
464	2506	3697	3595	3597	5	7	253.6	973.5	0.0	562.7	1206.0	21.1	642.4	1295.5	LCL X 59.4	
								380.0	966.8	0.0	567.7	1312.4	34.4	639.0	1295.6	55.7
STRESSES IN BENDING TRIANGULAR PLATES (TRI163 ELM)																
486	3597	3596	3687	5	7	170.1	977.7	0.0	-10.2	977.8	170.0	403.9	904.9	LCL X-89.3		
								170.6	962.9	0.0	-10.1	962.9	170.4	396.2	890.0	-89.3

TABLE C-3 FINE MESH #4, STBD  
(Cont'd)

SELECTIVE STRESS OUTPUT FOR LOAD CASE 53

ELEM	CORNER POINTS	L.C.	TOP	SIGMA X	SIGMA Y	SIGMA Z	TAU	SIGMA 1	SIGMA 2	TAU MAX	HENCKY-MON	ANG SIG1
											MIS. STRESS	....AXIS
427	3497 3496 3596 3597	17	T	127.6	852.1	0.0	47.1	855.2	124.6	365.3	800.2	LCL X 86.3
	2ND 1ST 3RD 4TH			124.3	834.9	0.0	44.6	837.7	121.5	358.1	784.1	86.4
<b>STRESSES IN QUADRILATERAL BENDING PLATE (QUAB4 ELM)</b>												
437	3497 3496 3596 3597	17	T	1003.0	210.0	0.0	-455.4	1210.4	2.7	603.8	1209.0	LCL X 24.5
	2ND 1ST 3RD 4TH			1001.4	209.0	0.0	-457.0	1210.2	1.1	604.6	1209.7	-24.6
<b>STRESSES IN BENDING TRIANGULAR PLATES (TRIB3 ELM)</b>												
439	3596 3496 3587	17	T	1003.0	210.0	0.0	-455.4	1210.4	2.7	603.8	1209.0	LCL X 24.5
	2ND 1ST 3RD 4TH			1001.4	209.0	0.0	-457.0	1210.2	1.1	604.6	1209.7	-24.6
<b>STRESSES IN QUADRILATERAL BENDING PLATE (QUAB4 ELM)</b>												
440	3420 3410 3610 3620	17	T	490.0	164.1	0.0	523.0	868.3	-224.3	546.3	999.5	LCL X 36.6
	2ND 1ST 3RD 4TH			-469.1	96.1	0.0	475.9	367.0	-739.9	553.5	976.6	60.4
441	3420 3620 3640 3440	17	T	110.0	-184.4	0.0	-520.6	503.8	-578.2	541.0	937.8	LCL X -37.1
	2ND 1ST 3RD 4TH			356.3	75.2	0.0	-566.9	801.0	-367.5	586.3	1035.0	-38.0
<b>STRESSES IN BENDING TRIANGULAR PLATES (TRIB3 ELM)</b>												
461	3596 3496 3486	17	T	1017.2	187.4	0.0	-49.3	1020.1	104.5	417.6	941.5	LCL X -3.4
	2ND 1ST 3RD 4TH			968.5	170.1	0.0	-40.3	970.5	168.0	401.3	898.4	-2.9
<b>STRESSES IN QUADRILATERAL BENDING PLATE (QUAB4 ELM)</b>												
465	3402 3702 3732 3632	17	T	0.0	-674.2	-126.8	-342.6	38.0	-839.0	438.5	858.7	GLB Y-64.3
	2ND 1ST 3RD 4TH			0.0	-675.1	-129.8	-342.7	35.5	-840.4	438.0	858.7	-64.3
484	3586 3687 3596 3587	17	T	262.4	936.5	0.0	528.0	1245.3	33.7	605.8	1228.8	LCL X 59.7
	2ND 1ST 3RD 4TH			355.9	929.2	0.0	530.1	1245.2	39.9	602.6	1225.7	59.2

TABLE C-4 FINE MESH #5, PORT  
 SELECTIVE STRESS OUTPUT FOR LOAD CASE 50  
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 STRESSES IN QUADRILATERAL BENDING PLATE (QUADS FLM)

ELEM	CORNER POINTS	L.C.	TOP	SIGMA X	SIGMA Y	SIGMA Z	TAU	SIGMA 1	SIGMA 2	TAU MAX	HENCKY-VON MIS. STRESS	ANG SIGI	....AXIS
307	1994 1992 2092 2094	14	T	230.9	1297.4	0.0	-10.1	1297.5	230.8	533.4	1198.9	LCL X-89.5	
	NO 1ST 2ND 3RD 4TH DAISY DOT			225.1	1255.9	0.0	-7.2	1256.0	225.0	515.5	1160.0	-89.6	
308	1996 1994 2094 2096	14	T	248.0	1079.1	0.0	93.1	1089.4	237.6	425.9	992.2	LCL X 83.7	
				198.3	1019.6	0.0	105.4	1032.9	185.0	423.9	953.9	32.8	
309	1998 1996 2096 2098	14	T	180.5	941.4	0.0	122.1	960.5	161.4	399.5	890.8	LCL X 81.1	
				146.1	084.1	0.0	126.5	905.2	125.0	390.1	849.6	80.5	
310	2000 1998 2098 2100	14	T	88.0	932.0	0.0	135.5	953.2	66.8	443.2	921.6	LCL X 81.1	
				95.0	888.4	0.0	157.1	918.4	65.1	426.7	887.6	79.2	
320	1980 2080 2100 2000	14	T	0.0	277.9	847.0	215.1	919.1	205.8	356.7	835.5	GLB Y 71.5	
				0.0	232.5	825.0	209.8	891.8	165.7	363.1	821.6	72.3	
324	1942 2042 2052 1952	14	T	0.0	159.6	870.2	-2.3	870.2	159.6	355.3	802.4	GLB Y-89.8	
				0.0	111.2	827.3	-20.2	827.8	110.7	358.6	778.4	-88.4	
325	1952 2052 2072 1972	14	T	0.0	155.6	1009.7	55.2	1013.3	152.0	430.6	946.5	GLB Y 86.3	
				0.0	141.5	963.5	54.7	967.1	137.9	414.6	906.1	86.2	
326	1972 2072 2092 1992	14	T	0.0	131.7	1258.9	231.1	1304.4	86.2	609.1	1263.5	GLB Y 78.9	
				0.0	143.8	1229.2	231.0	1276.3	96.7	569.8	1230.8	78.5	
329	1986 2086 2092 1992	14	T	765.3	343.3	0.0	723.7	1308.1	-199.5	73.8	1418.4	LCL X 36.9	
				754.3	384.7	0.0	684.4	1278.4	-139.4	708.9	1353.5	37.4	
333	2094 2092 2192 2194	14	T	415.9	1511.6	0.0	-64.8	1515.4	412.1	551.6	1357.1	LCL X-86.6	
				408.2	1503.8	0.0	-45.4	1505.7	406.3	549.7	1349.3	-87.6	
334	2096 2094 2194 2196	14	T	308.9	997.9	0.0	-7.8	998.0	308.8	347.6	884.9	LCL X-89.4	
				289.6	1005.0	0.0	2.1	1005.0	289.6	357.7	896.0	89.6	
335	2098 2096 2196 2198	14	T	193.4	884.2	0.0	30.0	885.5	192.1	366.7	806.8	LCL X 87.5	
				191.3	897.8	0.0	36.8	899.7	189.4	355.2	821.5	87.0	
347	2172 2192 2092 2072	14	T	0.0	121.7	1486.3	113.0	1493.6	112.4	690.6	1440.7	GLB Y 85.3	
				0.0	135.1	1438.0	117.8	1448.5	124.5	662.0	1390.5	84.9	
348	2152 2172 2072 2052	14	T	0.0	184.8	1006.6	64.4	1011.6	179.8	415.9	934.8	GLB Y 85.5	
				0.0	123.3	956.0	65.8	959.2	116.2	420.5	905.9	85.5	
349	2142 2152 2052 2042	14	T	0.0	143.6	870.7	39.0	872.8	141.5	365.6	811.4	GLB Y 86.9	
				0.0	129.5	840.1	36.9	842.0	127.6	357.2	786.0	87.0	

TABLE C-4 FINE MESH #5, PORT

(Cont'd)

SELECTIVE STRESS OUTPUT FOR LOAD CASE 50

STRESS : KG/CM\*\*2

ELEM NO	1ST CORN	2ND CORN	3RD CORN	4TH CORN	L.C. TOP	SIGMA X	SIGMA Y	SIGMA Z	TAU		SIGMA 1	SIGMA 2	TAU MAX	HENCKY-YOUNG MIS. STRESS	ANG SIGI ....AXIS
									TAU	TAU	SIGMA 1	SIGMA 2	TAU MAX	HENCKY-YOUNG MIS. STRESS	ANG SIGI ....AXIS
<b>STRESSES IN BENDING TRIANGULAR PLATES (TR103 ELEM)</b>															
374	2132	2122	14	1	17.8	-131.4	0.0	466.2	415.3	-528.9	-341.5	472.1	619.7	GLB X 40.5	
					39.6	-121.2	0.0	469.4	435.3	-517.1	-386.9	476.2	625.9	40.1	
378	2196	2116	2192	14	1	714.1	-219.2	0.0	415.3	872.5	-377.6	625.0	1110.5	GLB X 20.8	
					201.8	-270.1	0.0	441.4	929.3	-425.6	697.4	1236.1	19.6		
<b>STRESSES IN QUADRILATERAL BENDING PLATE (QUA84 ELEM)</b>															
384	2141	2191	2183	2173	14	1	201.9	3.9	0.0	464.1	627.3	-341.5	484.4	851.1	GLB X 36.7
					260.1	-53.6	0.0	464.4	593.4	-446.1	-386.9	490.2	855.2	35.7	
385	2173	2183	2186	2174	14	1	327.3	-151.5	0.0	400.2	554.2	-378.4	466.3	812.5	GLB X 29.6
					271.9	-237.1	0.0	387.4	480.9	-140.7	1401.4	215.2	593.1	803.0	830.0
388	2194	2192	2292	2294	14	1	232.1	1384.5	0.0	-140.7	1379.5	196.7	591.4	1307.2	LCL X-83.1
					218.6	1357.6	0.0	-157.7	1379.5	196.7	591.4	1292.5	-82.2		
329	2196	2194	2294	2296	14	1	244.6	1049.7	0.0	-111.2	1064.8	229.5	417.7	970.6	LCL X-82.3
					203.7	1030.1	0.0	-109.1	1044.4	194.5	425.0	962.0	-82.6		
390	2198	2196	2296	2298	14	1	142.5	876.1	0.0	-48.2	879.3	139.3	370.0	818.6	LCL X-86.3
					147.3	878.2	0.0	-41.0	860.5	145.0	367.8	817.7	-86.8		
392	2104	2102	2202	2204	14	1	188.4	0.0	767.5	169.1	813.3	142.7	335.3	752.2	GLB Z 15.1
					175.4	0.0	757.3	172.9	804.8	127.9	339.4	749.1	15.6		
402	2272	2292	2192	2172	14	1	0.0	38.8	1348.8	178.2	1372.6	15.0	678.8	1365.2	GLB Y 82.4
					71.3	1323.9	197.9	1354.5	40.8	626.8	626.8	1334.5	81.2		
403	2252	2272	2172	2152	14	1	0.0	139.5	1037.6	135.6	1057.7	119.5	469.1	1003.3	GLB Y 81.6
					51.5	974.6	159.6	1001.4	24.7	488.4	488.4	989.3	80.5		
404	2242	2252	2152	2142	14	1	0.0	116.7	889.5	161.2	929.9	76.3	426.8	894.2	GLB Y 77.4
					95.5	852.7	168.8	880.7	59.5	414.6	414.6	660.4	70.0		
405	2232	2242	2142	2132	14	1	0.0	101.0	804.0	126.5	826.8	79.0	373.9	790.3	GLB Y 80.1
					84.1	773.9	134.2	799.1	58.9	370.1	370.1	771.3	79.4		

TABLE C-5 SL-7, FINE MESH #6, STBD							SELECTIVE STRESS OUTPUT FOR LOAD CASE 51							STRESS : KG/CMM <sup>2</sup>	
ELEM	CORNER POINTS	L.C.	TOP	SIGMA X	SIGMA Y	SIGMA Z	TAU	SIGMA 1	SIGMA 2	TAU MAX	HENCKY-VON MISES STRESS	ANG SIG1	ANG SIG2		
NO 1ST 2ND 3RD 4TH DAISY BOT															
FORCES AND STRESSES IN FLANGES OR RCD'S (R002 ELM)															
ELEM															
ELEM		CORNERS POINTS		LOAD		AXIAL		SIGMA X		SIGMA Y		SIGMA Z		TAU	
NO		START		END		CASE		SIGMA 1		SIGMA 2		TAU		SIGMA 1	
124	4484	4485	15	15	0.91970-01			919.67							
STRESSES IN QUADRILATERAL BENDING PLATE (QUAB4 ELM)															
403	4484	4521	4465	4465	15	T	433.1	0.0	694.9	397.8	982.7	165.2	410.8	918.8	GLB 2 35.9
							244.6	0.0	447.5	297.7	660.5	31.6	314.5	645.3	35.6
STRESSES IN BENDING TRIANGULAR PLATES (TRIB3 ELM)															
409	4521	4466	4465	15	1	1	428.7	0.0	678.8	-304.8	883.2	224.3	329.5	795.2	GLB 2-33.8
							39.3	0.0	248.3	-262.8	426.6	-139.0	262.8	510.4	-34.2
410	4465	4495	4465	15	1	1	137.9	0.0	672.5	-319.6	821.7	-11.3	416.5	827.5	GLB 2-25.0
							244.2	0.0	648.9	-361.9	861.1	32.0	414.6	845.6	-30.4
420	4523	4486	4521	15	T	427.8	0.0	690.5	-235.0	828.4	269.9	729.2	729.1	GLB 2-30.4	
							53.1	0.0	279.5	-191.4	368.7	-59.1	222.4	419.5	-29.7

